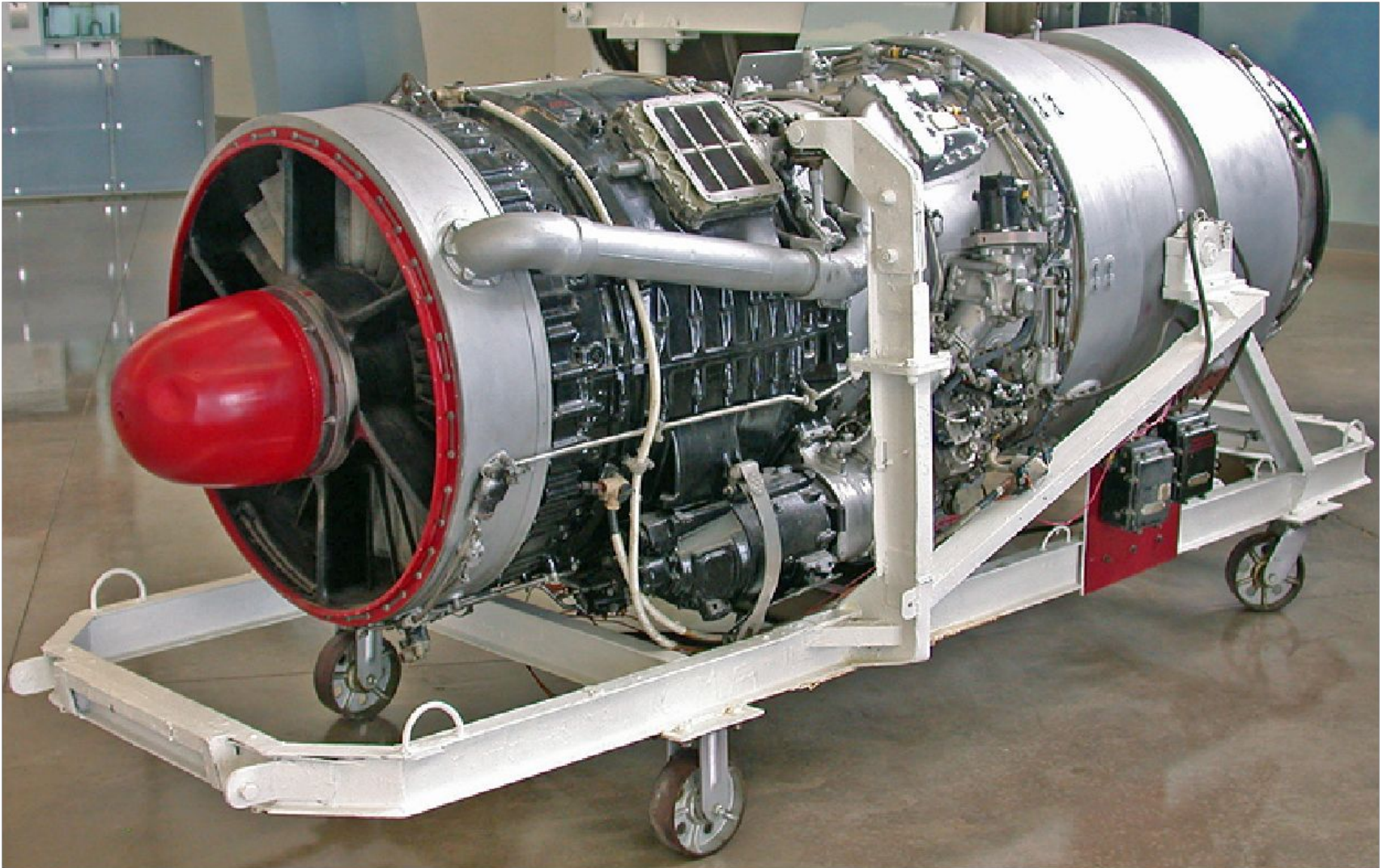


JET ENGINE MECHANICAL ARRANGEMENT

More generally known as '**GAS TURBINE**' Engine

THE WORKING PARTS OF A JET ENGINE

Rolls-Royce AVON – Canberra – 1950's technology



JET ENGINE MECHANICAL ARRANGEMENT

Rolls-Royce AVON

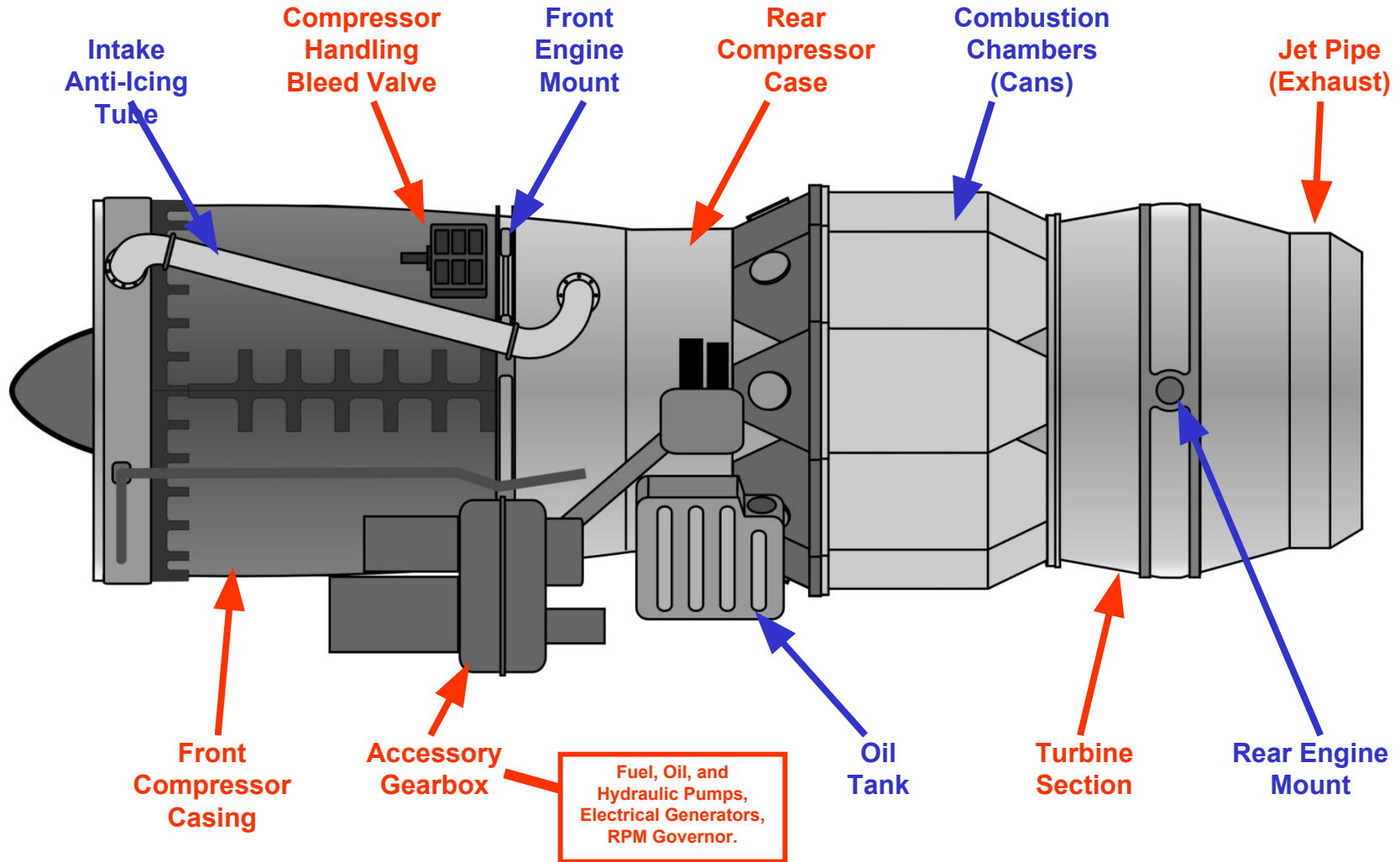
-

Cut away view

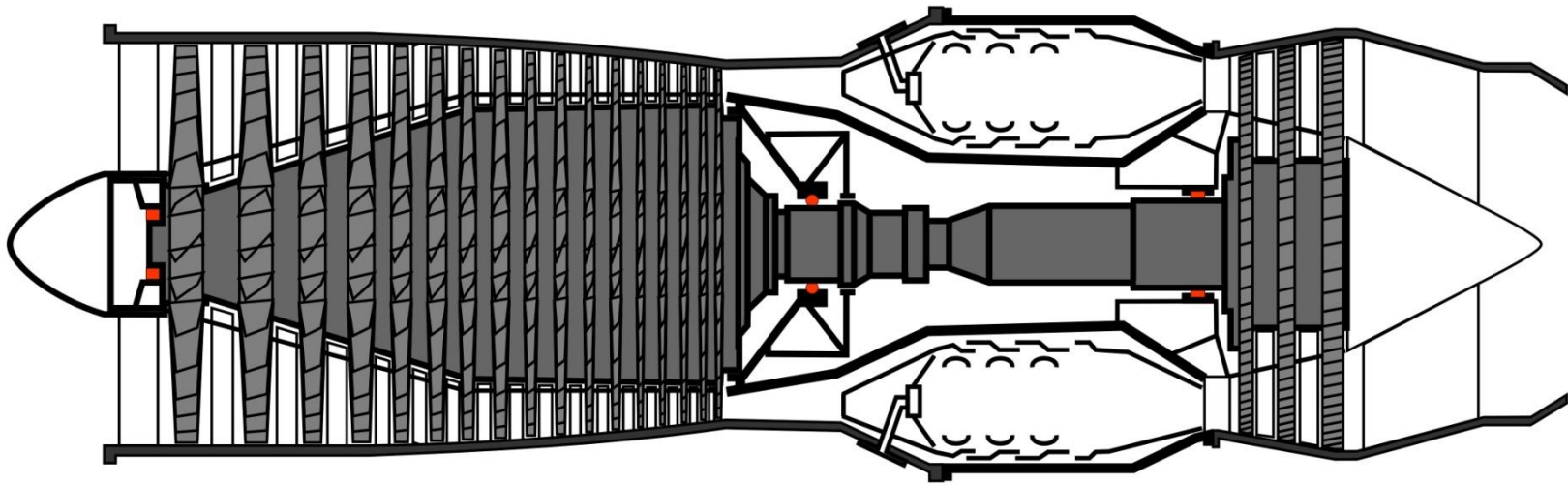


JET ENGINE MECHANICAL ARRANGEMENT

Rolls-Royce AVON - (Graphic Representation)



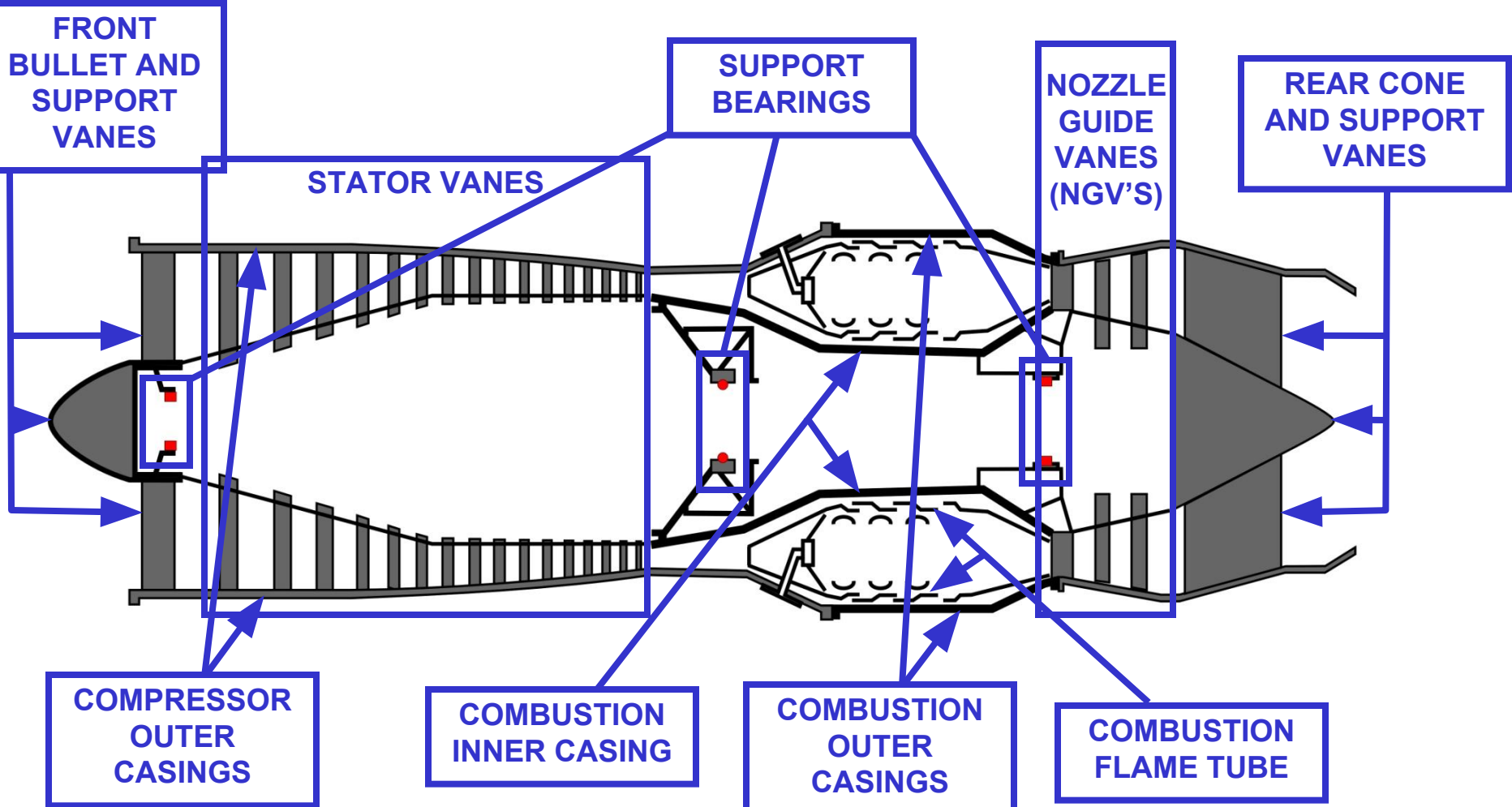
JET ENGINE MECHANICAL ARRANGEMENT



Turbo-jet = no Bypass Duct

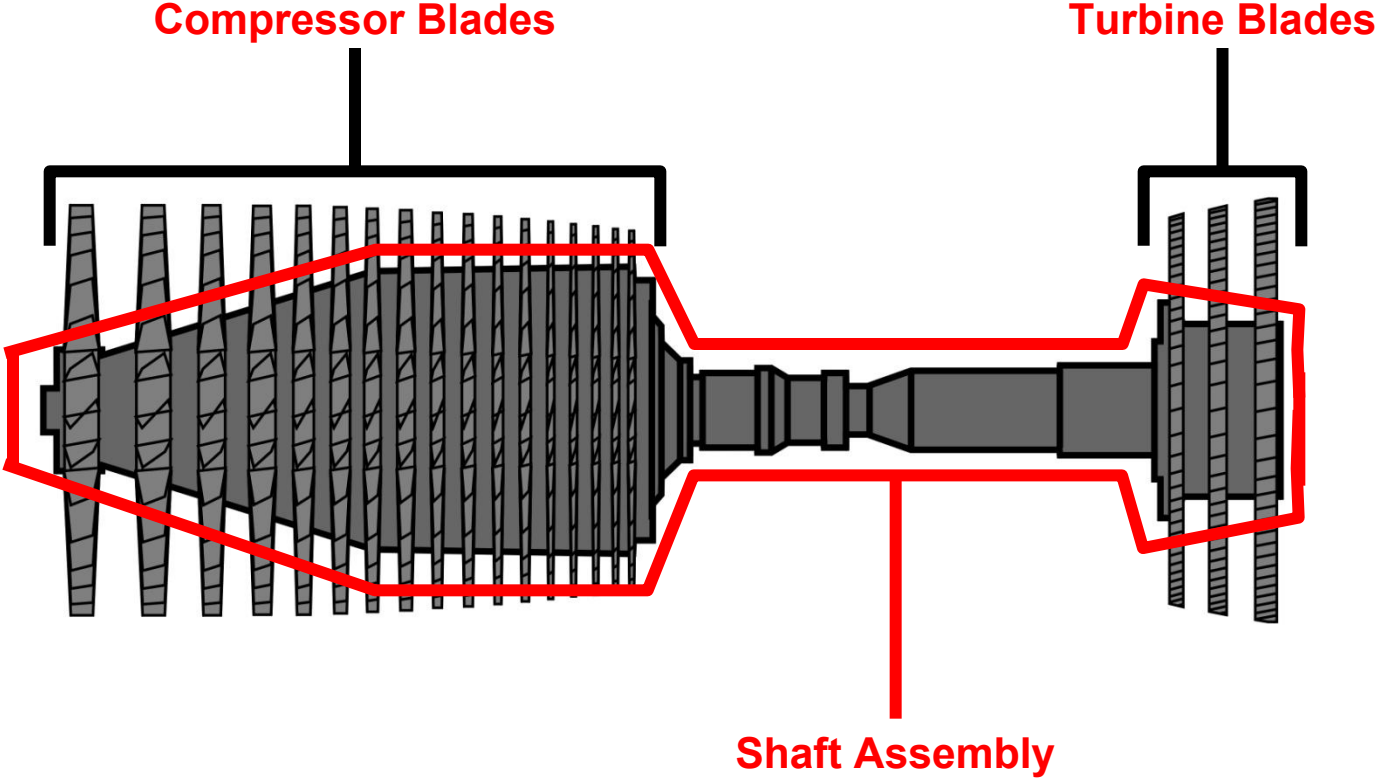
JET ENGINE MECHANICAL ARRANGEMENT

Engine Casings – Static Assembly



JET ENGINE MECHANICAL ARRANGEMENT

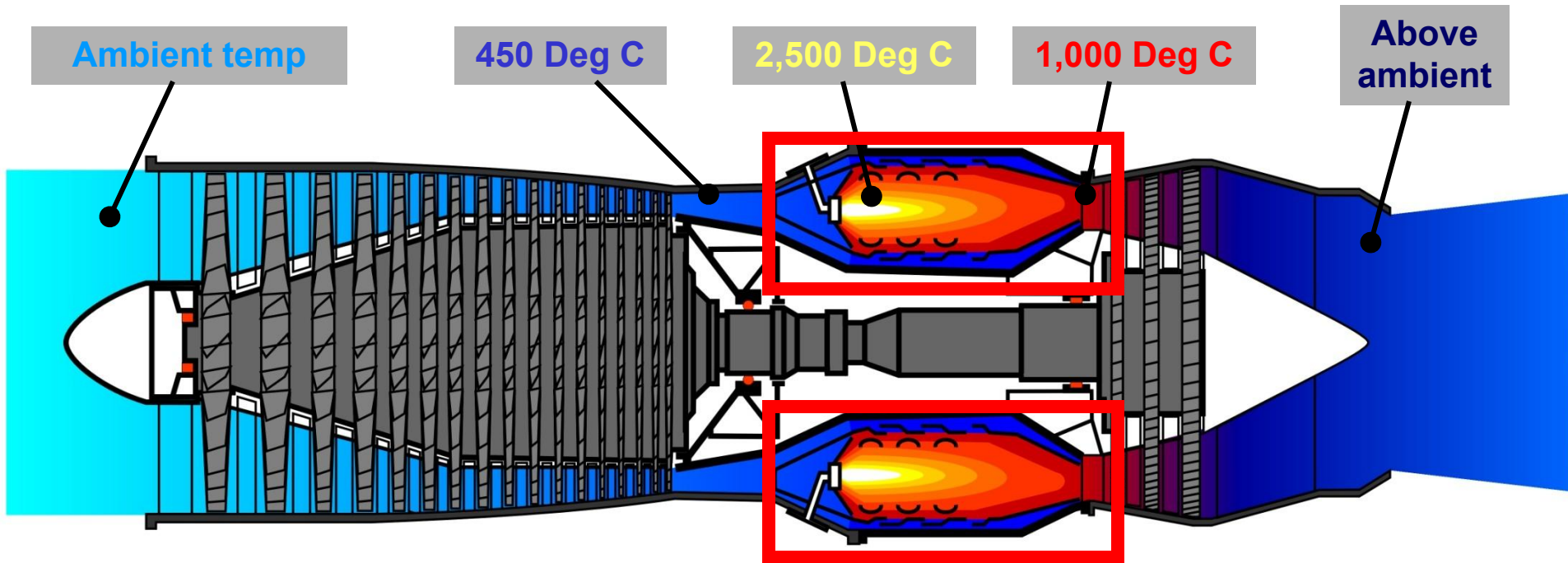
Rotating Assembly



JET ENGINE MECHANICAL ARRANGEMENT

Problem! Temperature changes cause the shaft and casings to expand and contract

But by different amounts!



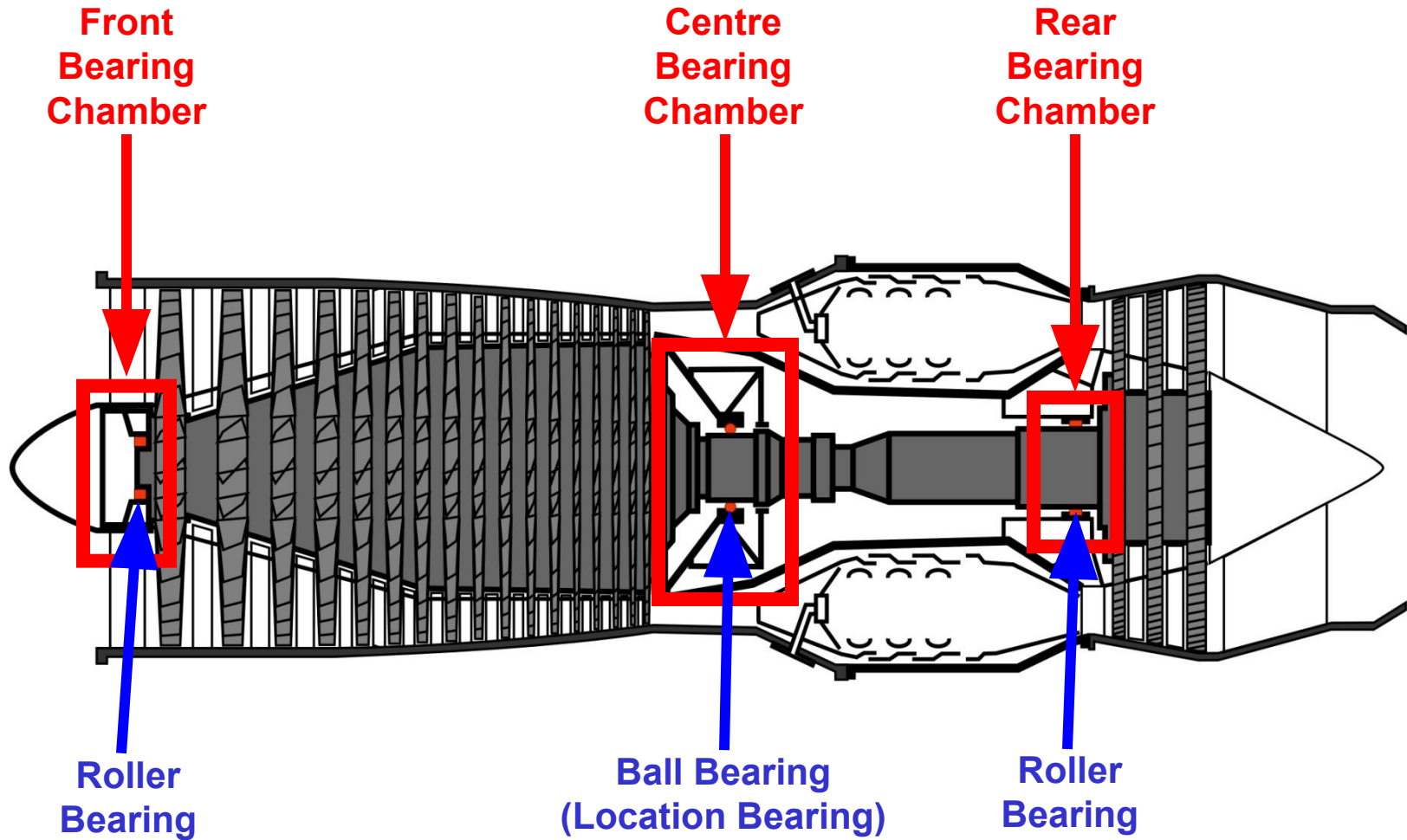
**These sections
expand first**

This could put tremendous unwanted forces on the casings and shaft leading to failures
(We'll look at what happens to the air passing through the **GAS PATH** in

Solution? Its all in the bearings!

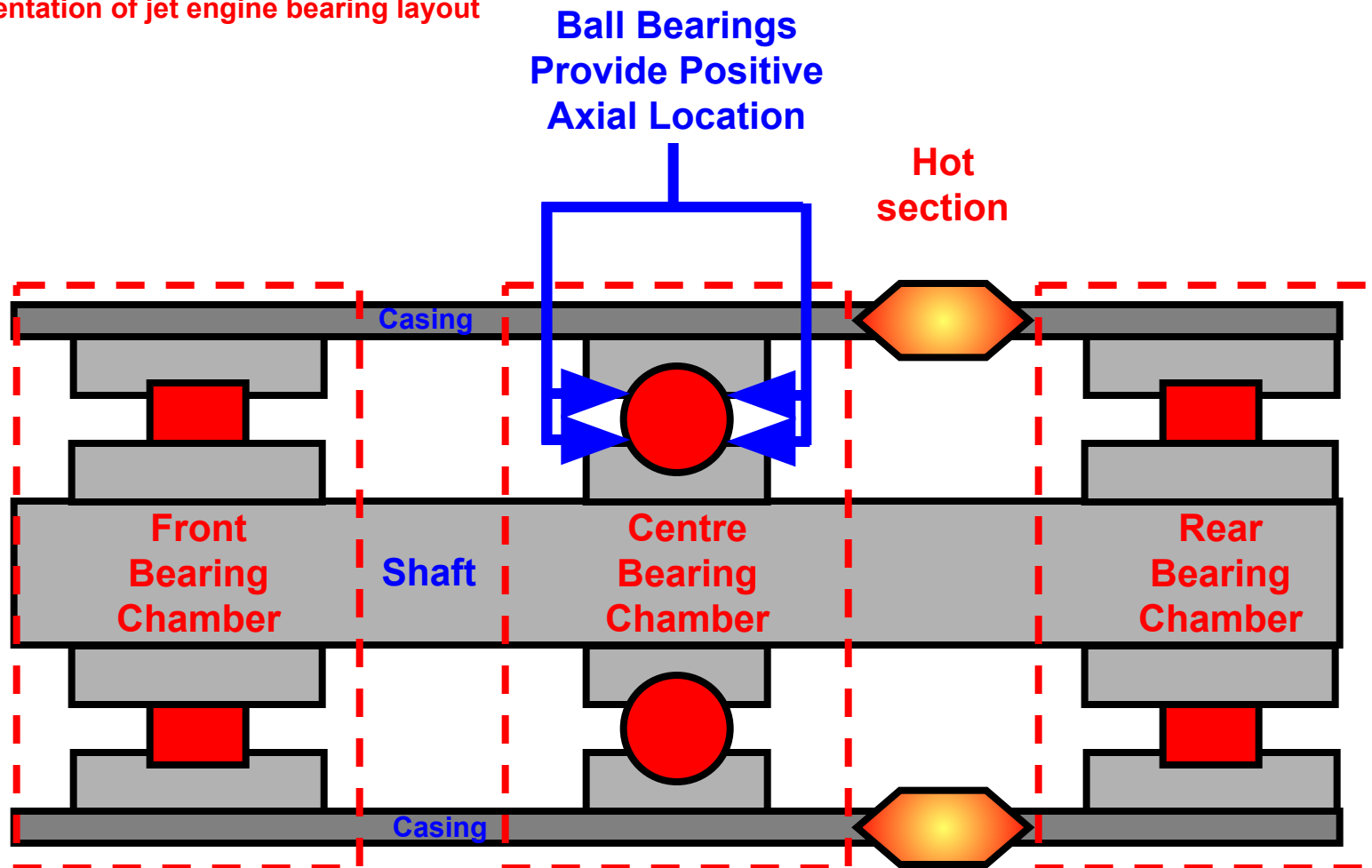
JET ENGINE MECHANICAL ARRANGEMENT

Bearing Chambers



JET ENGINE MECHANICAL ARRANGEMENT

Representation of jet engine bearing layout

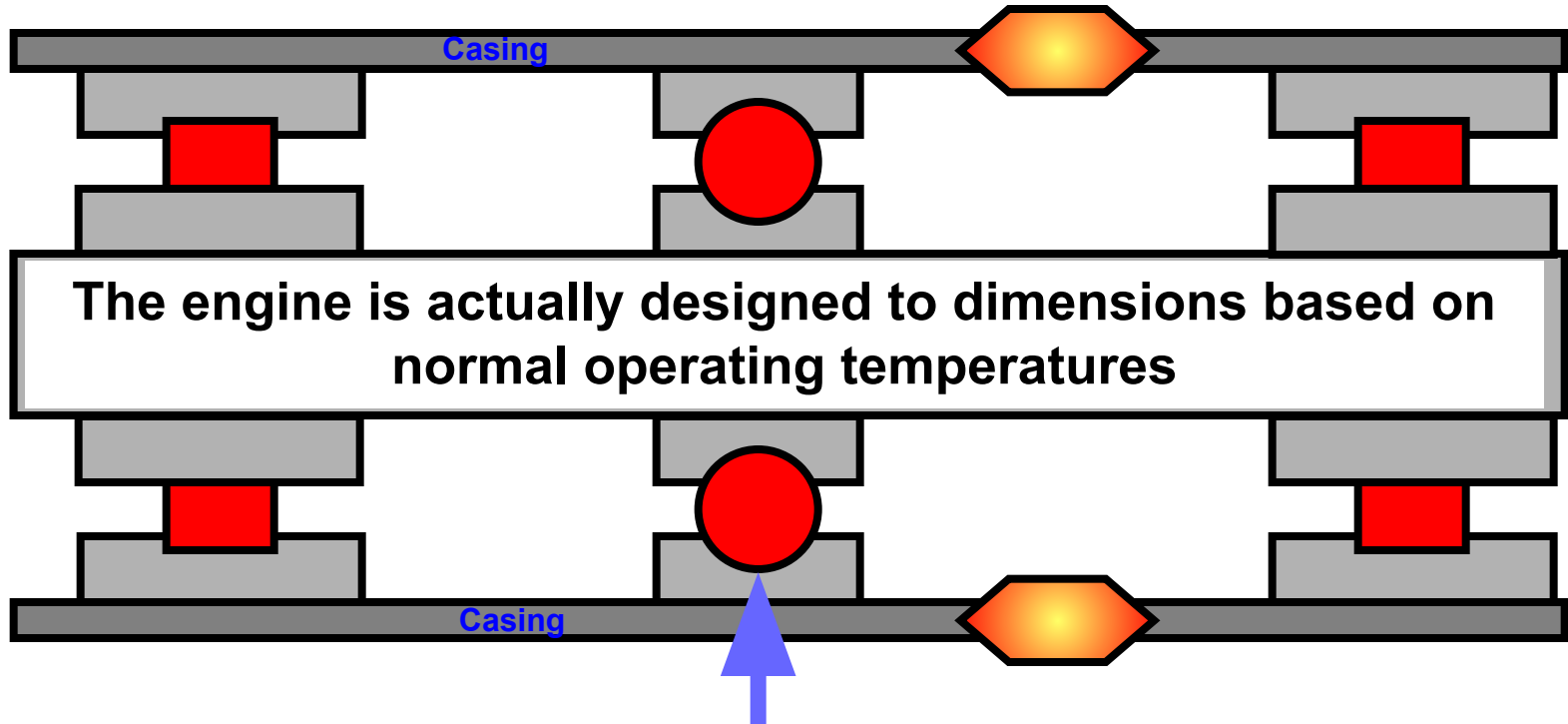


JET ENGINE MECHANICAL ARRANGEMENT

Representation of jet engine layout

The same happens at the front with pressured air and heat sink from the combustor heating the hardware

The casing expands first - pushing the rear outer bearing with it



The engine is actually designed to dimensions based on normal operating temperatures

Shaft and Casing have no differential movement at this point

JET ENGINE MECHANICAL ARRANGEMENT

JET ENGINE COMBUSTOR OPERATION

WHERE THE POWER IS PRODUCED

THE COMBUSTION PROCESS

IN A PISTON ENGINE: -

FUEL IS MIXED WITH AIR **BEFORE** ENTERING THE CYLINDERS

THE FUEL/AIR MIXTURE IS THEN COMPRESSED

THEN IT IS IGNITED BY A SPARK

ONCE FOR TWO REVS OF THE ENGINE (IN THE 4 STROKE CYCLE)

IN A JET ENGINE: -

AIR IS COMPRESSED AND FORCED INTO THE COMBUSTOR FIRST

THEN THE FUEL IS SPRAYED IN UNDER PRESSURE

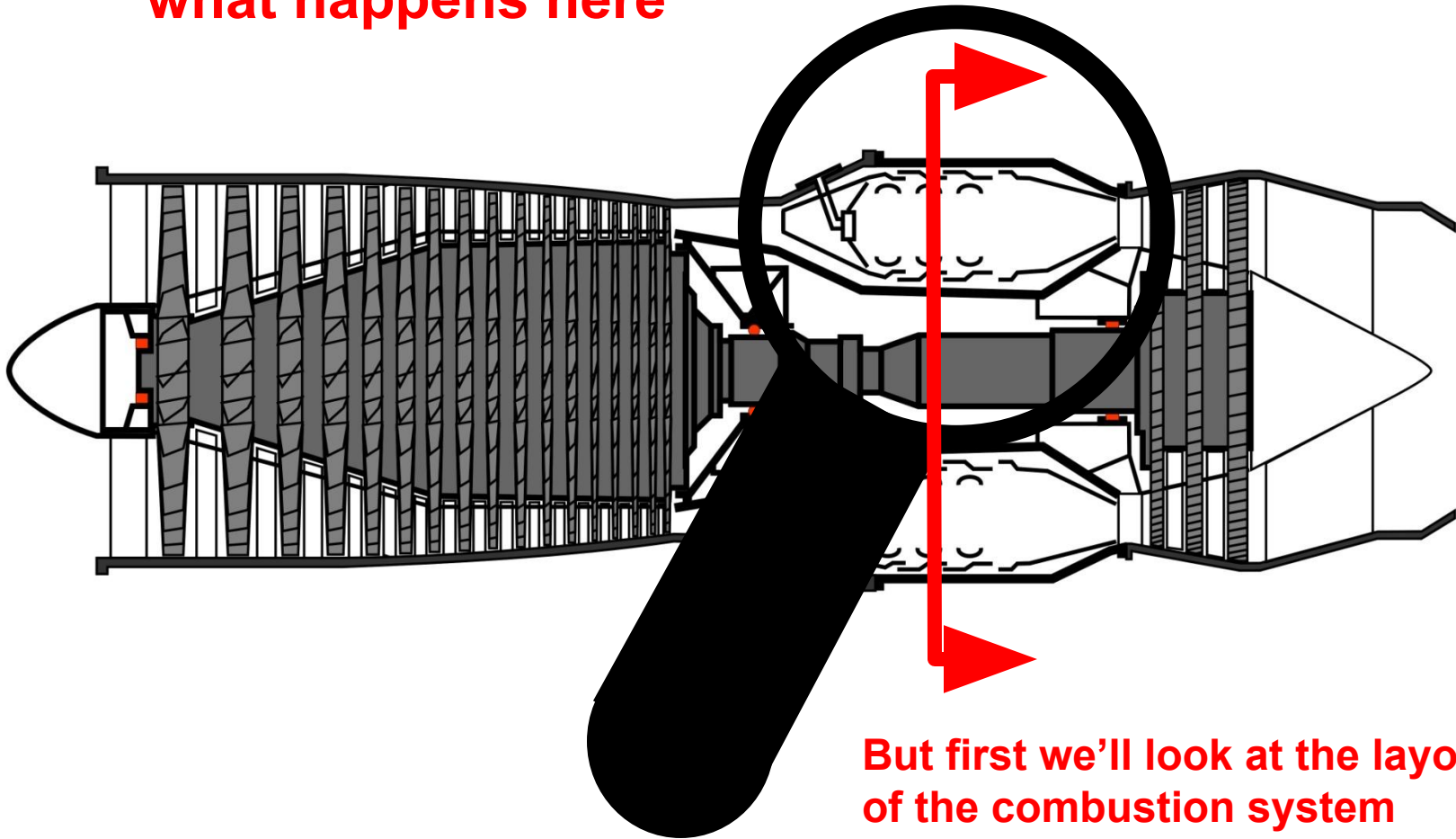
IT IS THEN IGNITED BY A SPARK (BUT ONLY ONCE FOR STARTING)

COMBUSTION IS THEN **CONTINUOUS** WHILST THE ENGINE IS RUNNING

THE SAME AS A PLUMBERS BLOW TORCH!

JET ENGINE – Combustion Process

**We're going to
look in detail at
what happens here**



JET ENGINE – Combustor Layout

COMBUSTOR LAYOUT – (ALSO HISTORICAL DEVELOPMENT)

INDIVIDUAL COMBUSTORS – ‘CANS’

1st JET ENGINES

WHITTLE ETC

ENGINE
STRUCTURE
AND SHAFTS

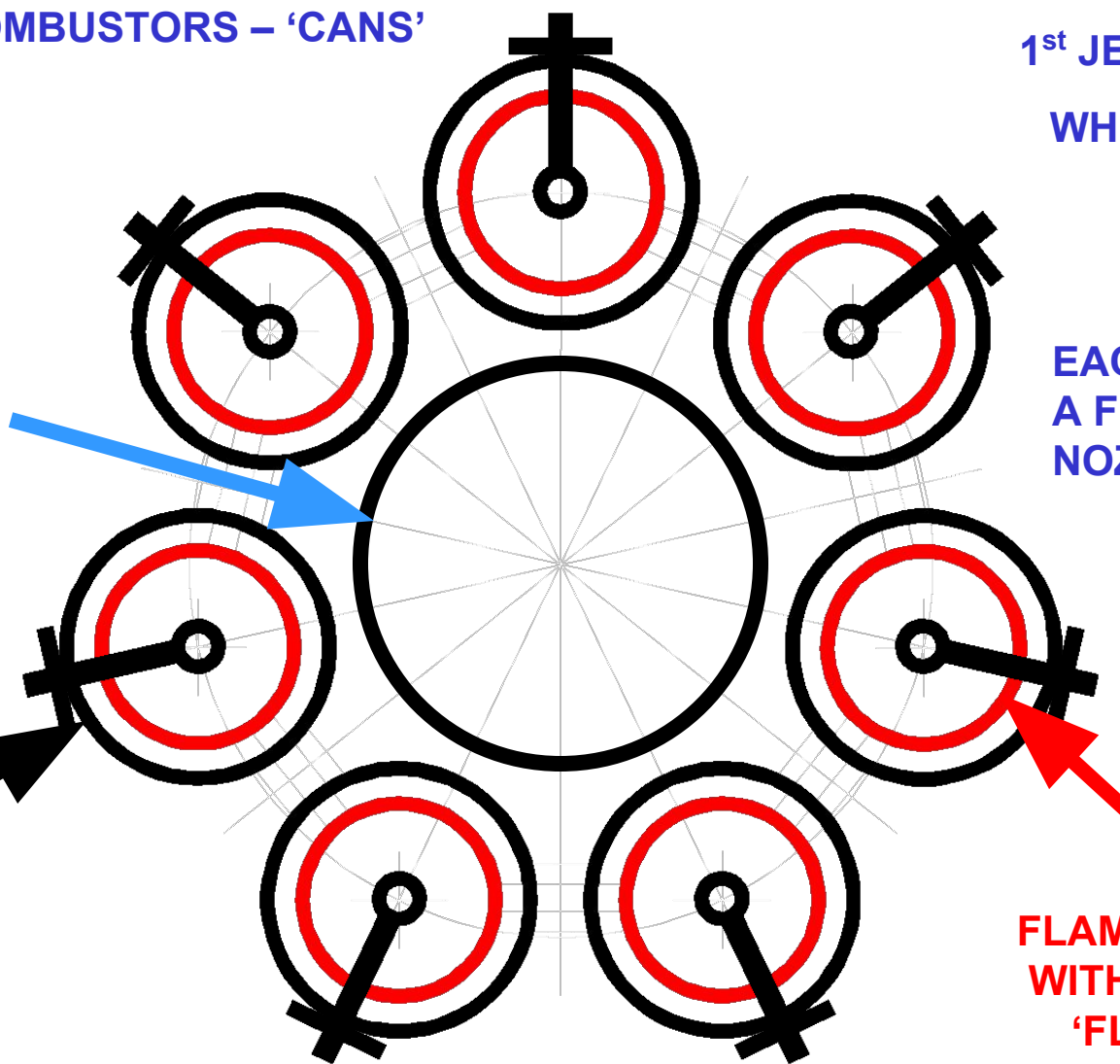
EACH CAN HAS
A FUEL SPRAY
NOZZLE (FSN)

OUTER
CASING

INNER
CASING

FLAME CONTAINED
WITHIN THIS TUBE
‘FLAME-TUBE’

JET ENGINE – Combustor Layout



COMBUSTOR LAYOUT – (ALSO HISTORICAL DEVELOPMENT)

CANS CONNECTED TOGETHER

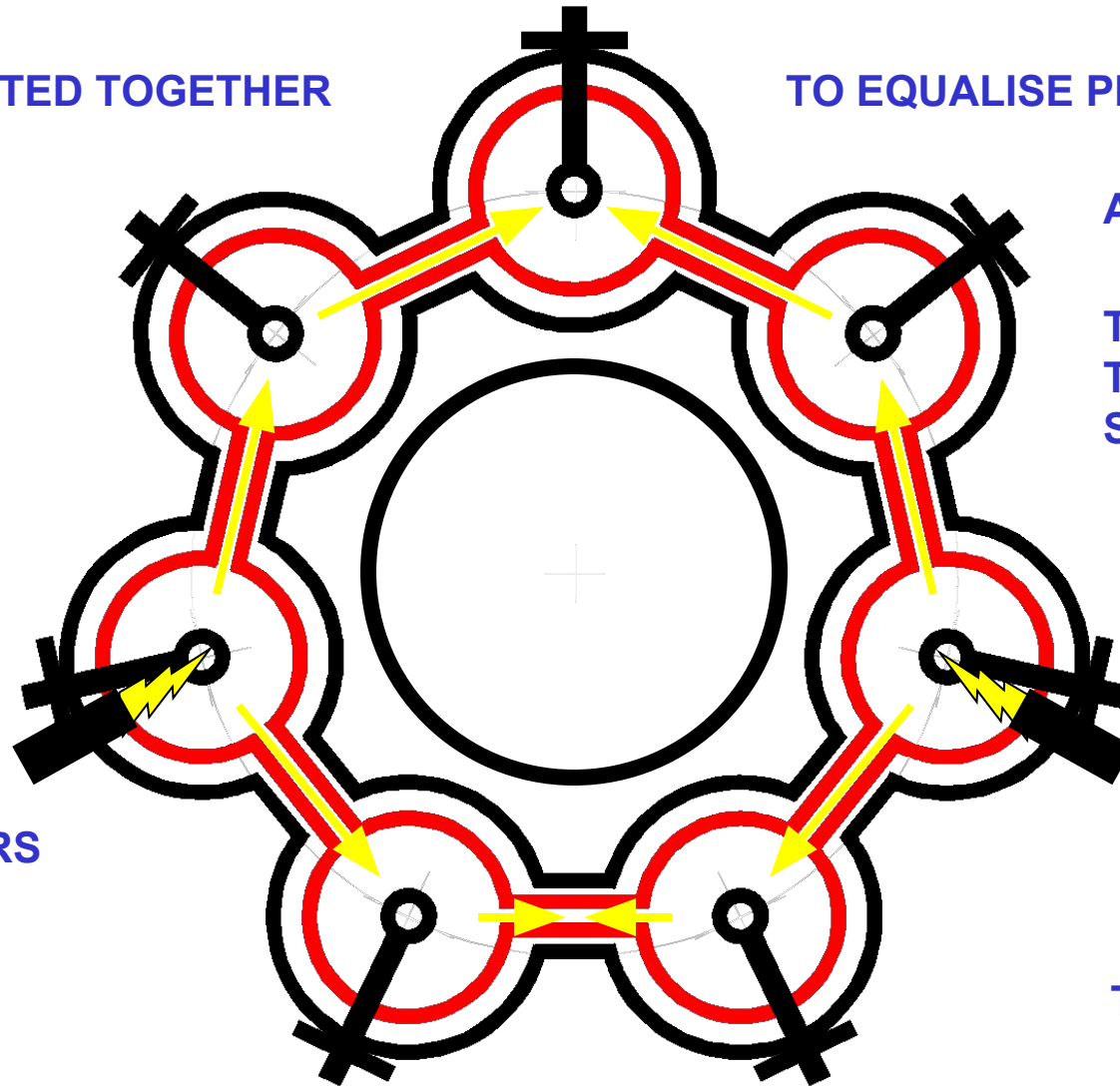
TO EQUALISE PRESSURES

AND

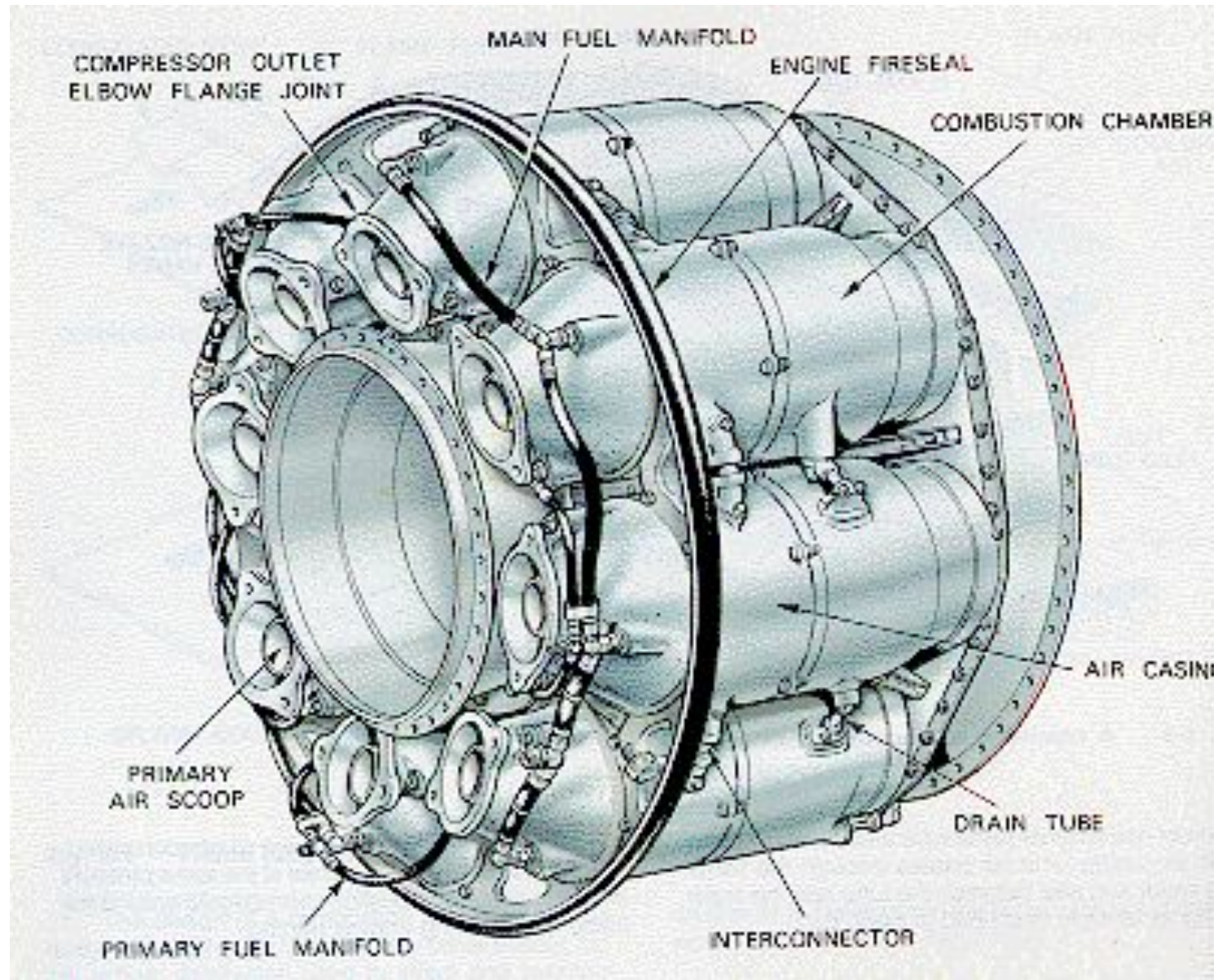
TO SPREAD
THE FLAME ON
START UP

IGNITERS

TYPICAL OF
DART SERIES
TURBO PROPS



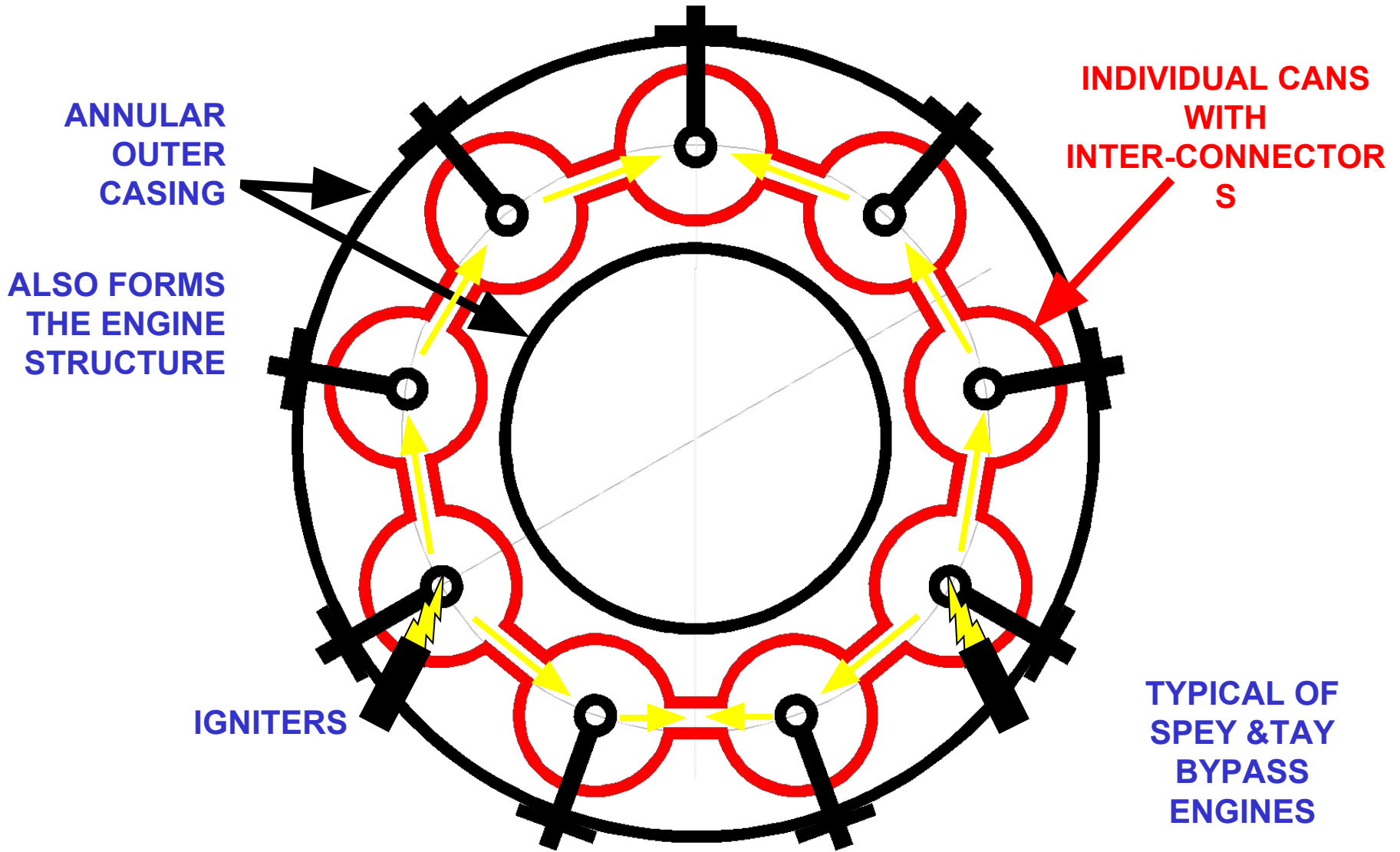
JET ENGINE – Combustor Layout



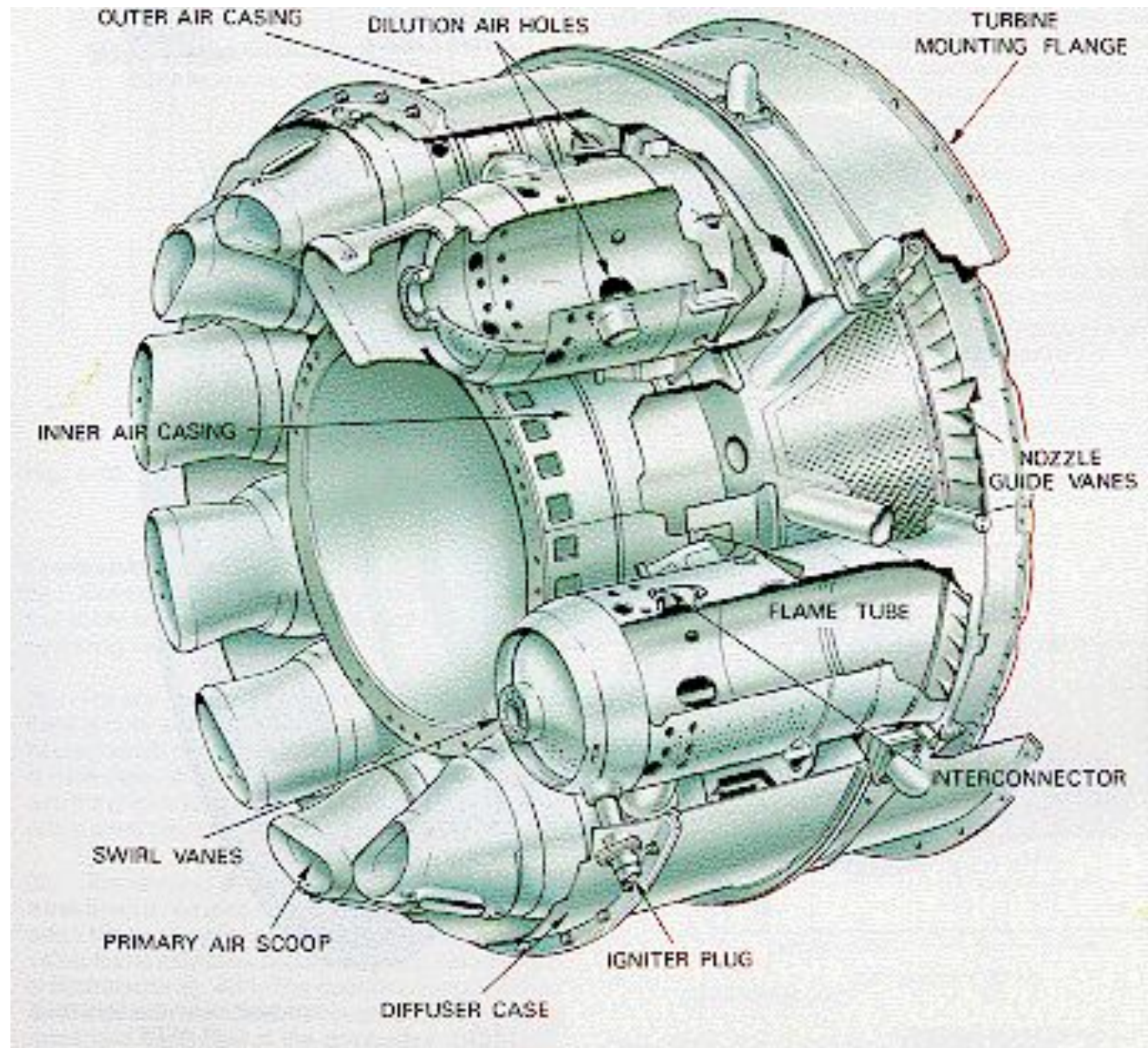
JET ENGINE COMBUSTER OPERATION

COMBUSTOR LAYOUT

'CAN-ANNULAR' LAYOUT



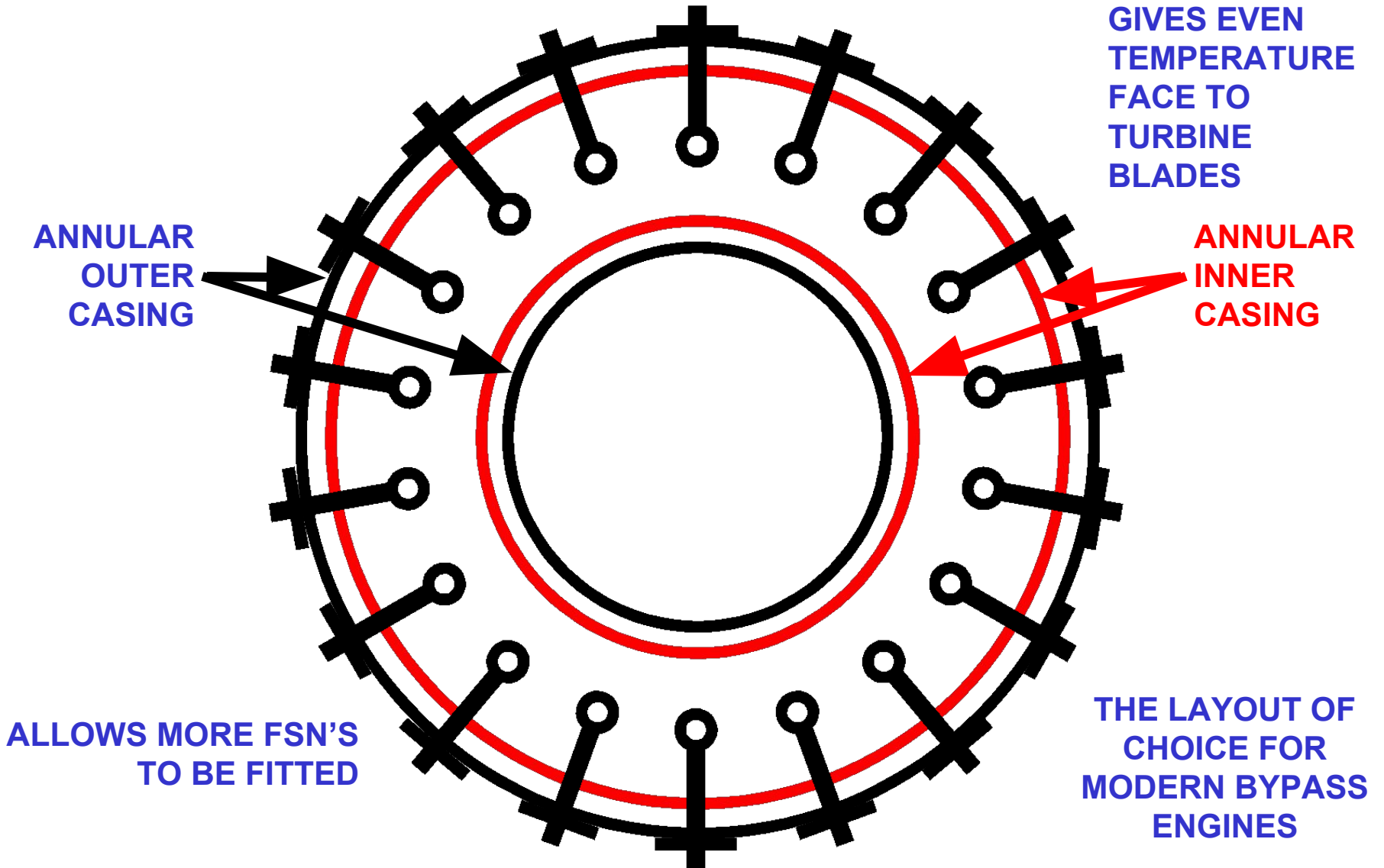
JET ENGINE – Combustor Layout



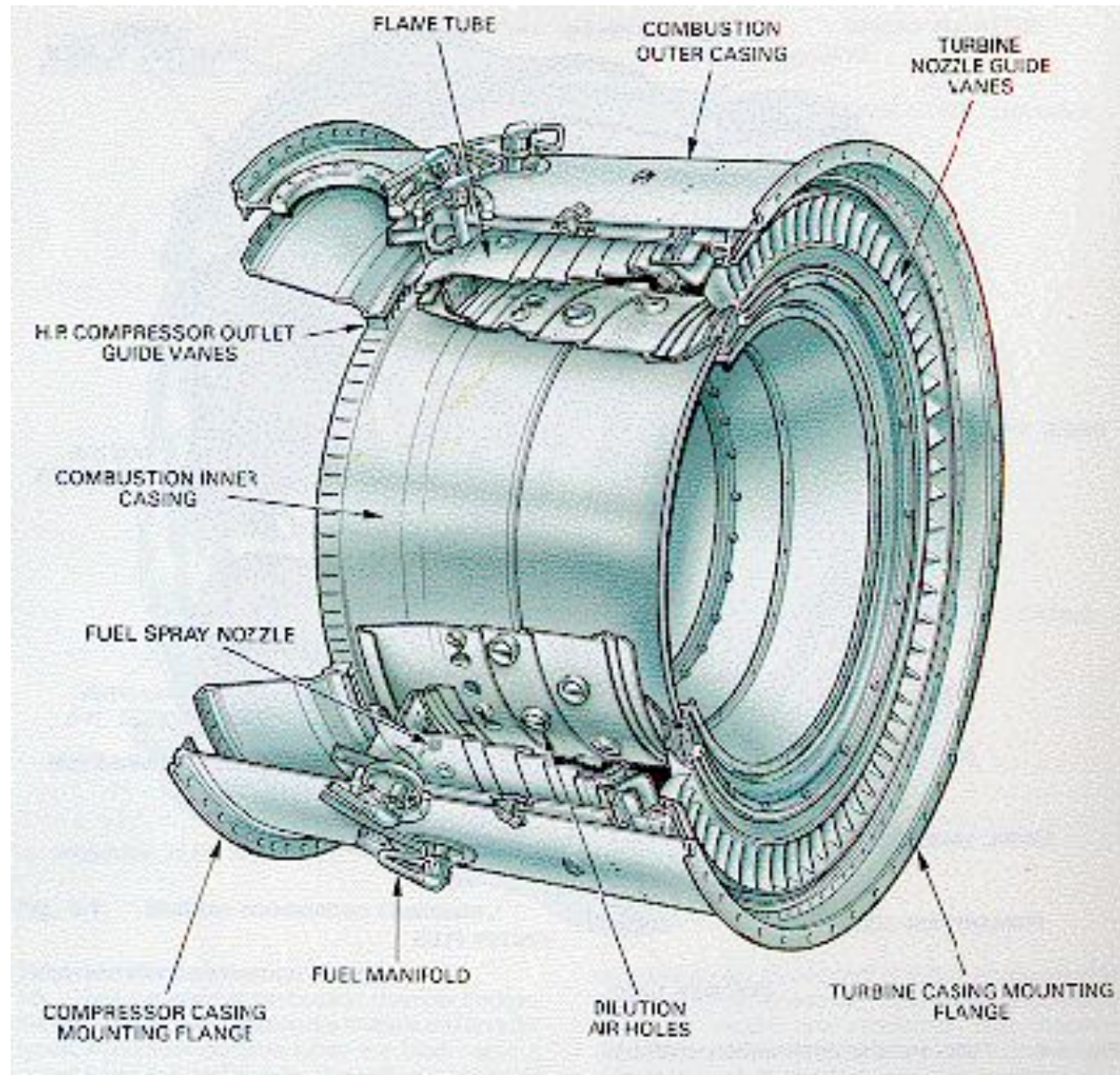
JET ENGINE COMBUSTOR OPERATION

COMBUSTOR LAYOUT

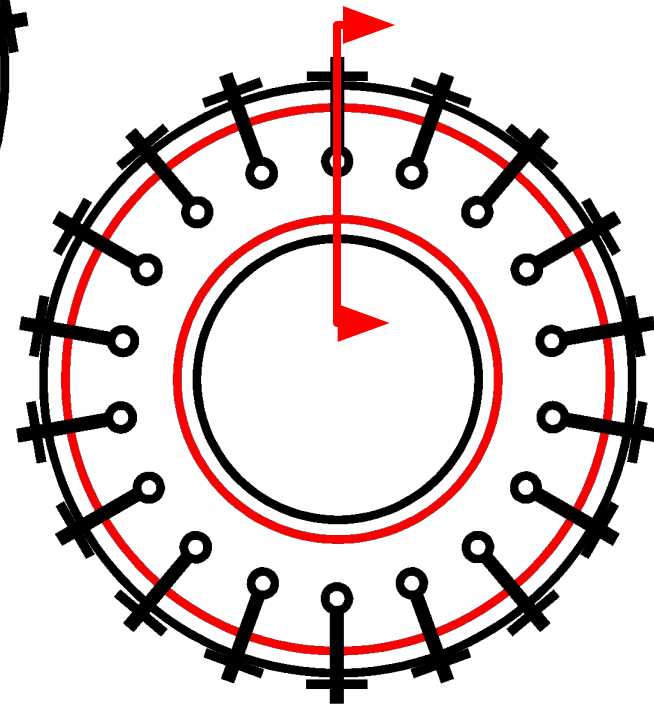
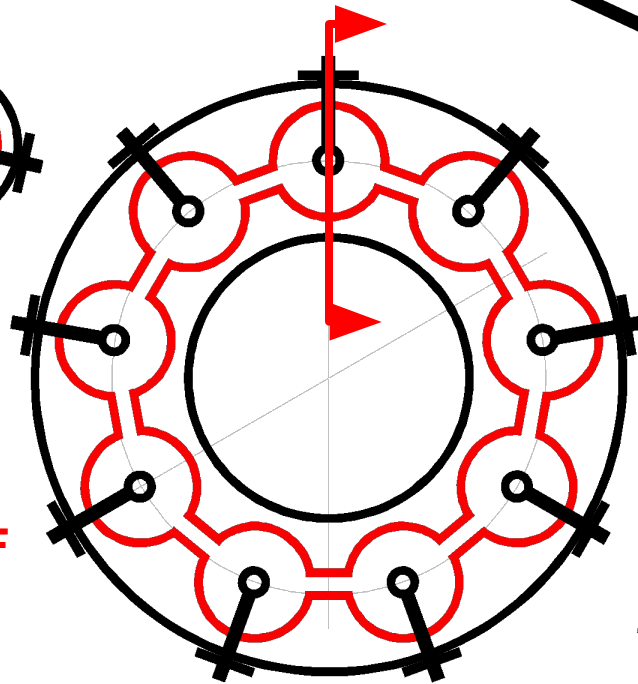
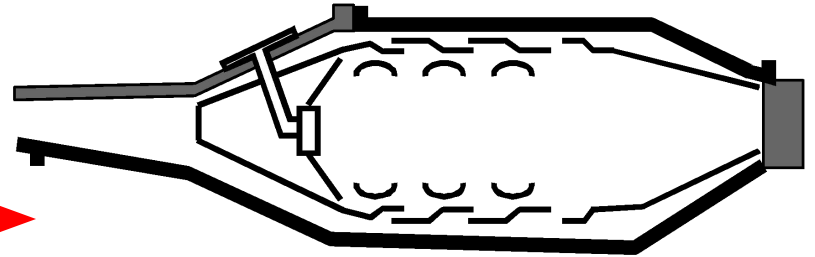
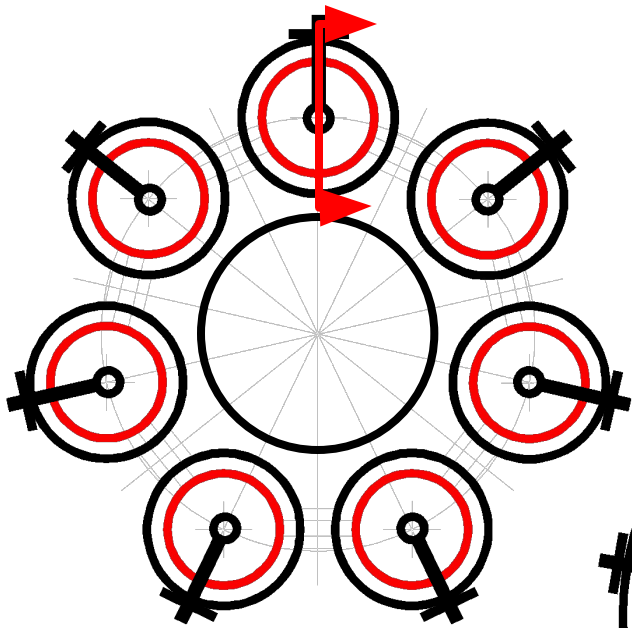
ANNULAR LAYOUT



JET ENGINE – Combustor Layout



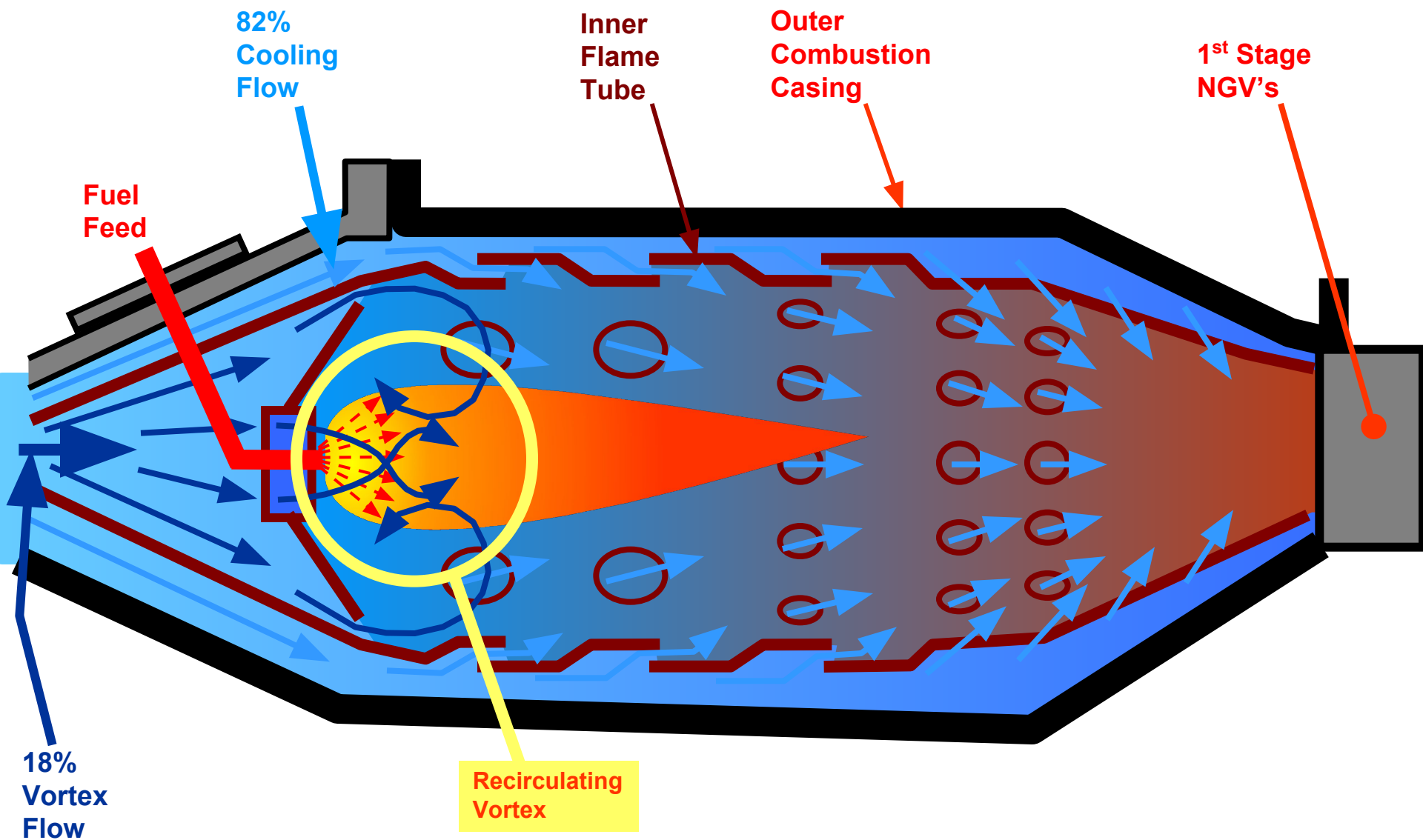
JET ENGINE COMBUSTER OPERATION



**THE CROSS SECTION OF
ALL THESE LAYOUTS IS
VIRTUALLY IDENTICAL**

AND THEREFORE SO IS THE AIRFLOW!

JET ENGINE – Combustor Layout



JET ENGINE – Combustion Process

JET ENGINE OPERATION

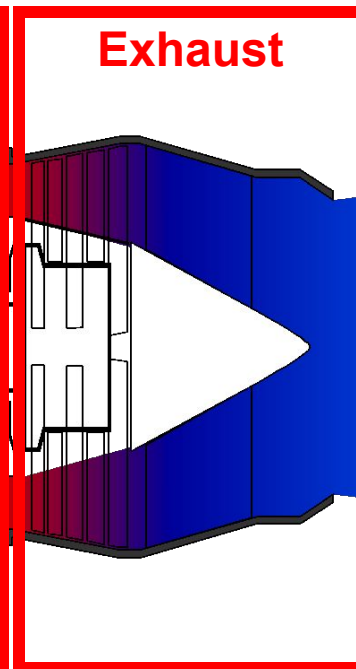
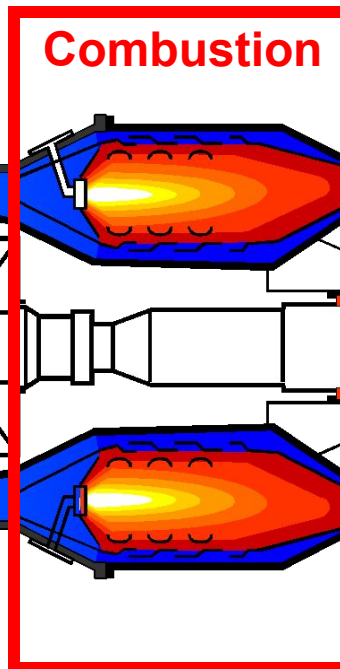
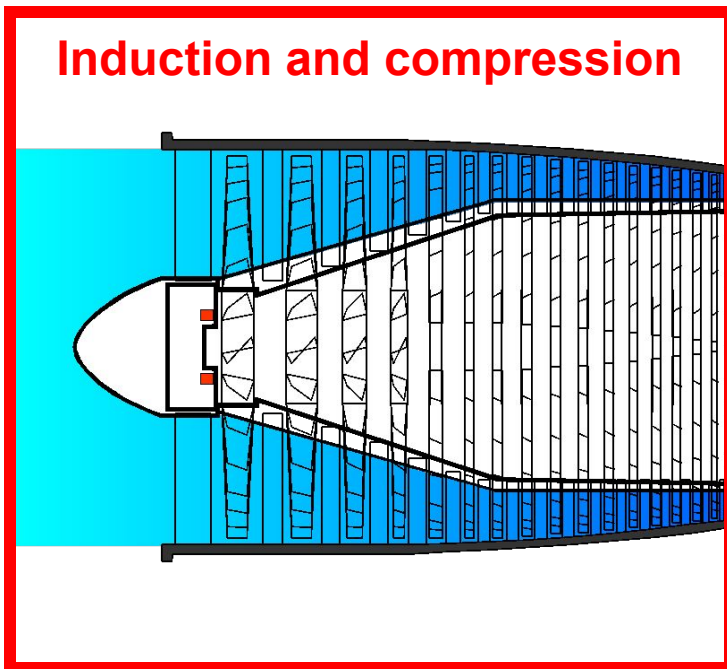
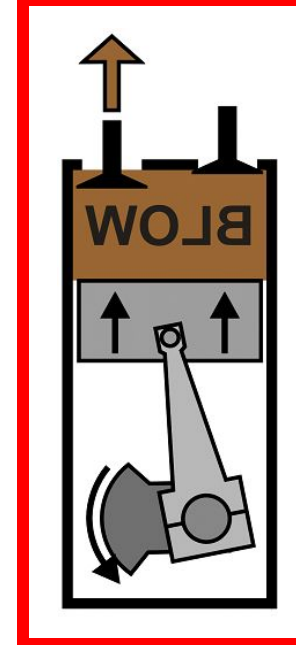
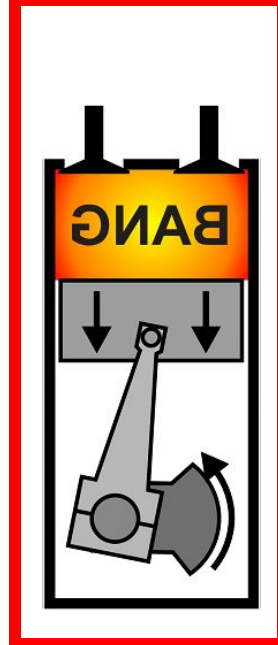
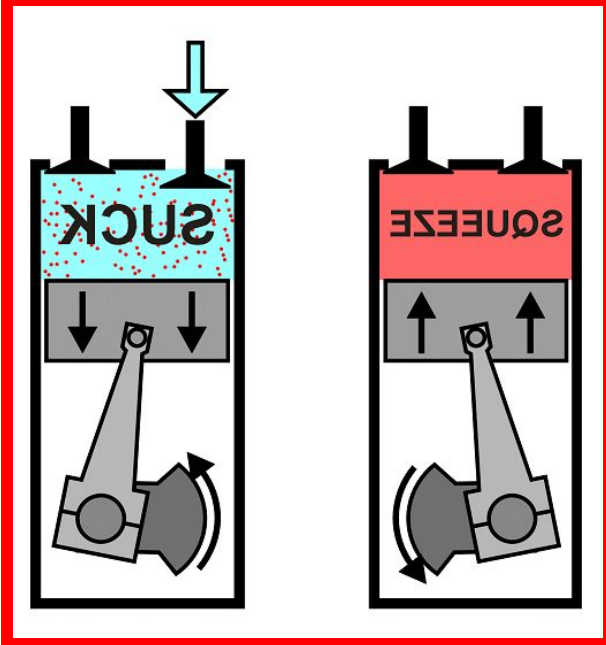
JET ENGINE HAVE THE SAME **OPERATING CYCLE** AS A PISTON ENGINE

I.E. SUCK - SQUEEZE - BANG - BLOW

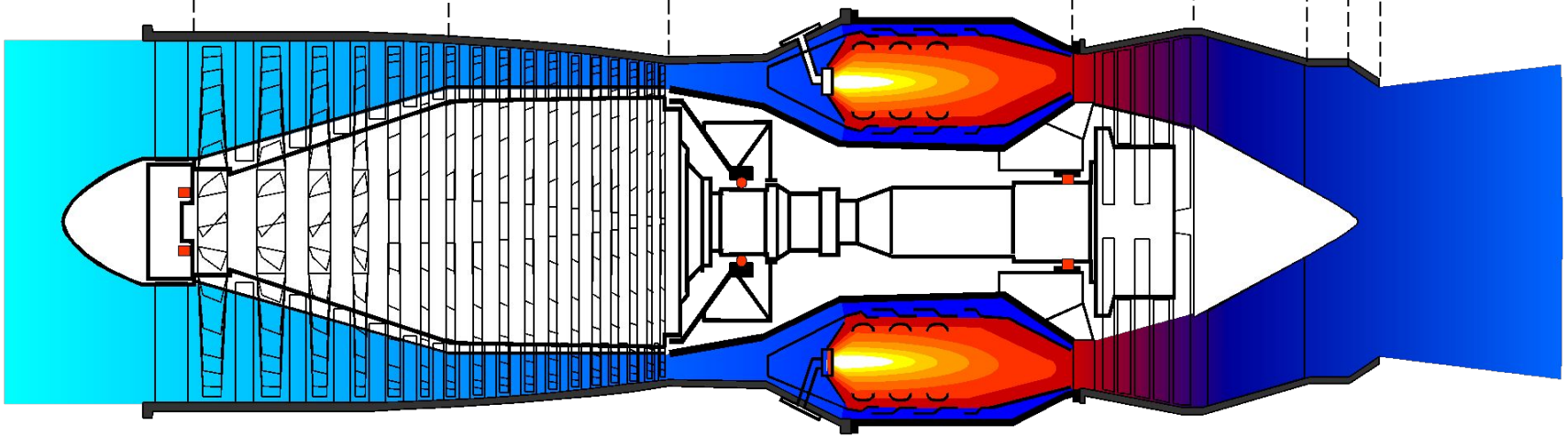
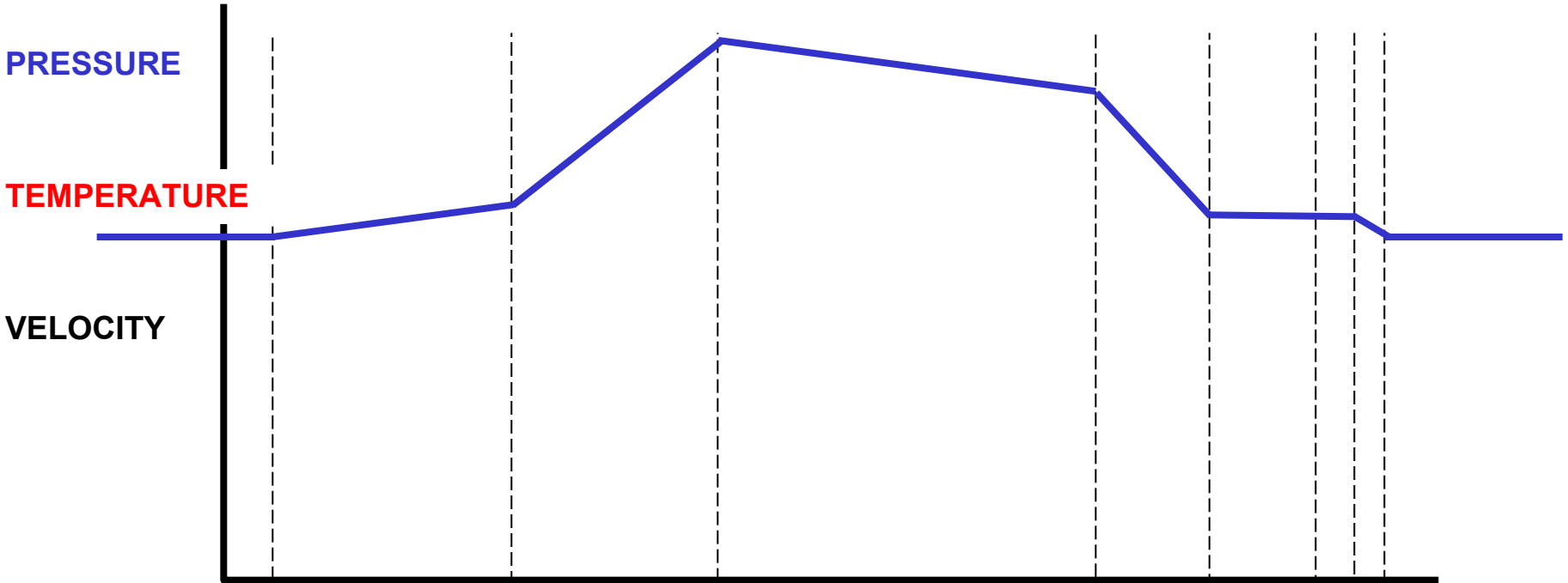
EXCEPT: -

PISTON ENGINES HAVE AN ***INTERMITTENT*** CYCLE

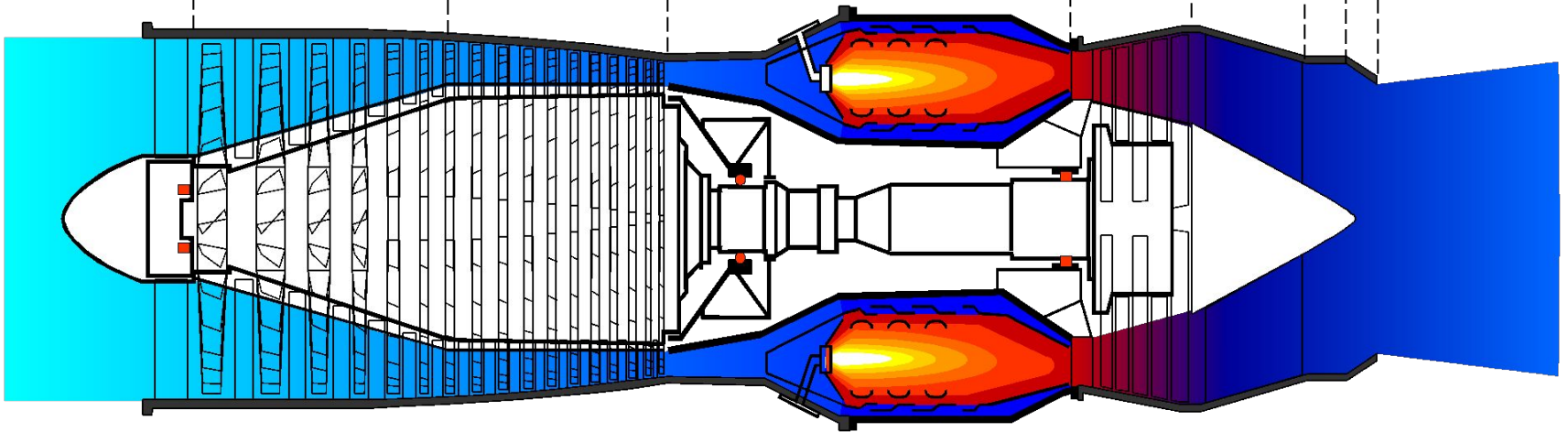
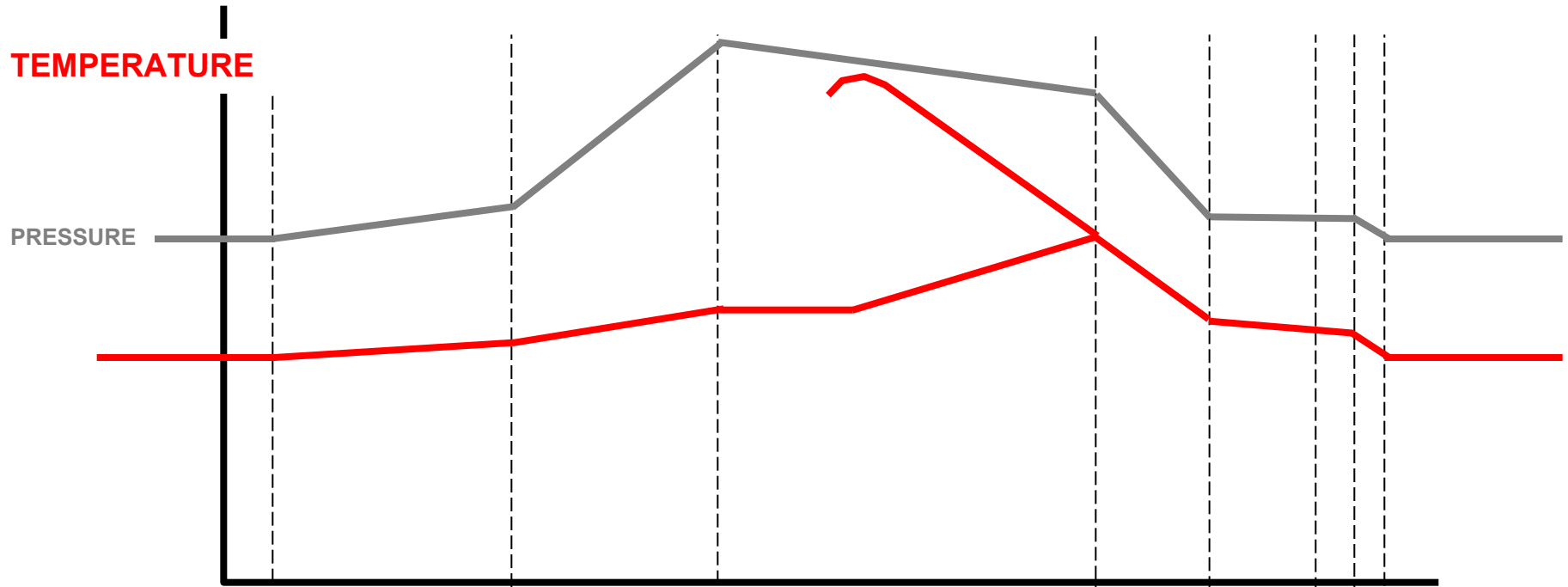
JET ENGINES HAVE A ***CONTINUOUS*** CYCLE



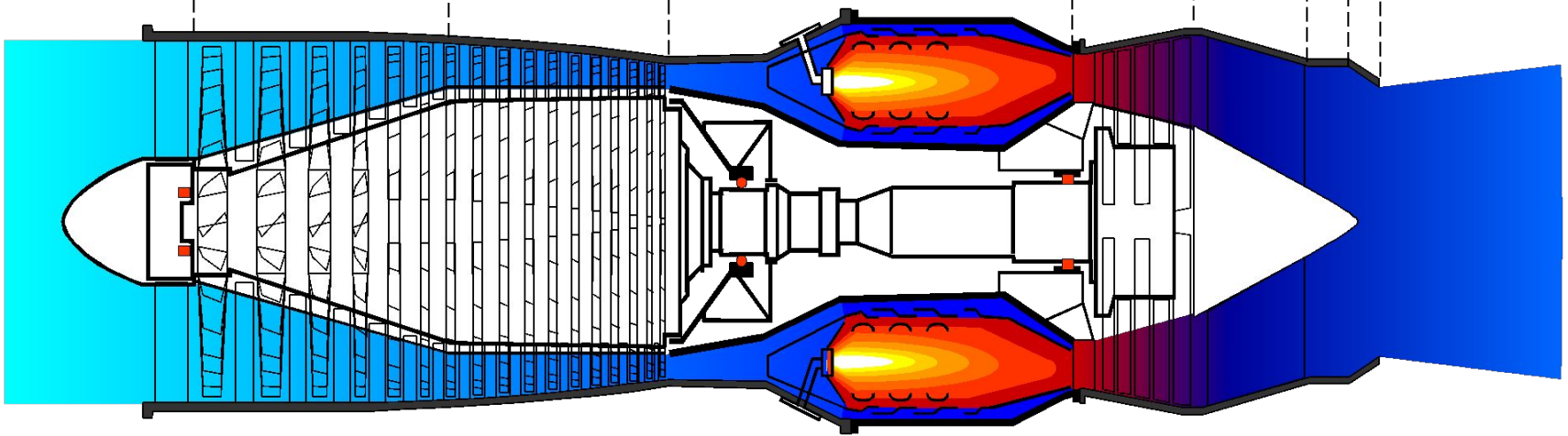
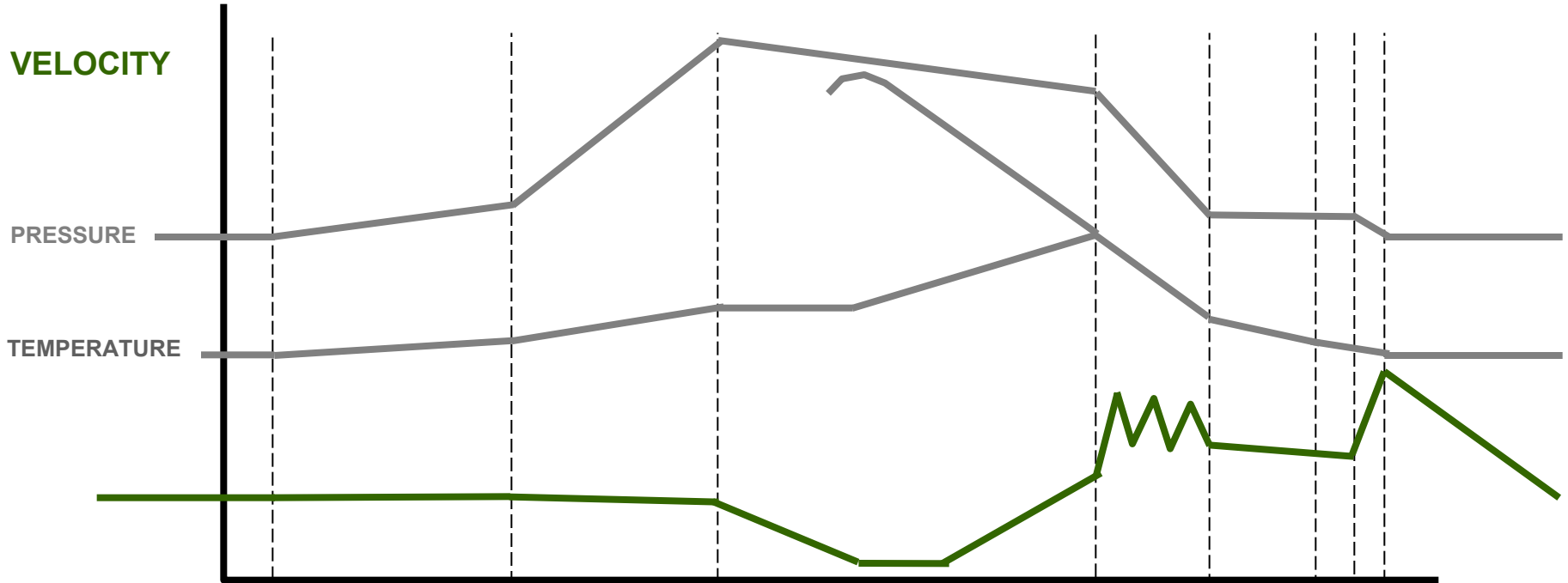
GAS TURBINE OPERATION



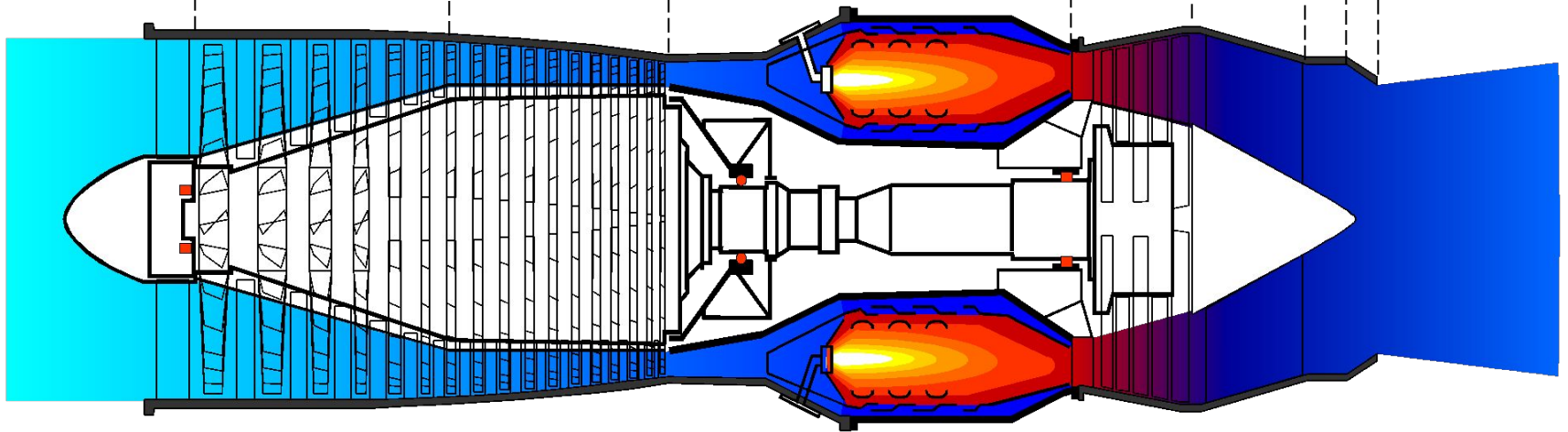
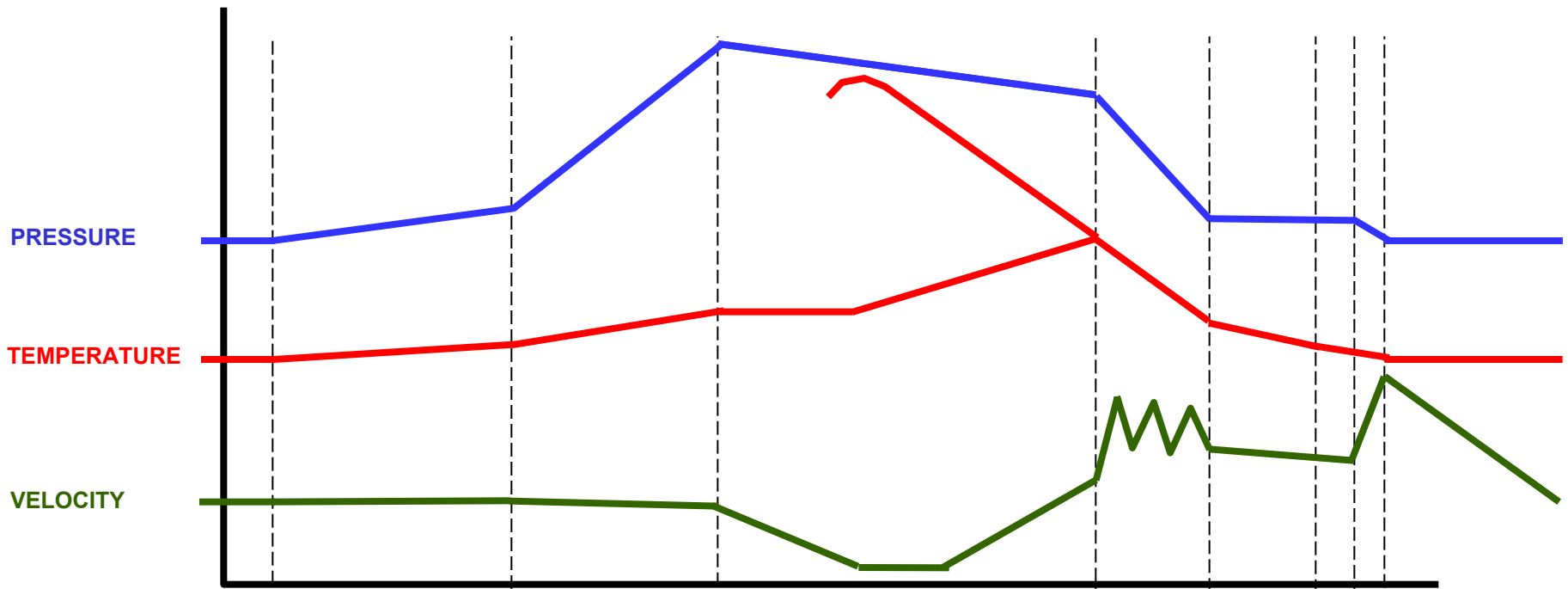
GAS TURBINE OPERATION



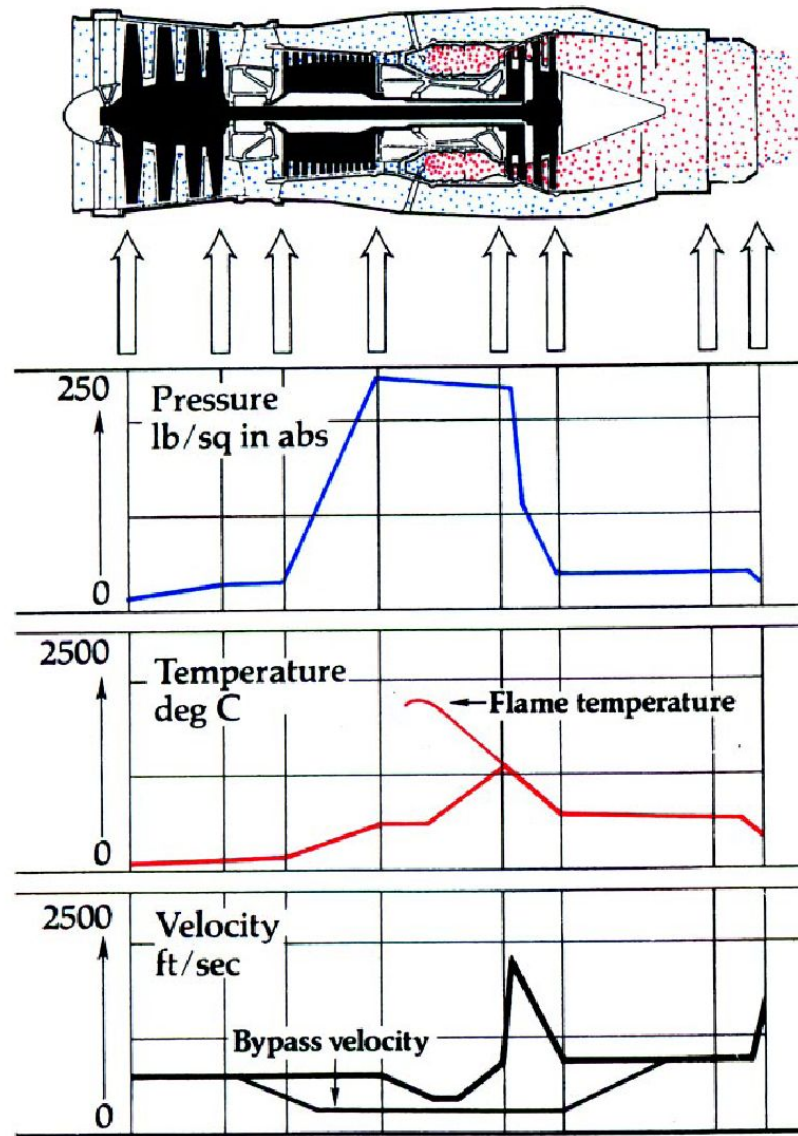
GAS TURBINE OPERATION



GAS TURBINE OPERATION



GAS TURBINE OPERATION



GAS TURBINE OPERATION

JET ENGINE TYPES

Different types of Jet Engines

Different types: -

Turbo Jet

Avon, Olympus 593

Bypass Engine

Small Bypass

Spey, BR700

Medium Bypass

Pegasus, BR715, V2500

Large Bypass

RB211, Trent Series

Turbo Shaft

Propeller

Dart, Tyne, BR715TP

Helicopter

Gem

RN Ships

Marine Spey

Power Generation

Industrial Spey and Trent

To understand what is meant by 'Turbo Jet', 'Bypass', 'Turbo Prop and 'Turbo Shaft'

We need to look at the development of Gas Turbine Engines

JET ENGINE TYPES

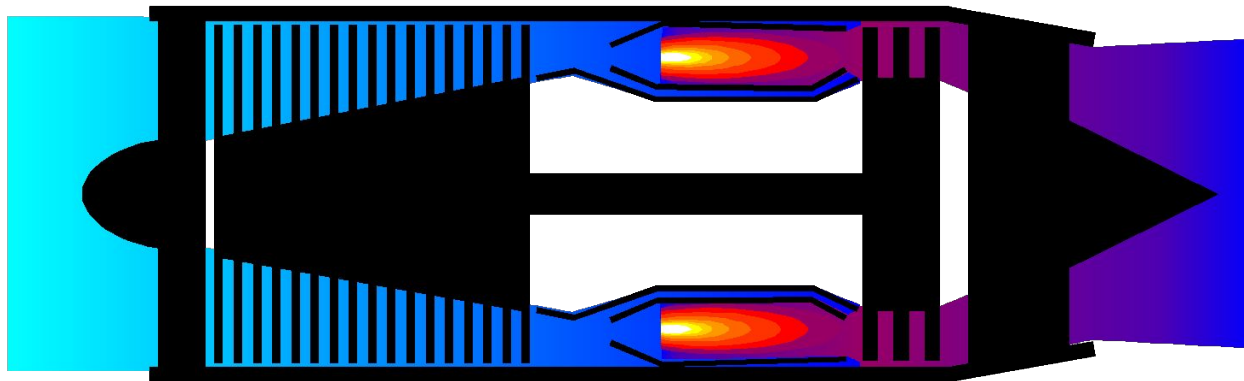
With RPM (I.e. Thrust) changes, the airflow velocity does not change in proportion to engine RPM

At low rpm – rear of compressor cannot get rid of all the air from the front of the compressor

Rolls-Royce AVON

Basic Shape and Layout

Turbo Jet



17 Stage Compressor

driven by: -

a 3 Stage turbine

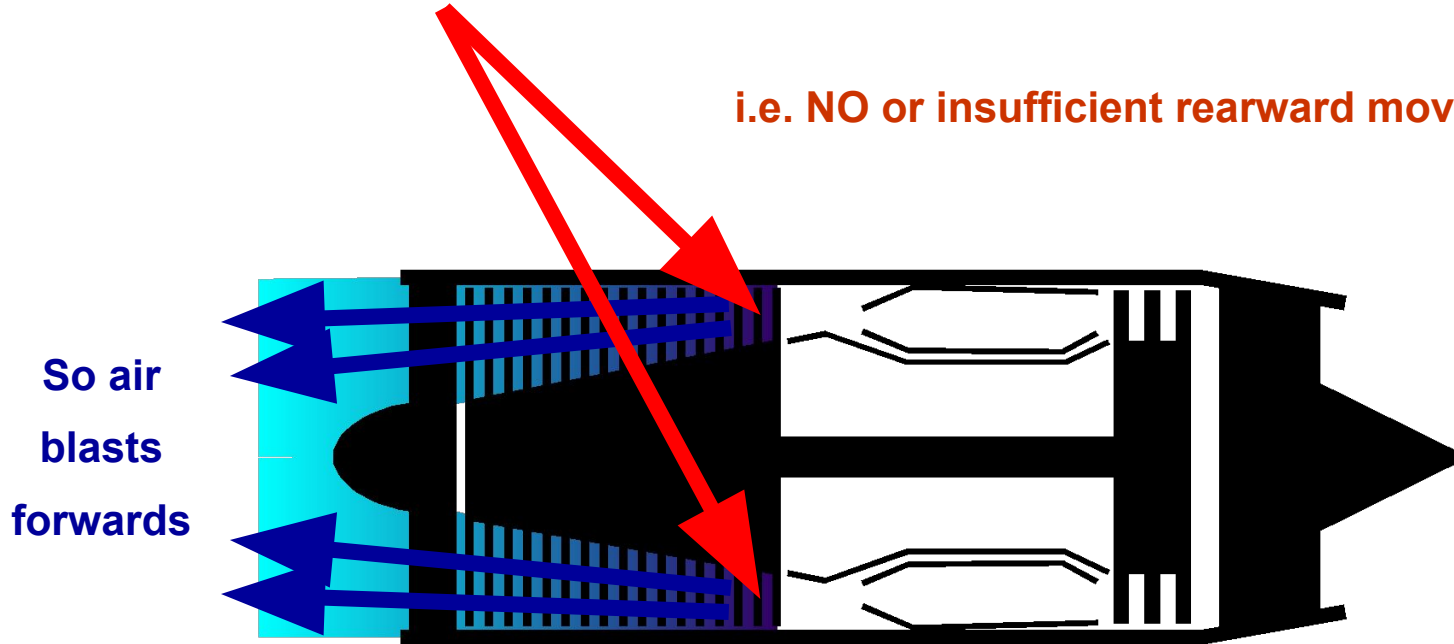
JET ENGINE TYPES

With RPM (I.e. Thrust) changes, the airflow velocity does not change in proportion to engine RPM

At low rpm – rear of compressor cannot get rid of all the air from the front of the compressor

This can result in the rear end of compressor ‘choking’

i.e. NO or insufficient rearward movement of air

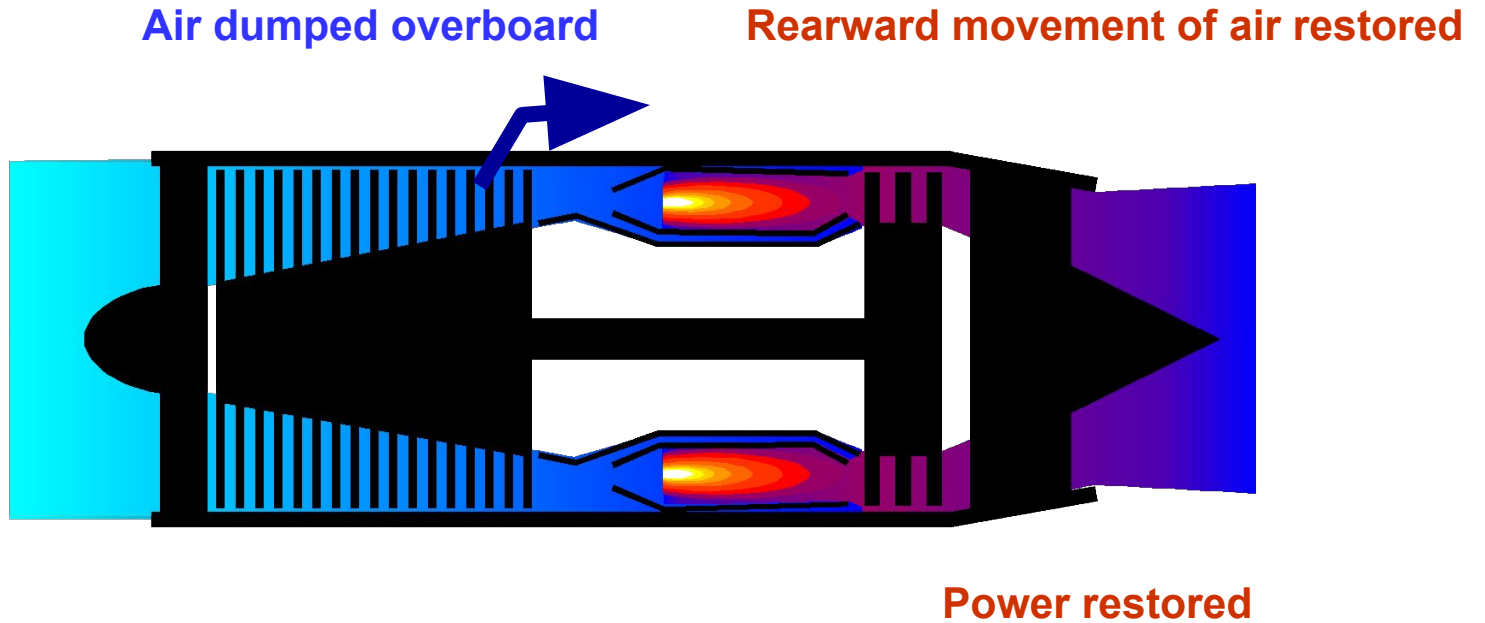


Flame almost goes out -
no power – no thrust

This is called: - Compressor SURGE

JET ENGINE TYPES

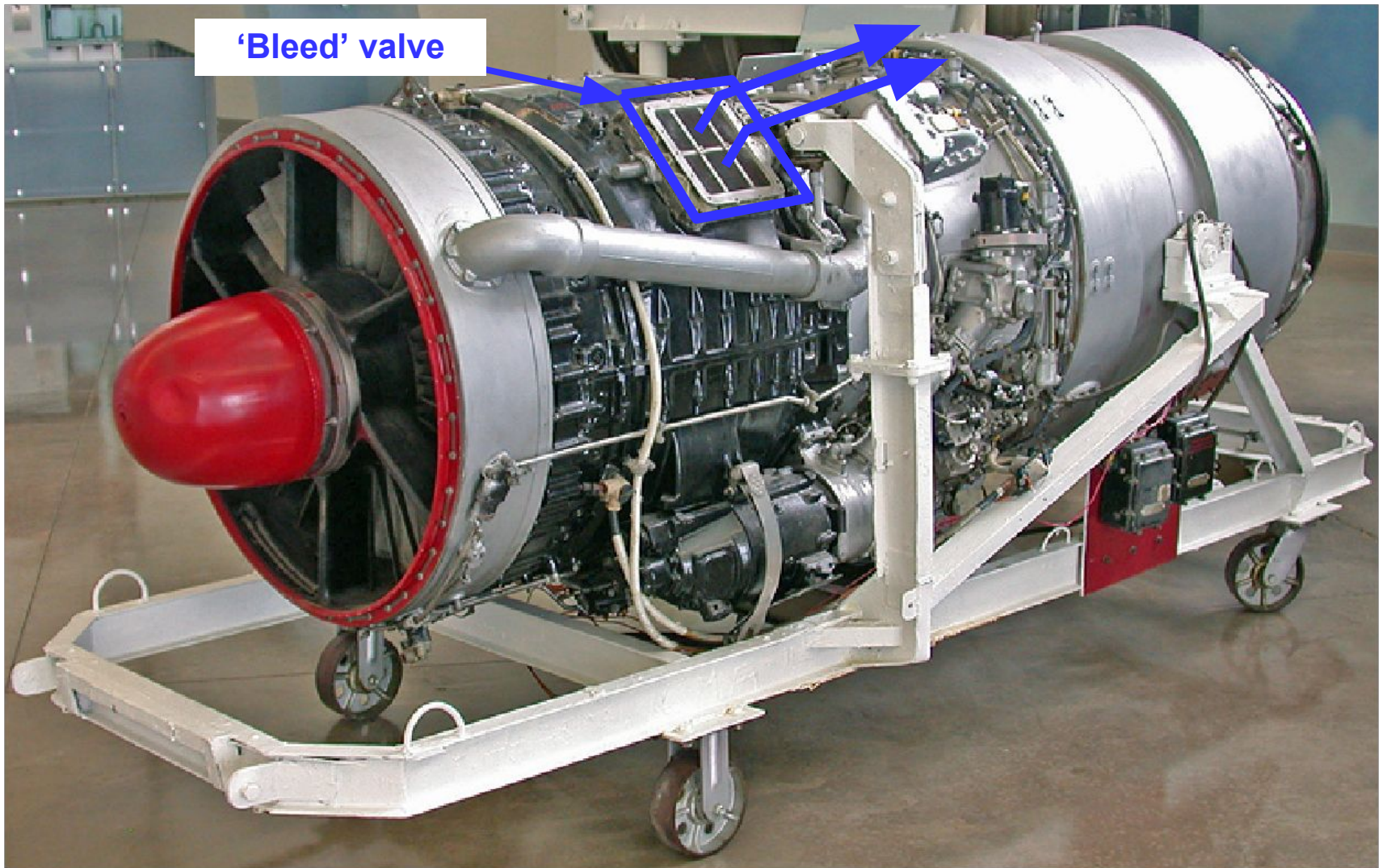
Rear end of compressor 'unchoked'



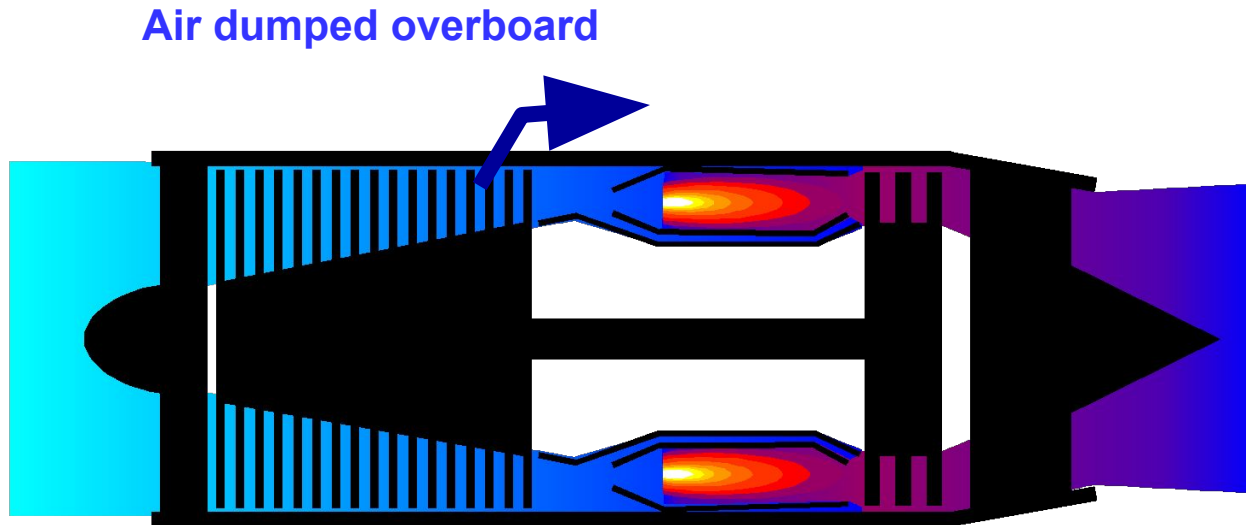
BUT: - air dumped sideways out of the engine (aircraft) produces no thrust

JET ENGINE TYPES

'Excess' air is dumped overboard through a valve



JET ENGINE TYPES



Air dumped overboard reduces thrust/wastes fuel

So a more efficient system was required

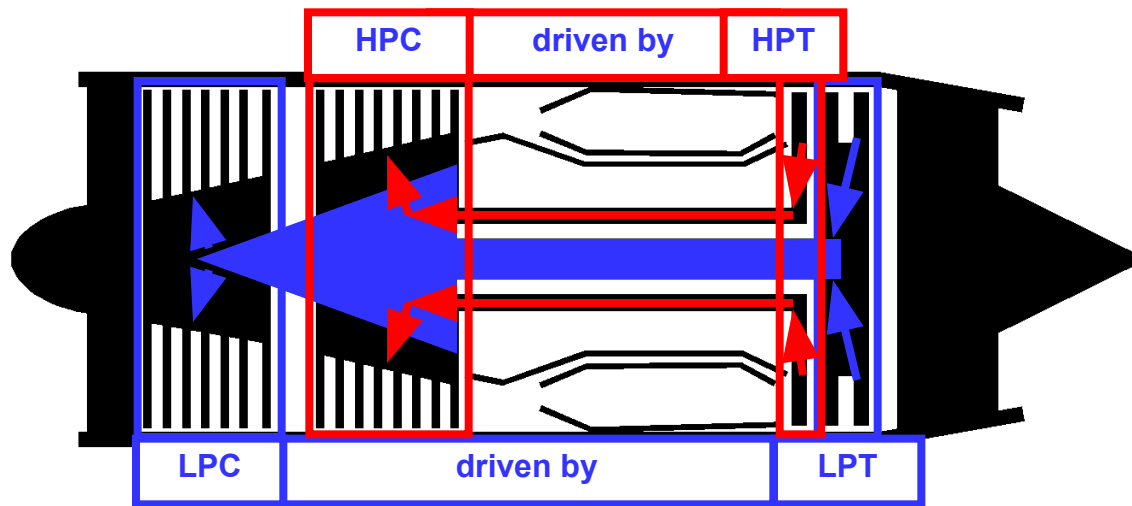
Enter - Multiple shaft engines and the Bypass system

JET ENGINE TYPES

Allows front end of compressor to slow down – move less air

AND

Allows rear end of compressor to speed up – move more air



Two shaft layout

Compressor surges lessened

No wasted energy/thrust

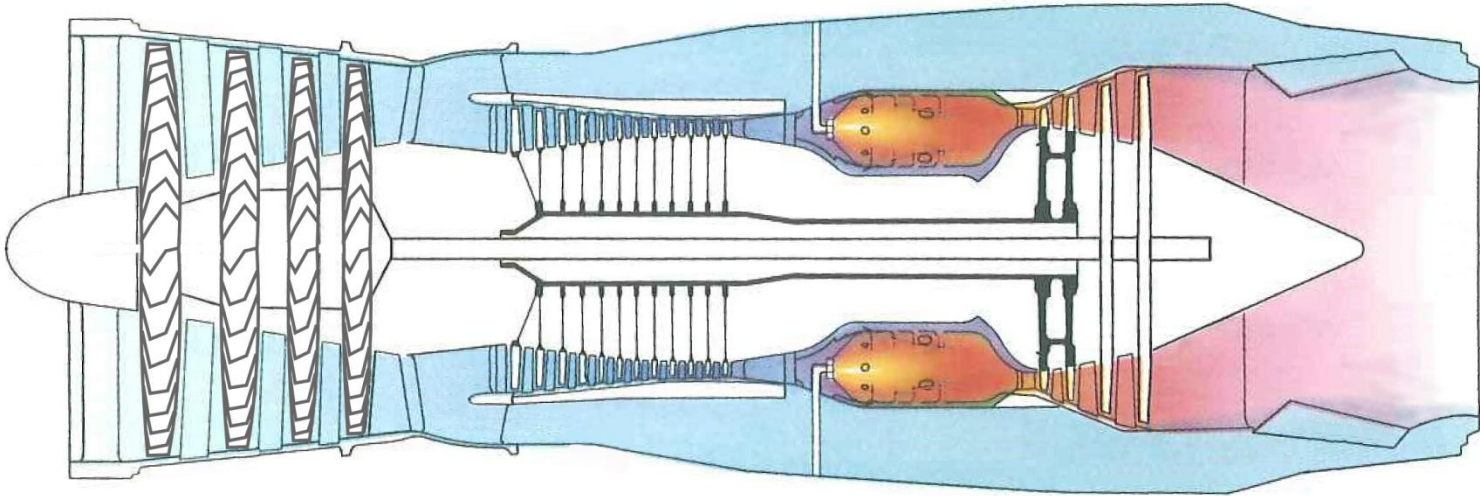
JET ENGINE TYPES

Bypass allows air not required in the HPC to flow around the outside of it

Rolls-Royce Spey

Adour

BR 710 series



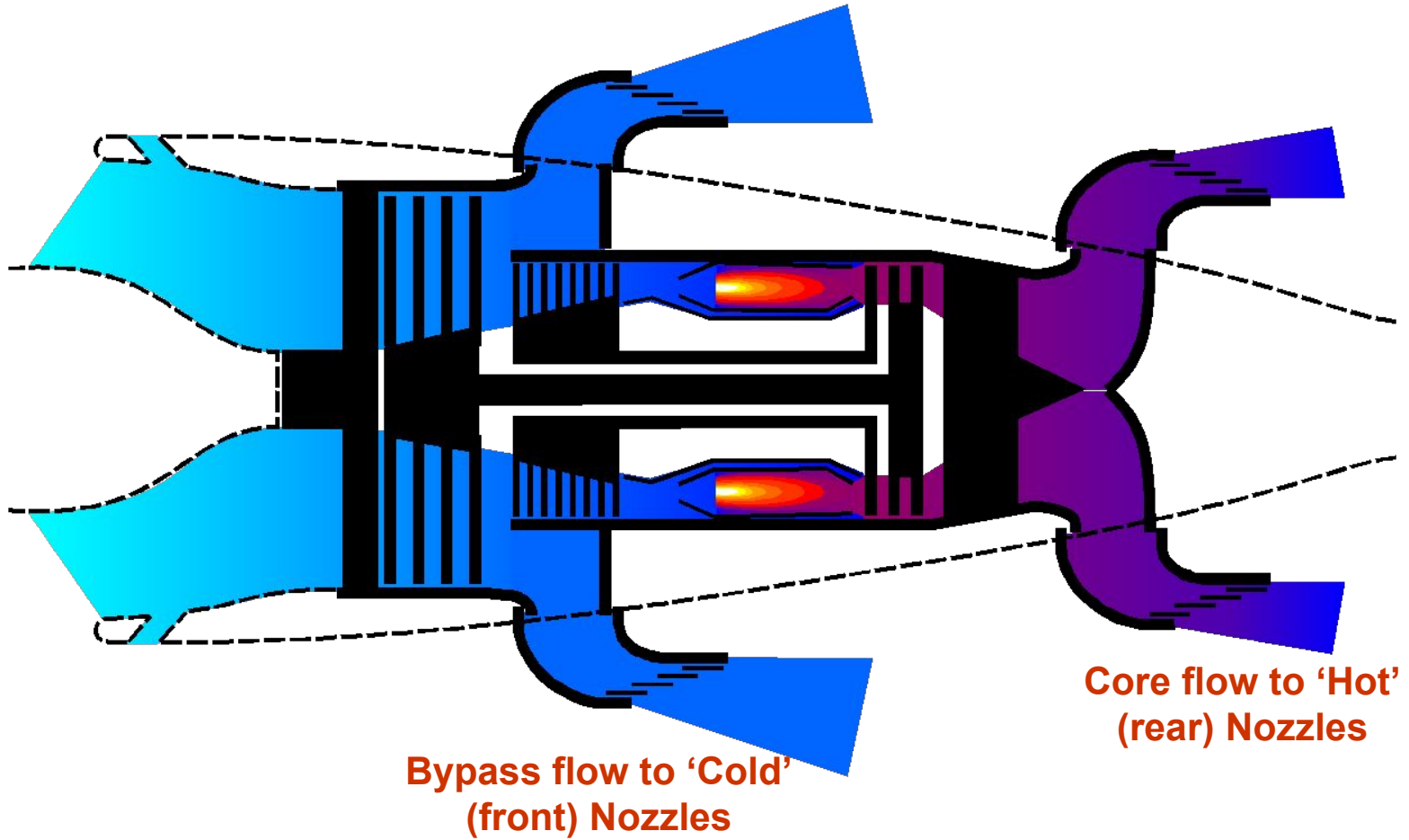
Bypass Layout

Low Bypass

JET ENGINE TYPES

Pegasus

Two shaft layout – Contra-rotation – Medium Bypass
LP and HP AXIAL Flow Compressors
In RAF Service – Harrier series



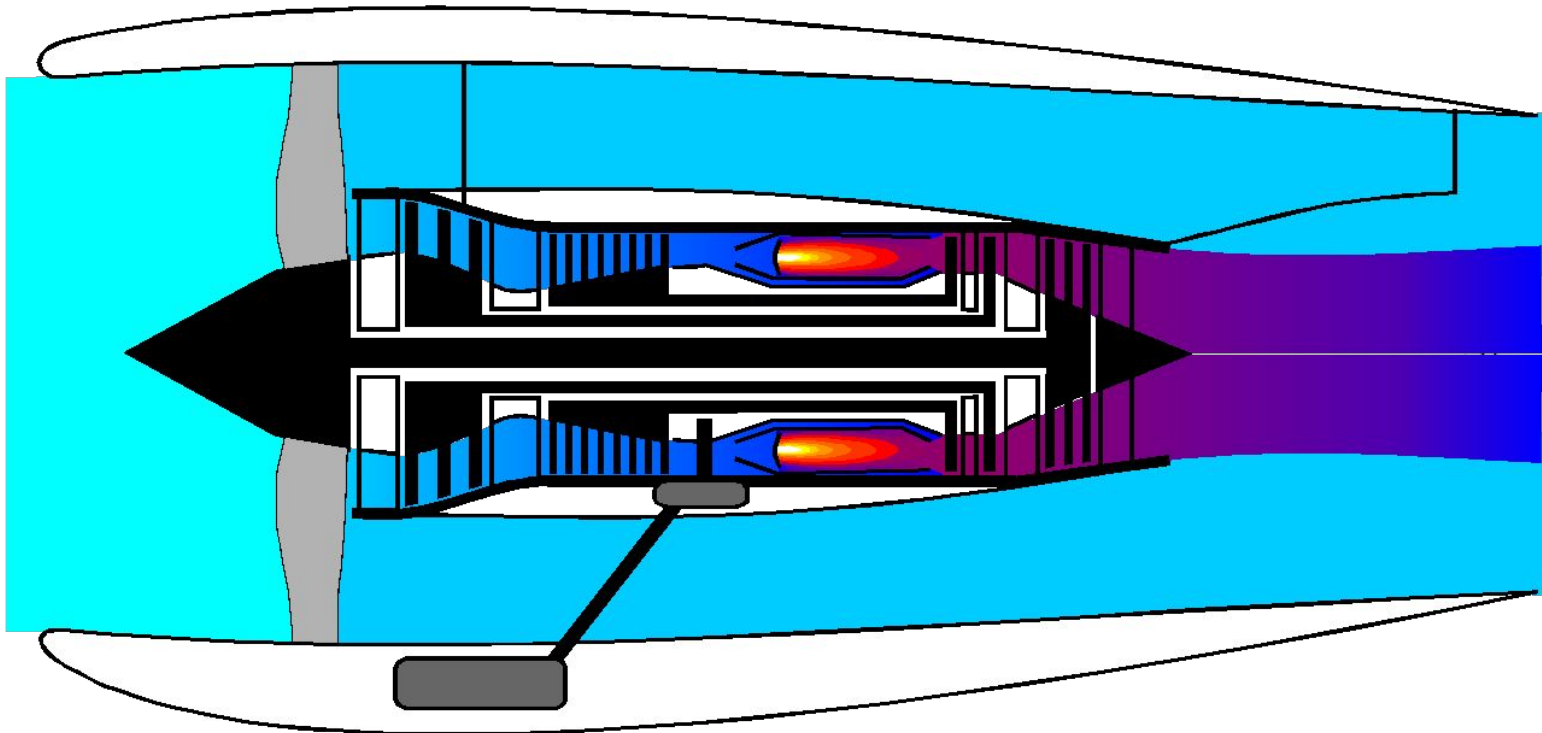
JET ENGINE TYPES

High Bypass Engine

Three shaft layout – all rotate same direction

RB211 Series (Trent)

LP(Fan) IP and HP AXIAL Flow Compressors



In service with RAF – Wide-body Tristar Tanker/Transport

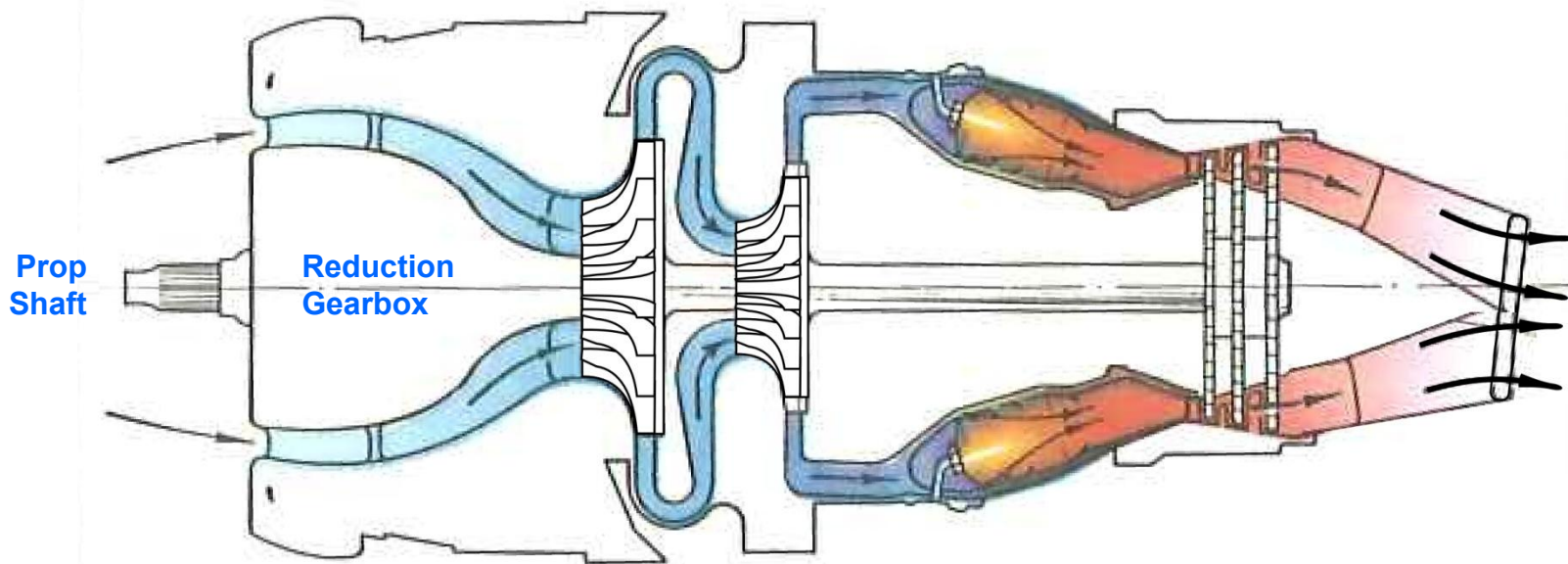
Boeing 747 and later series, Airbus A330 and larger a/c

JET ENGINE TYPES



JET ENGINE TYPES

RR Dart Turbo Prop



Dart RDa6 = 2 stage turbine

Dart RDa7 = 3 stage turbine

Turbine drives 2 stage CENTRIFUGAL Compressor and Propeller

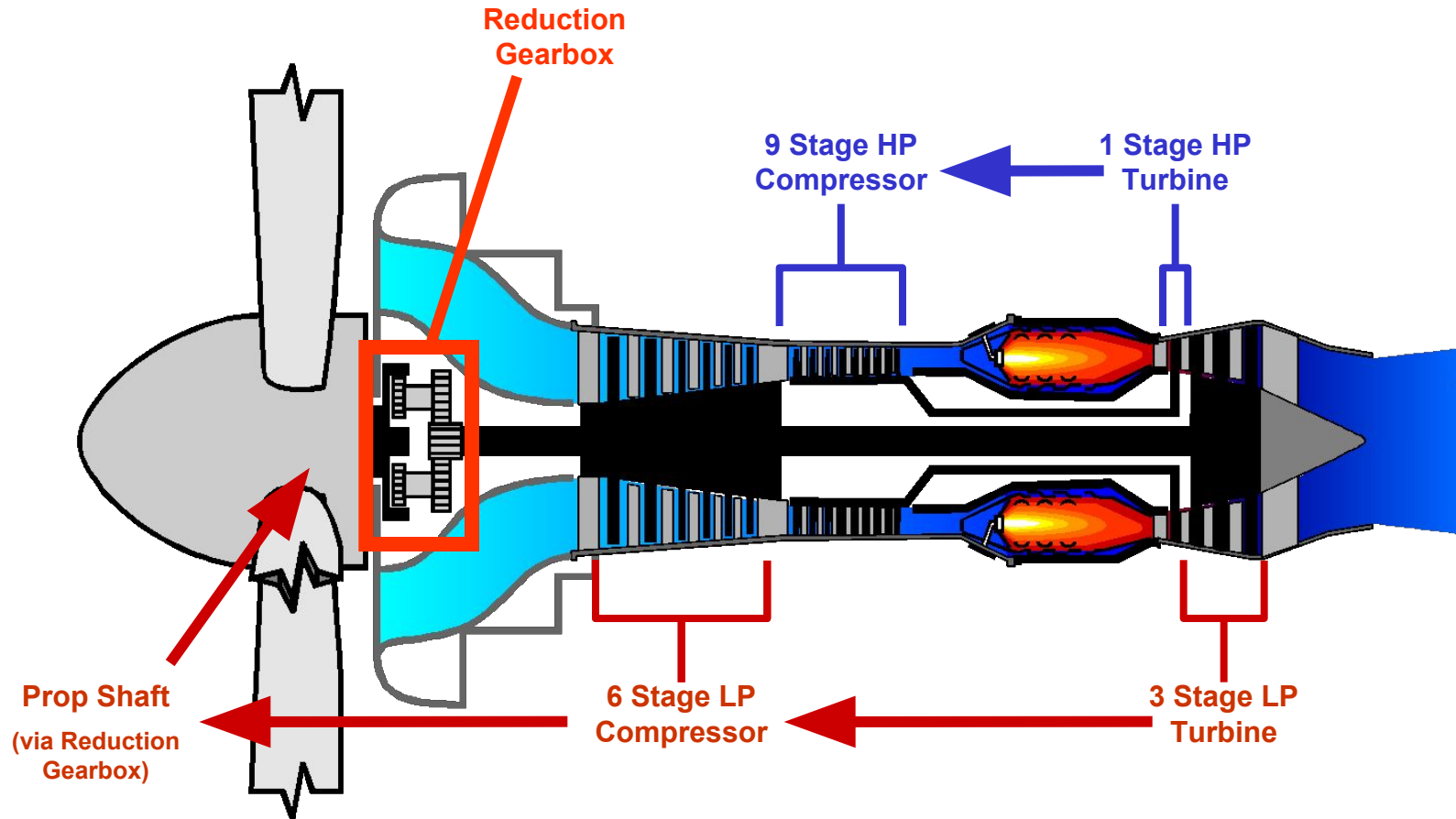
Over 70 different Marks of Engine – basically different power levels – 15 different aircraft

Was in service with RAF – ‘Kneeling’ Andover Transport (A BAe 748 with higher tail and rear loading ramp)

Fokker F27, Fairchild F27, Brigue Alize, (and P51 Mustang and Dakota!)

JET ENGINE TYPES

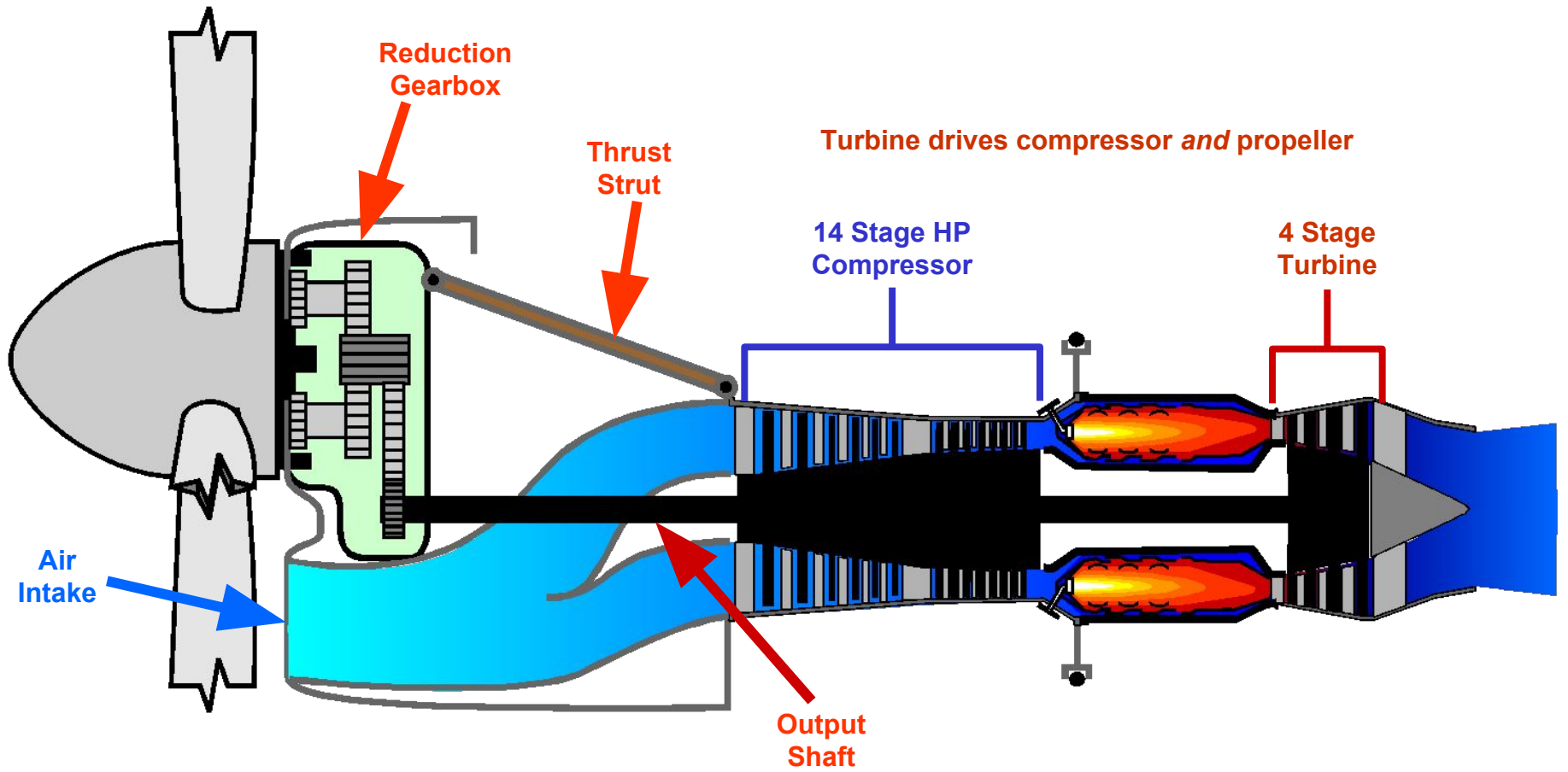
Tyne Turbo Prop



C160 Transall, Shorts Belfast, Brequet Atlantic, Avro Lincoln, Canadair CL44, Vickers Vanguard, Aeritalia 222.

JET ENGINE TYPES

RR Allison T56-A-15 Turbo Prop

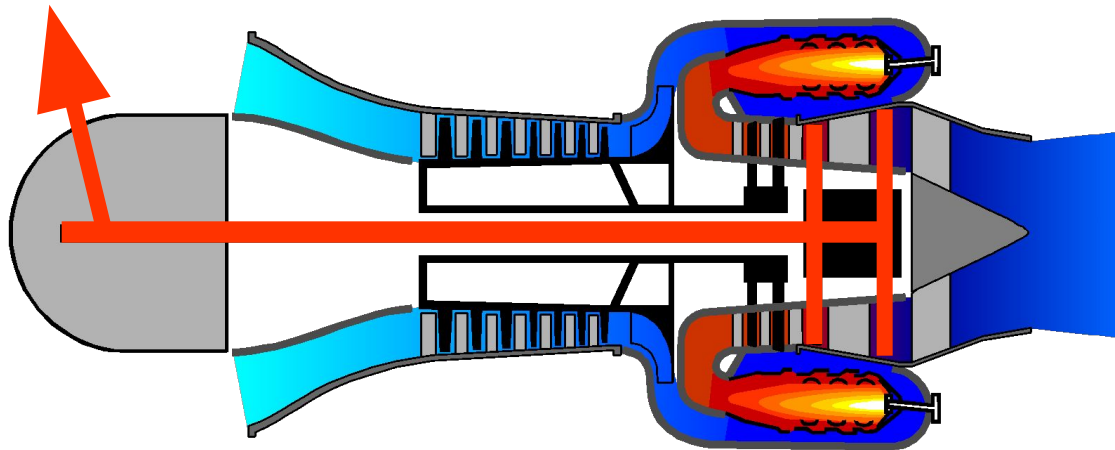
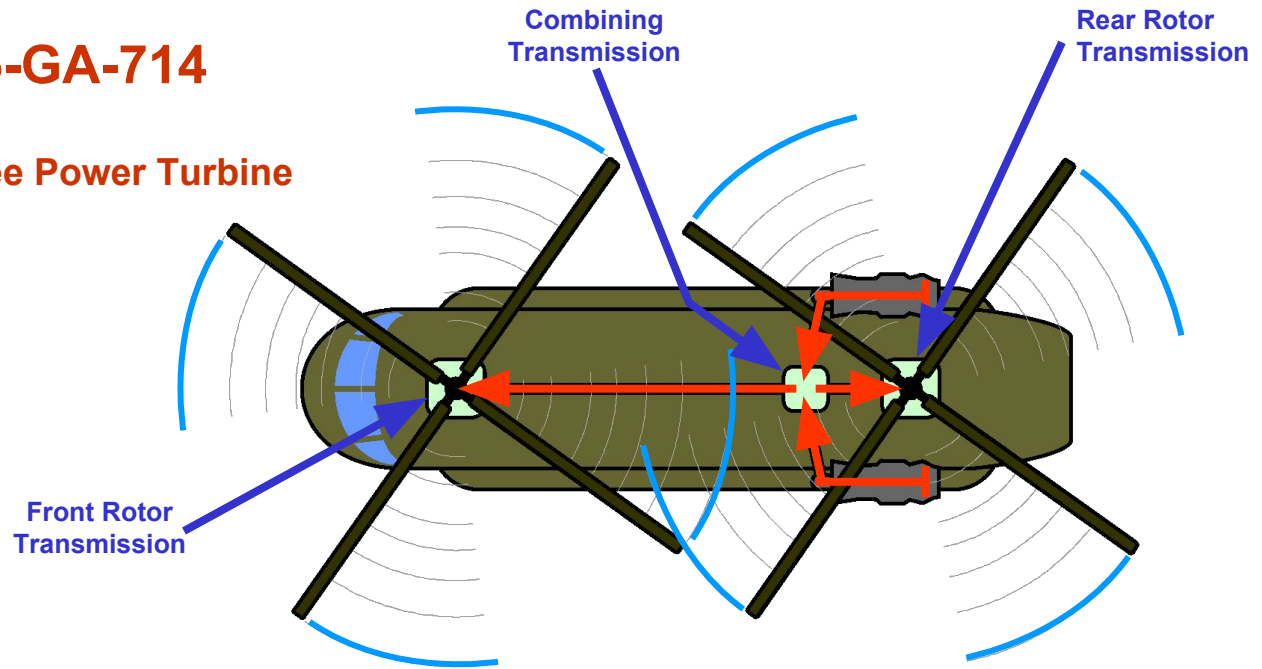


C130 Hercules.

JET ENGINE TYPES

Lycoming T55-GA-714

Turboshaft with Free Power Turbine



Boeing CH47 Chinook

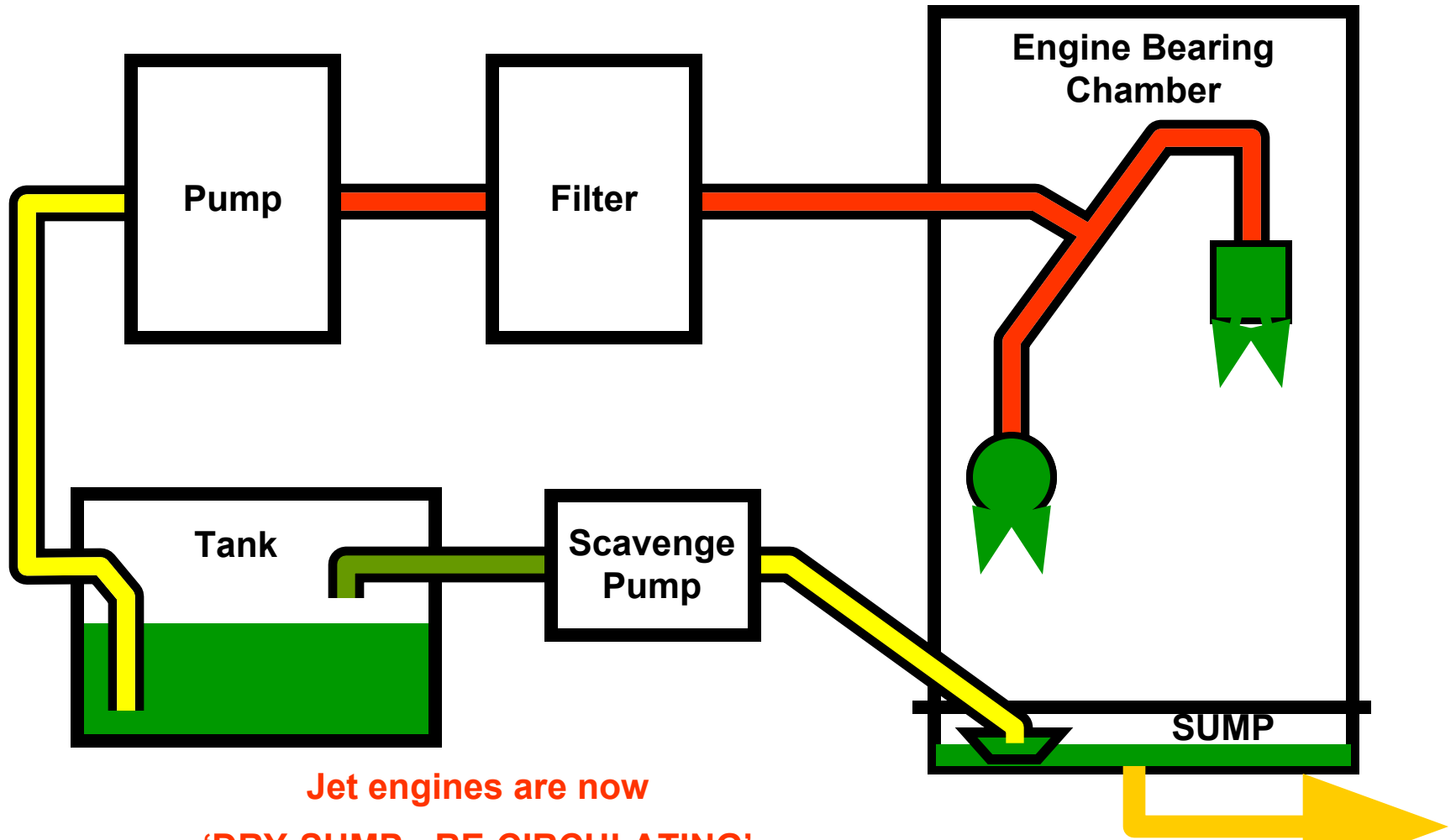
JET ENGINE TYPES

TYPICAL OIL SYSTEMS

Jet Engines

Distribution and Lubrication

Early jet engines were total loss system



Jet engines are now
'DRY-SUMP - RE-CIRCULATING'
Oil systems

i.e. oil was lost in the gas stream

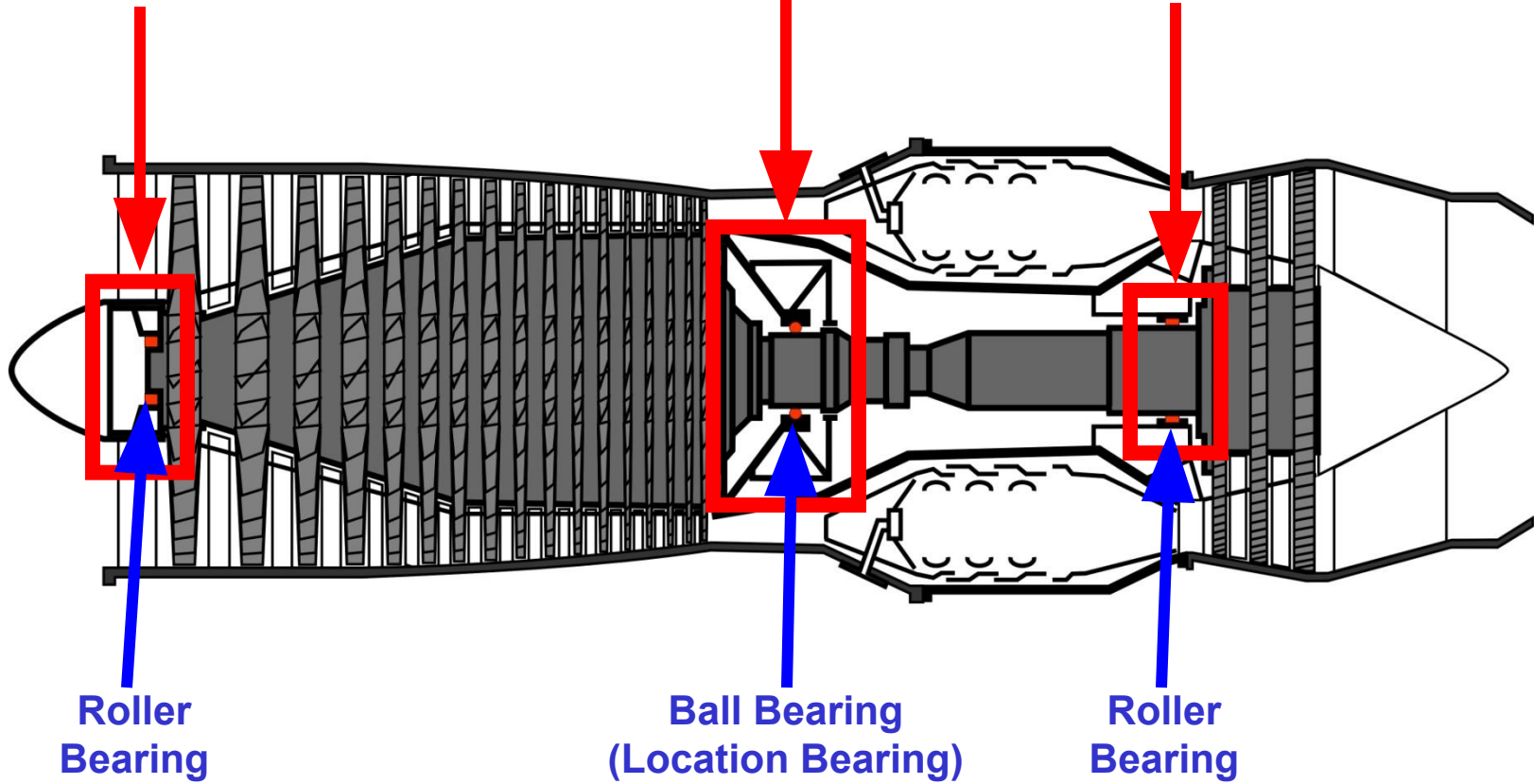
TYPICAL OIL SYSTEM - JET ENGINE

Bearing Chambers

Front Bearing Chamber

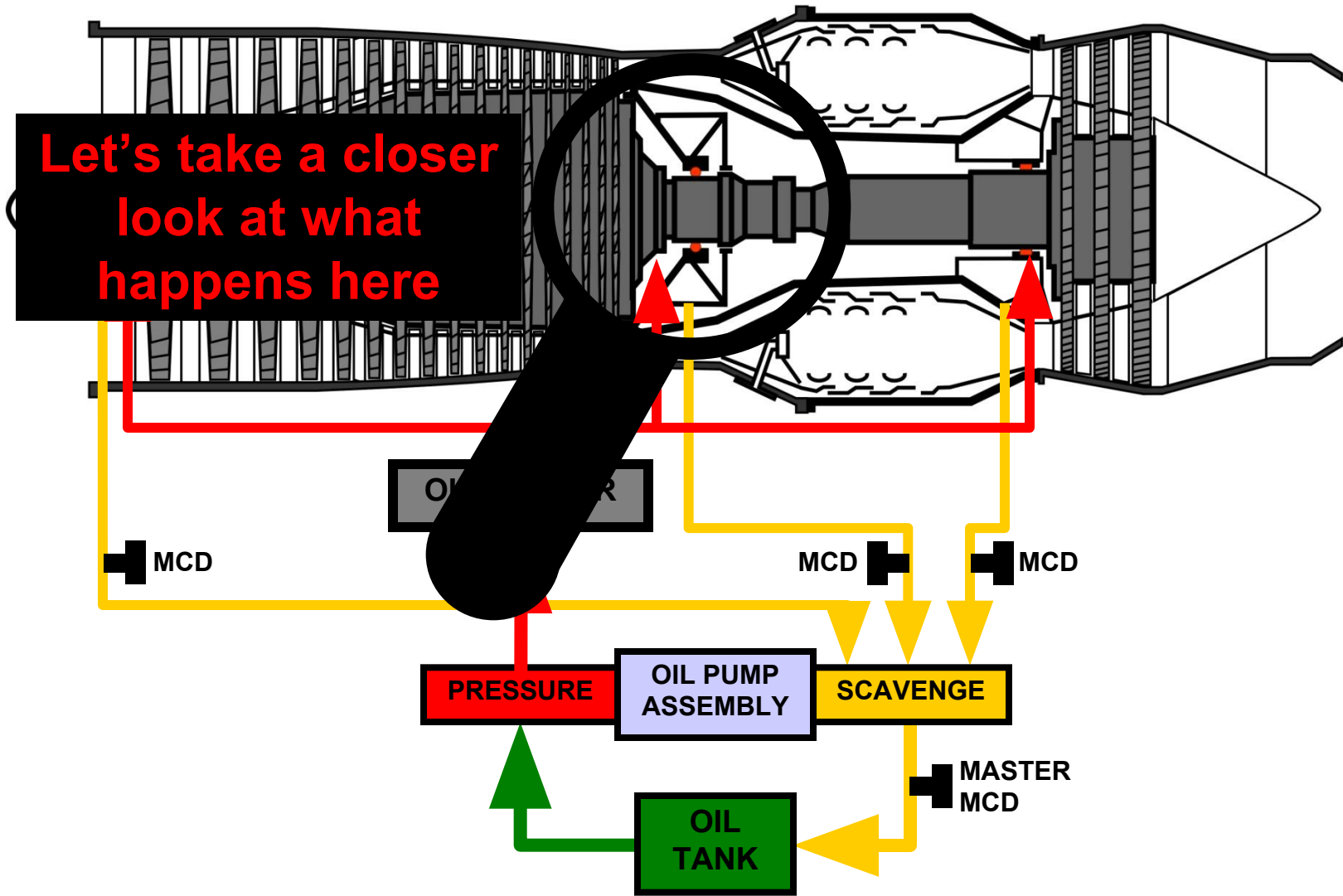
Centre Bearing Chamber

Rear Bearing Chamber



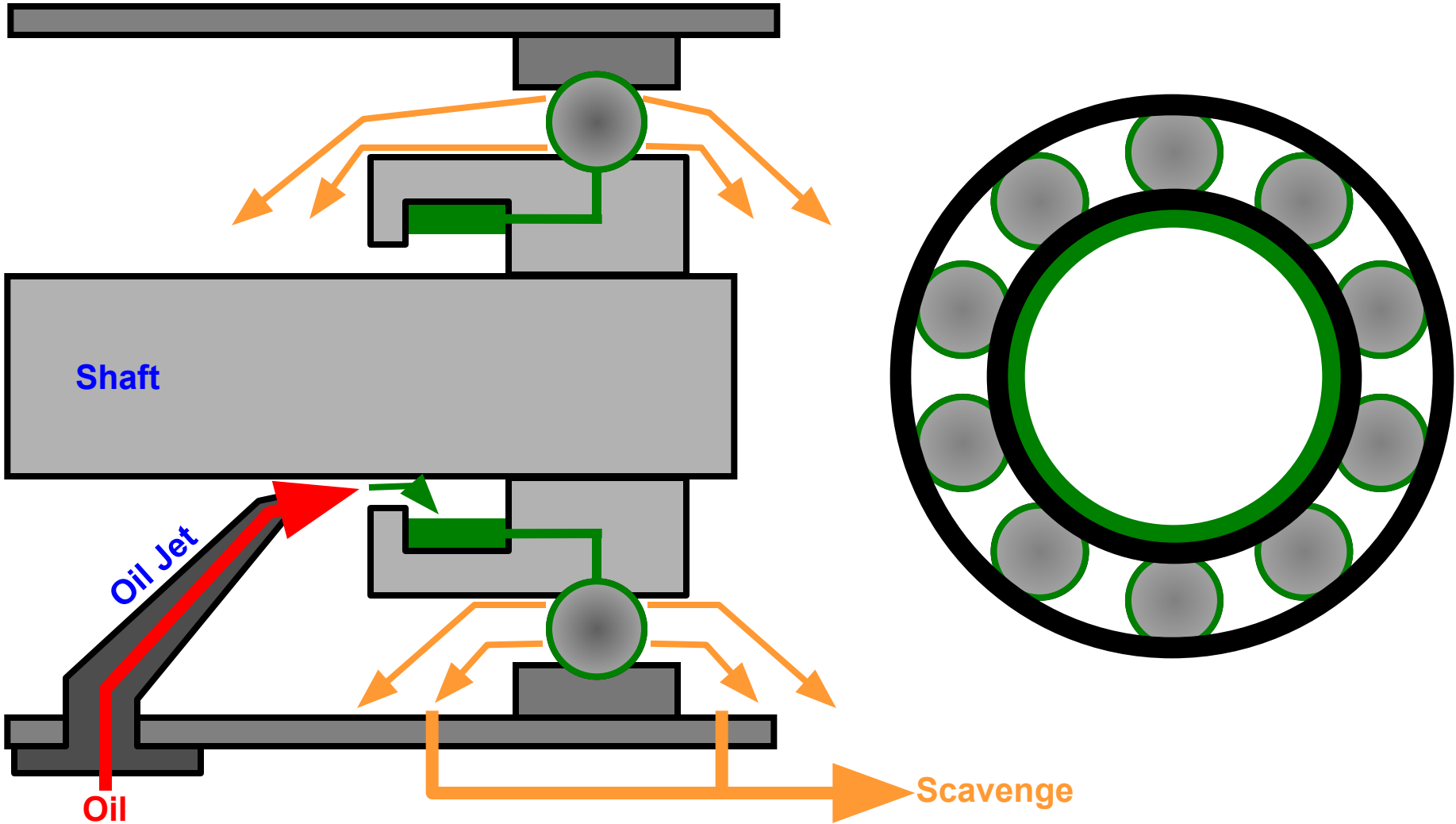
TYPICAL OIL SYSTEM - JET ENGINE

Notional oil system



TYPICAL OIL SYSTEM - JET ENGINE

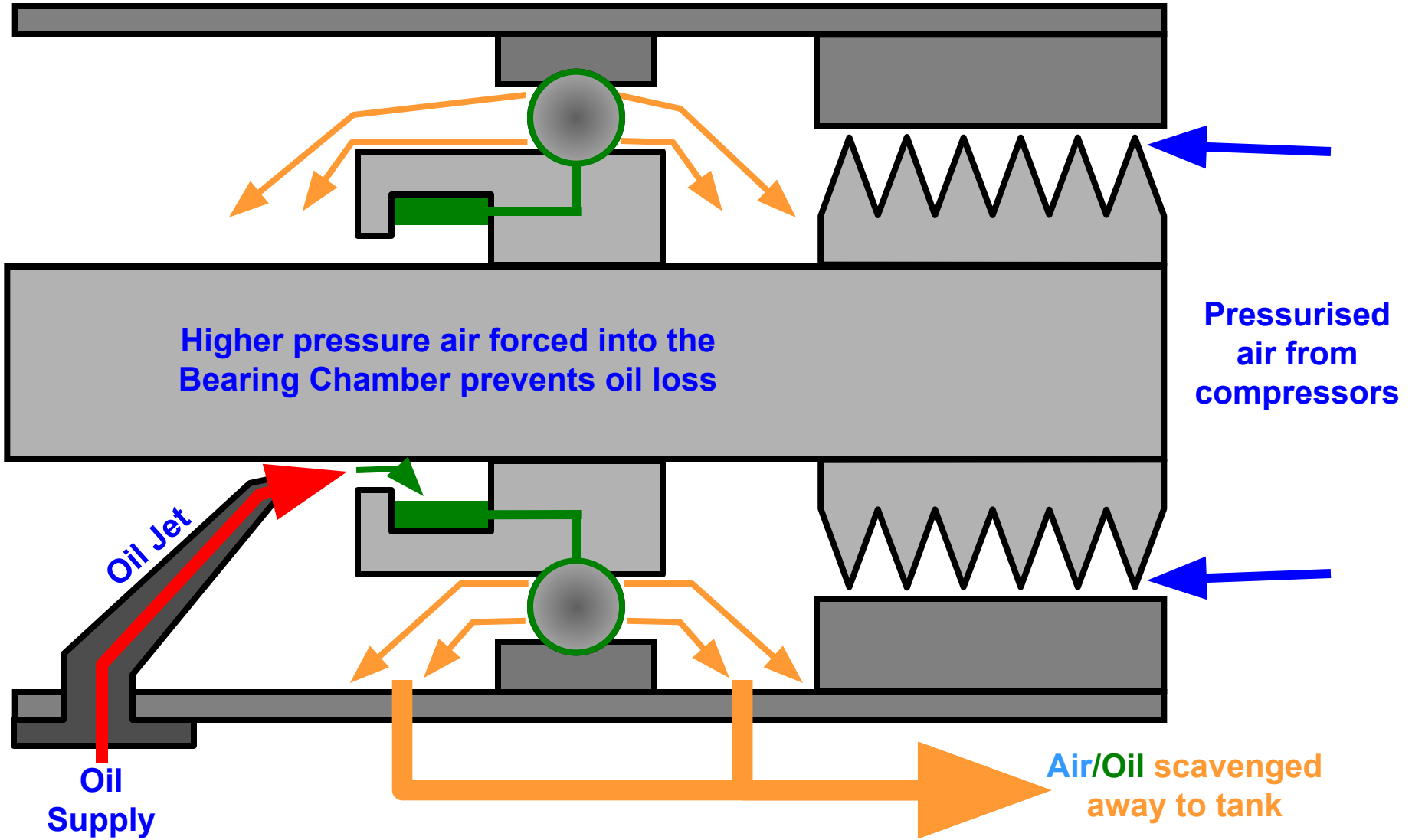
Bearing Support Structure



Bearing Oil Supply and Scavenge

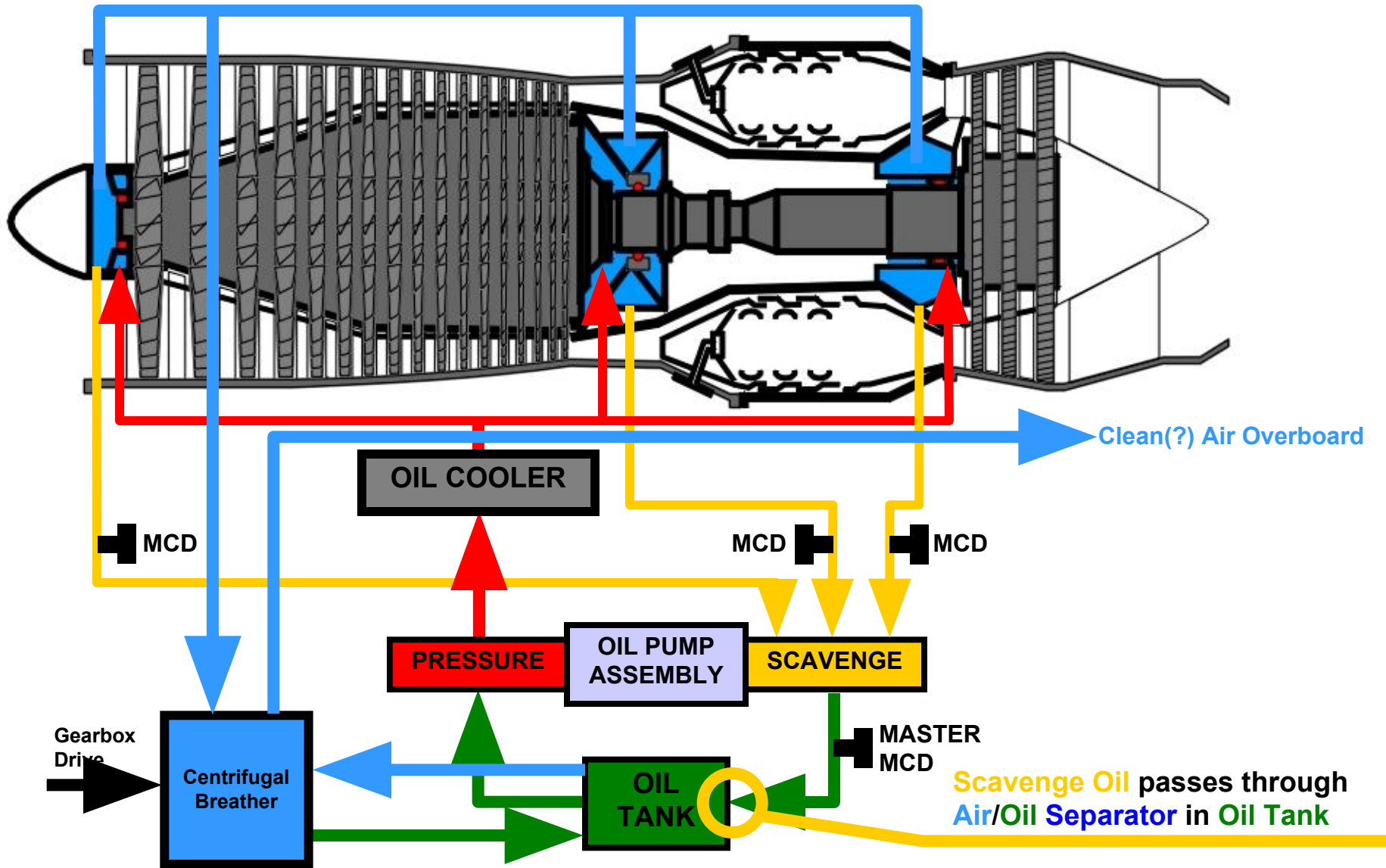
TYPICAL OIL SYSTEM - JET ENGINE

Bearing Support Structure = static part



TYPICAL OIL SYSTEM - JET ENGINE

Notional oil system with Air Sealing

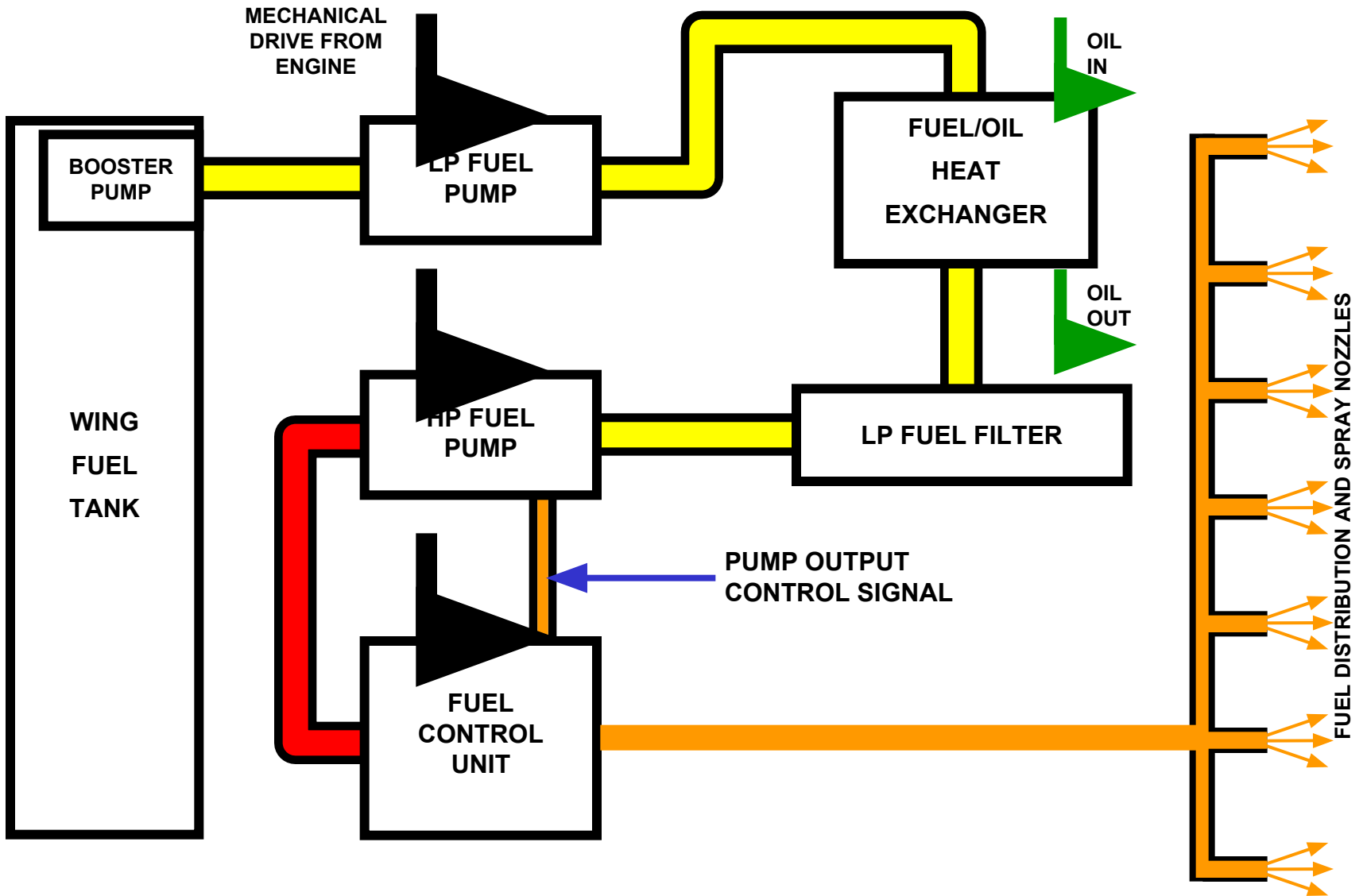


TYPICAL OIL SYSTEM - JET ENGINE

TYPICAL FUEL SYSTEM

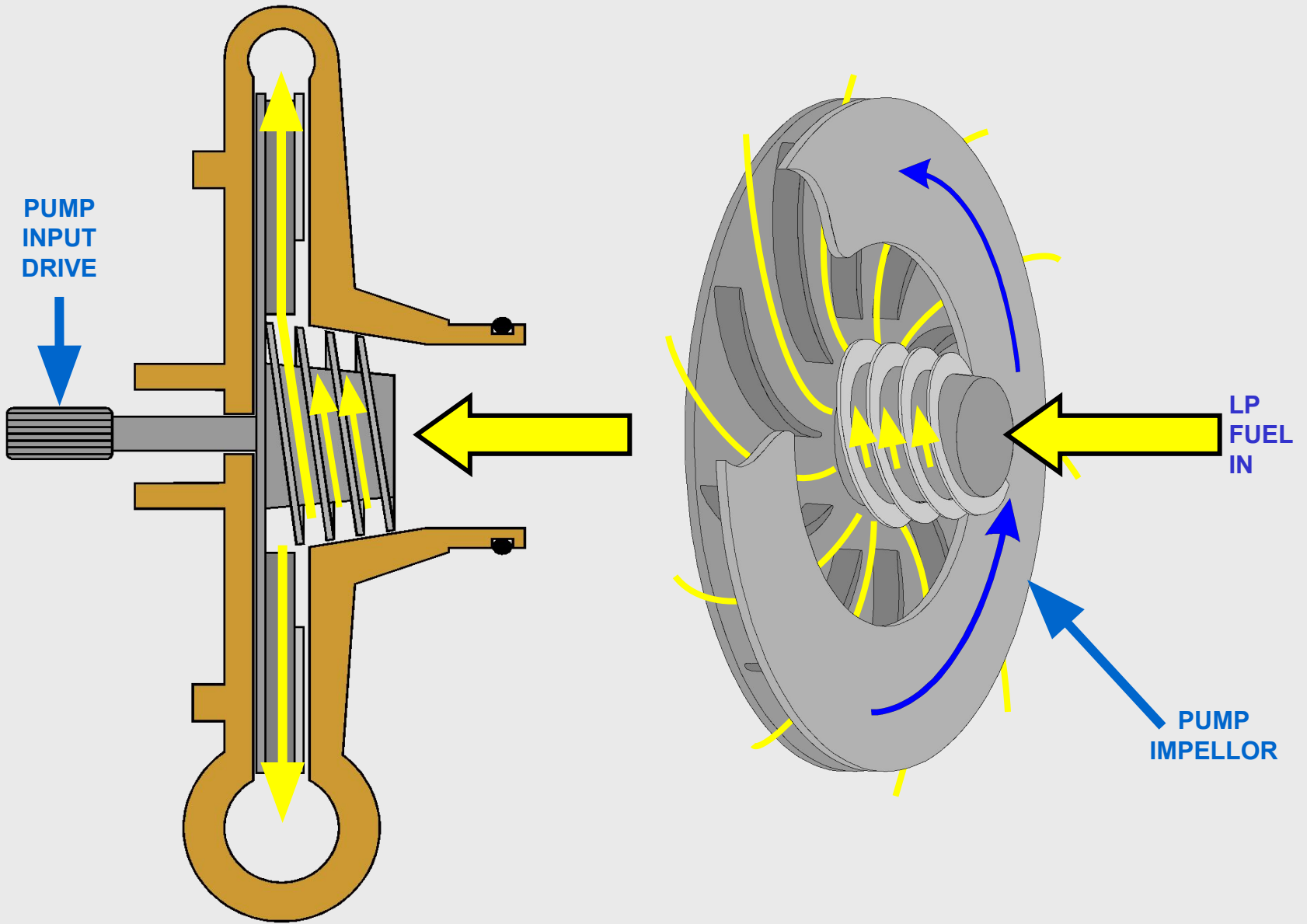
GAS TURBINE ENGINE

1ST – WE WILL LOOK AT HYDROMECHANICAL SYSTEMS



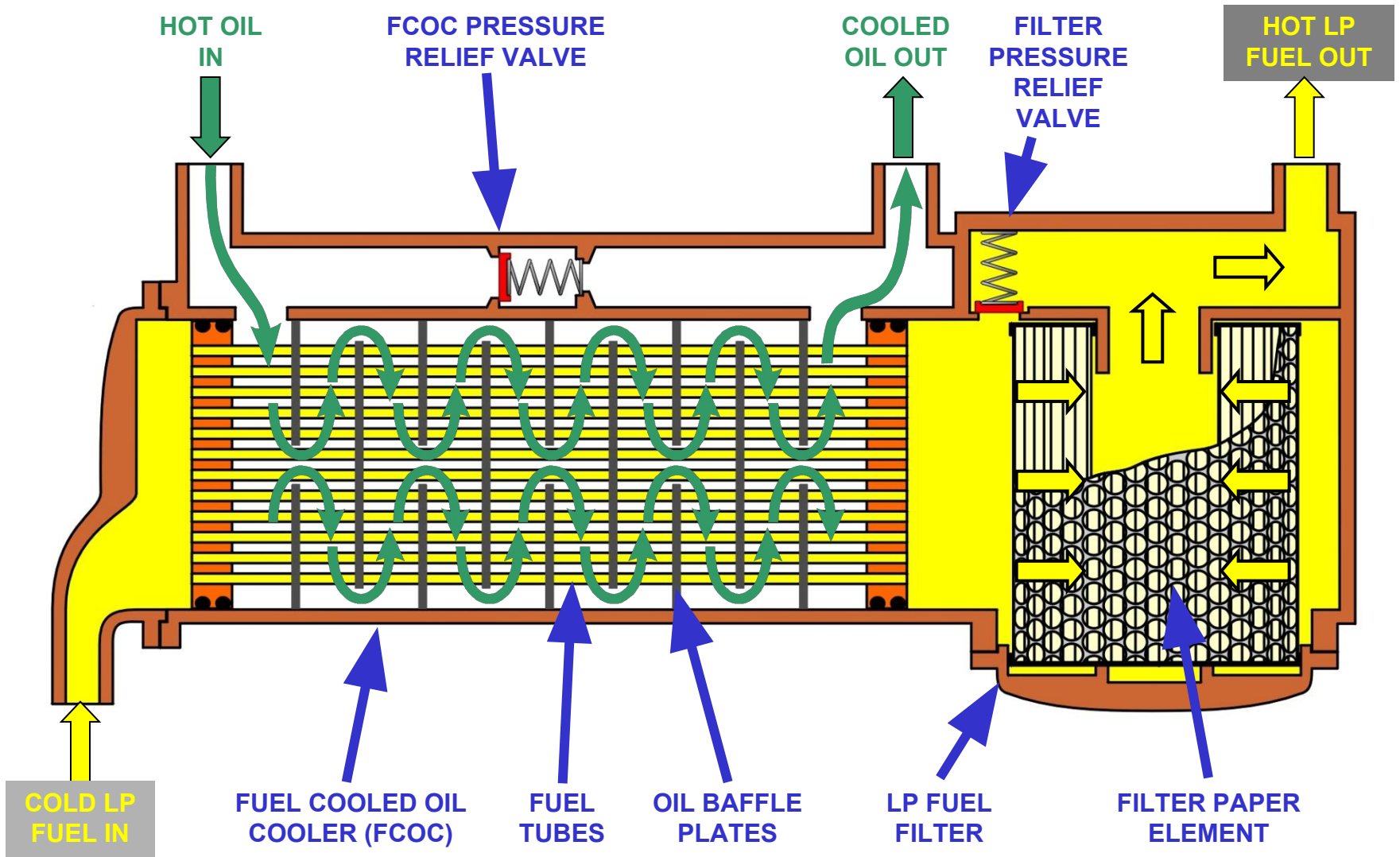
MULTI PLUNGER (SWASH-PLATE) PUMP

TYPICAL FUEL SYSTEM - GAS TURBINE ENGINE



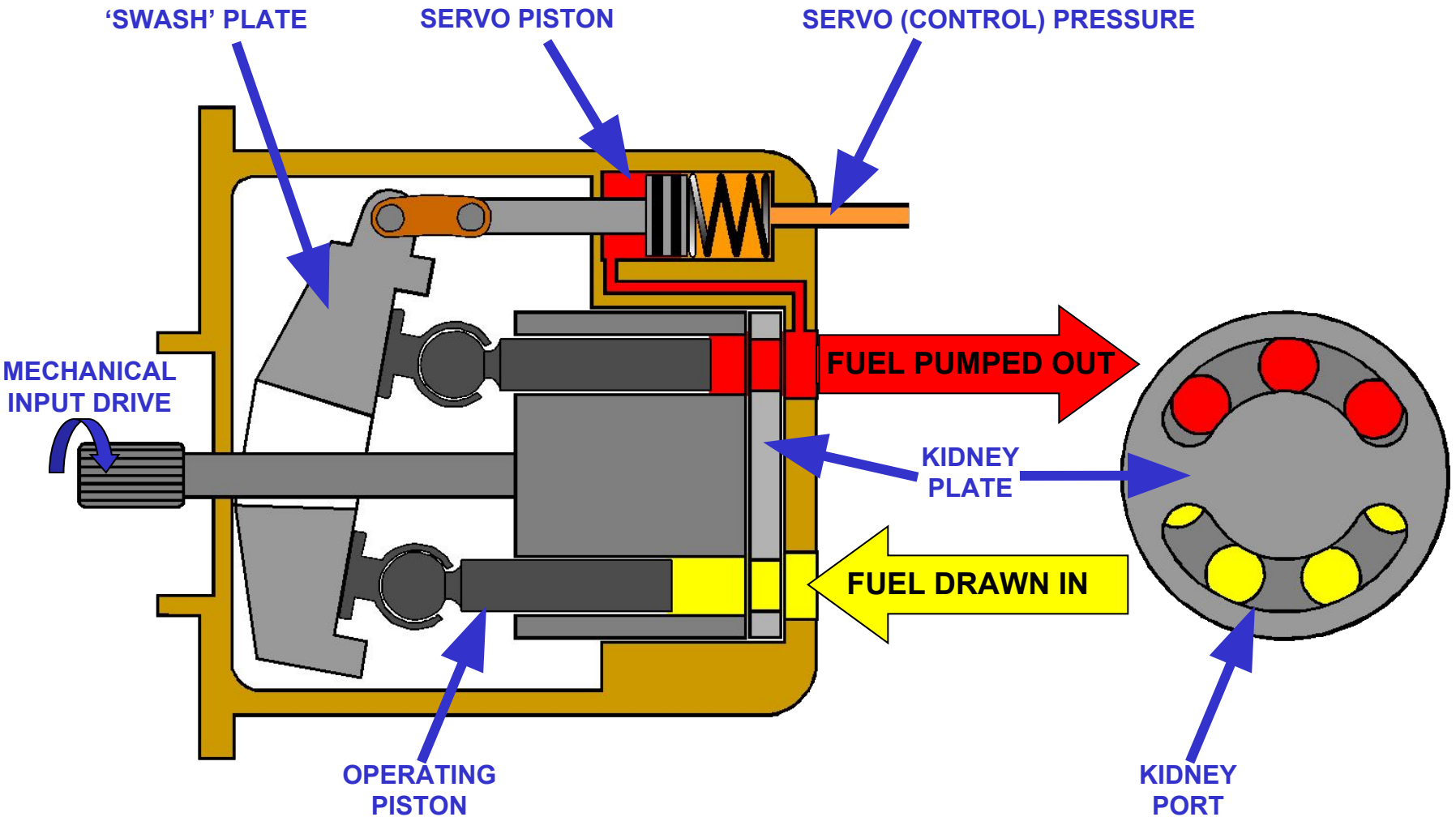
LOW PRESSURE PUMP

TYPICAL FUEL SYSTEM - GAS TURBINE ENGINE



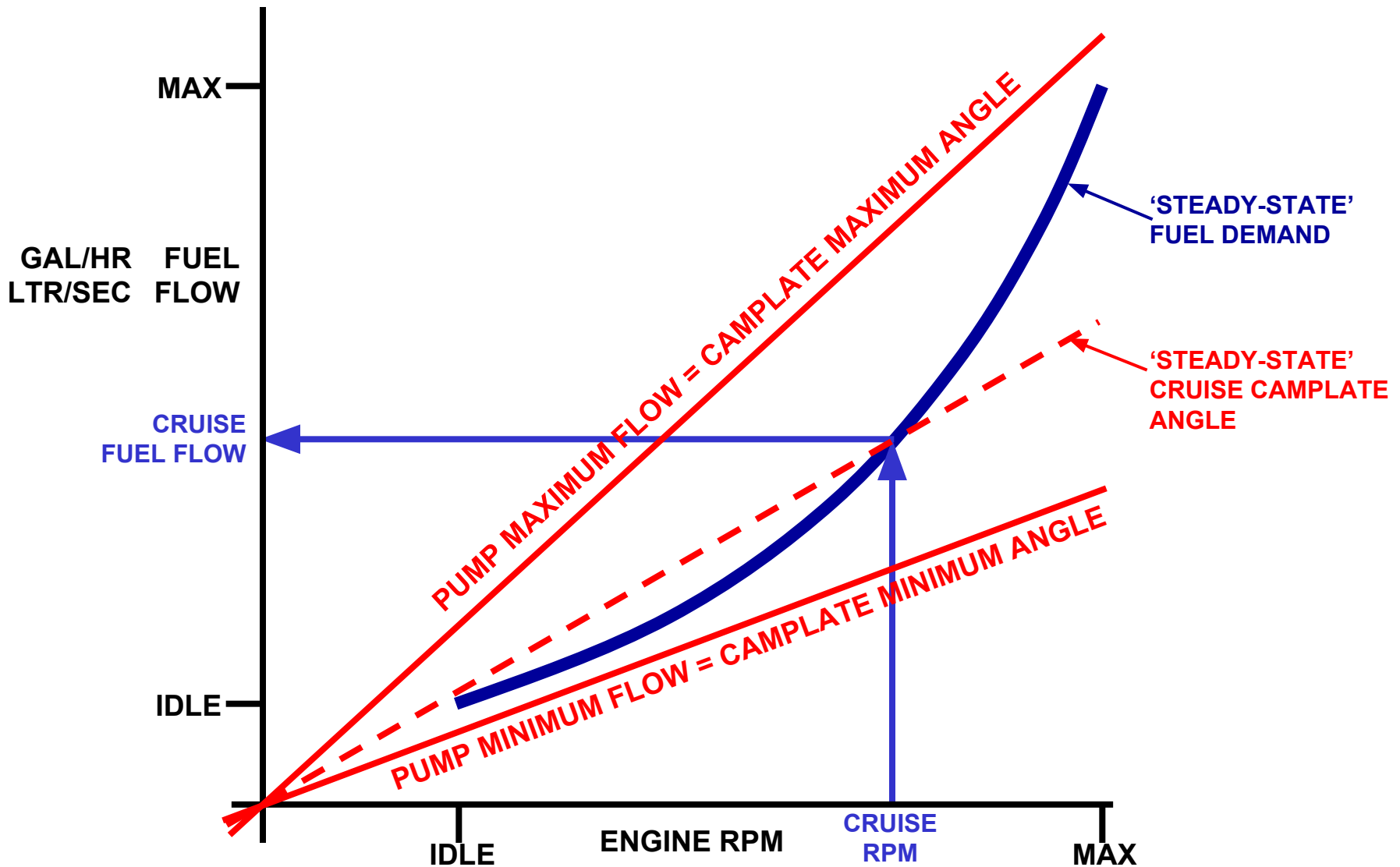
FUEL/OIL HEAT EXCHANGER AND FUEL FILTER

TYPICAL FUEL SYSTEM - GAS TURBINE ENGINE



MULTI PLUNGER (SWASH-PLATE) PUMP

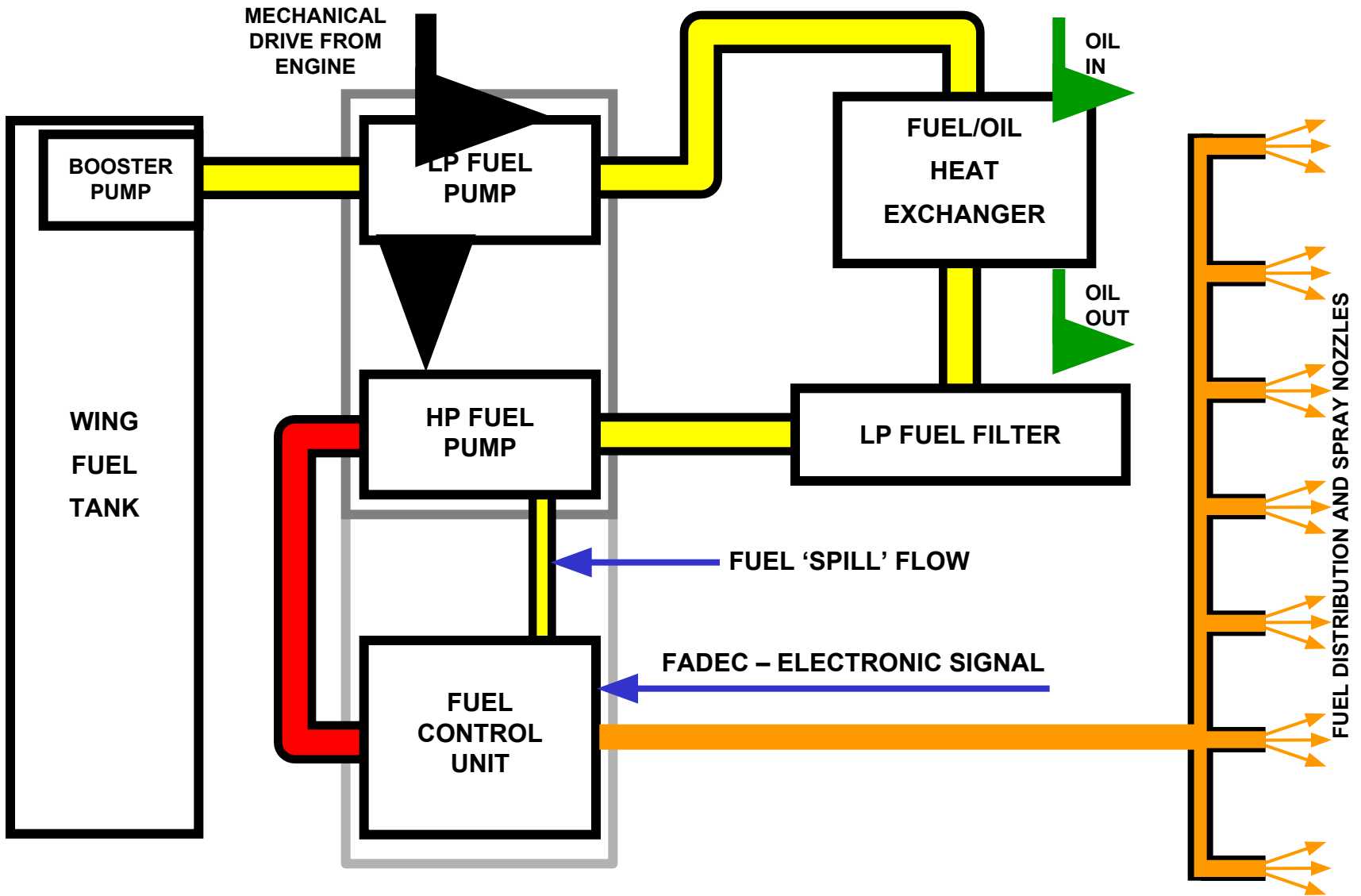
TYPICAL FUEL SYSTEM - GAS TURBINE ENGINE



MULTI PLUNGER (SWASH-PLATE) PUMP

TYPICAL FUEL SYSTEM - GAS TURBINE ENGINE

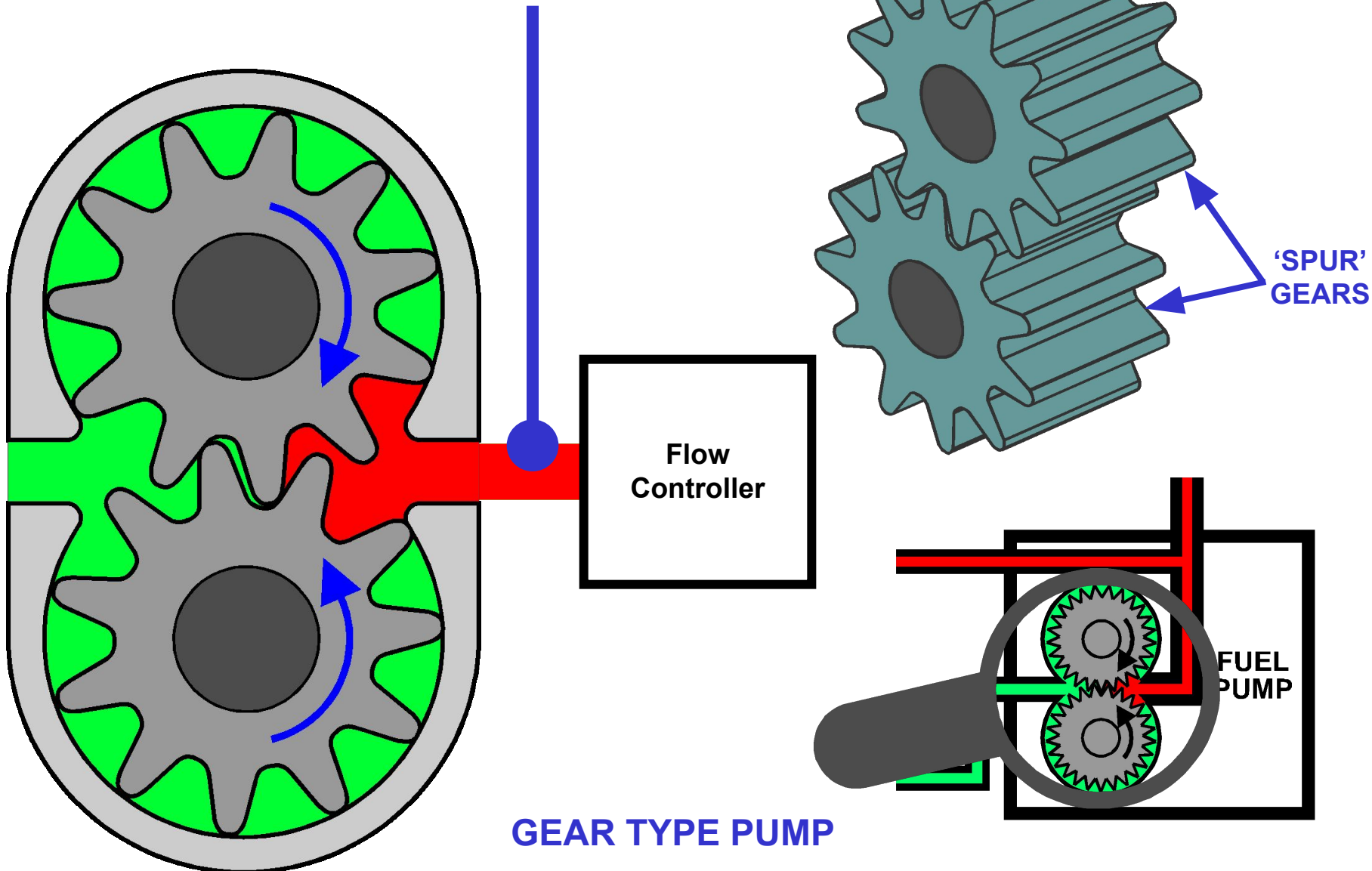
**NOW – WE WILL LOOK AT ELECTRONICALLY CONTROLLED
SYSTEMS**



GEAR TYPE PUMP

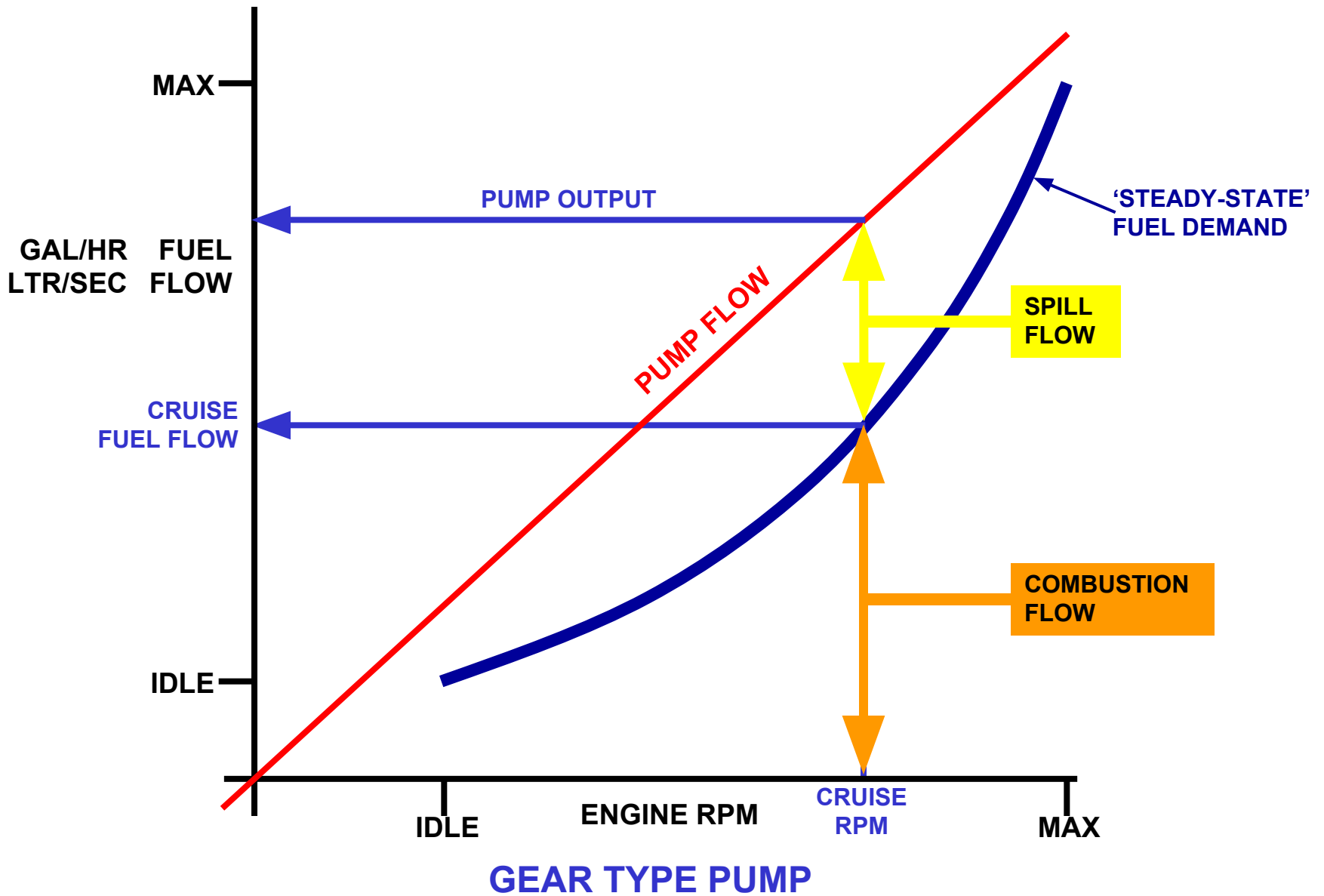
TYPICAL FUEL SYSTEM - GAS TURBINE ENGINE

PUMP FLOW AND RESTRICTION TO FLOW IN CONTROLLER CAUSES PRESSURE TO INCREASE

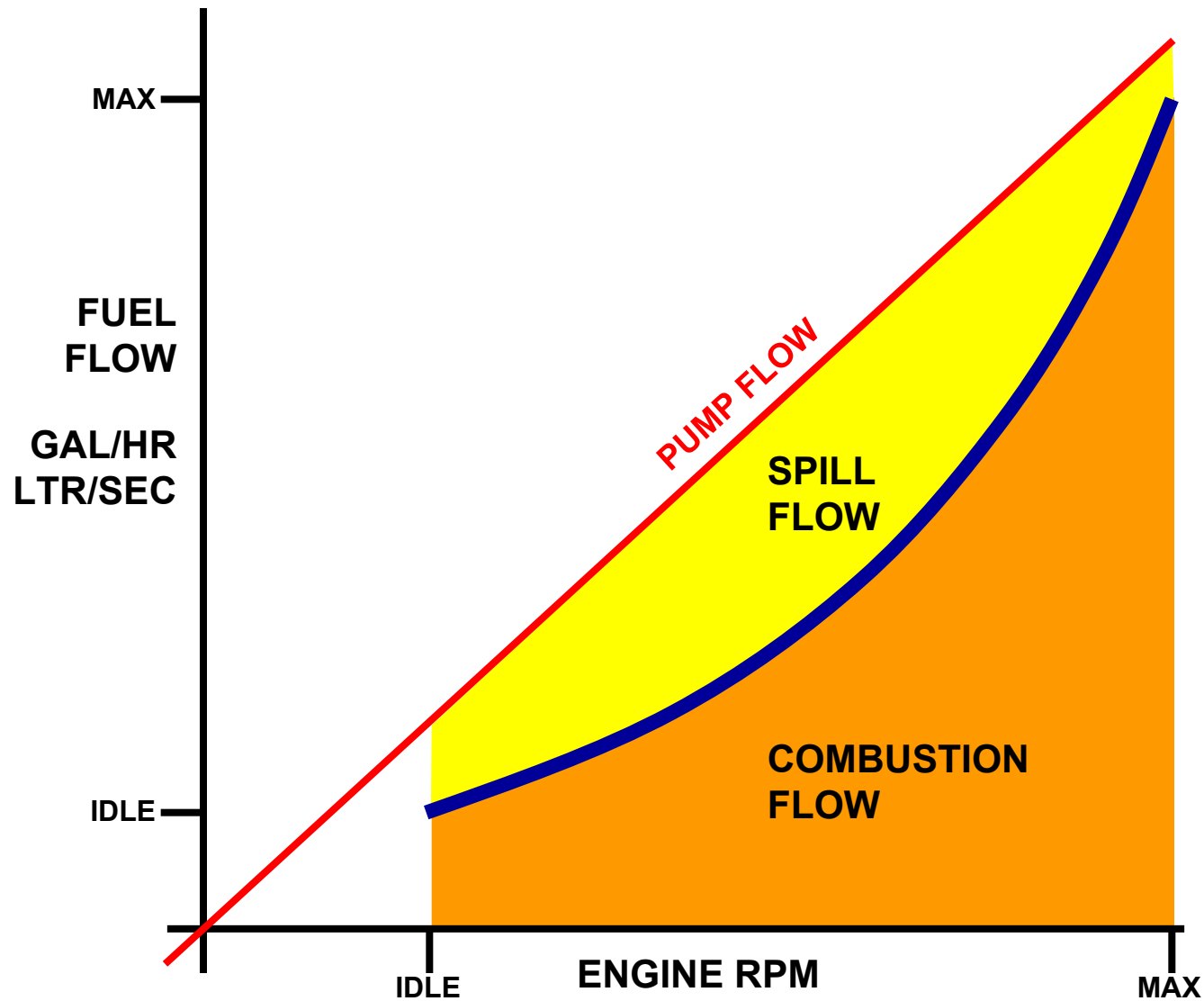


GEAR TYPE PUMP

TYPICAL FUEL SYSTEM - GAS TURBINE ENGINE



TYPICAL FUEL SYSTEM - GAS TURBINE ENGINE



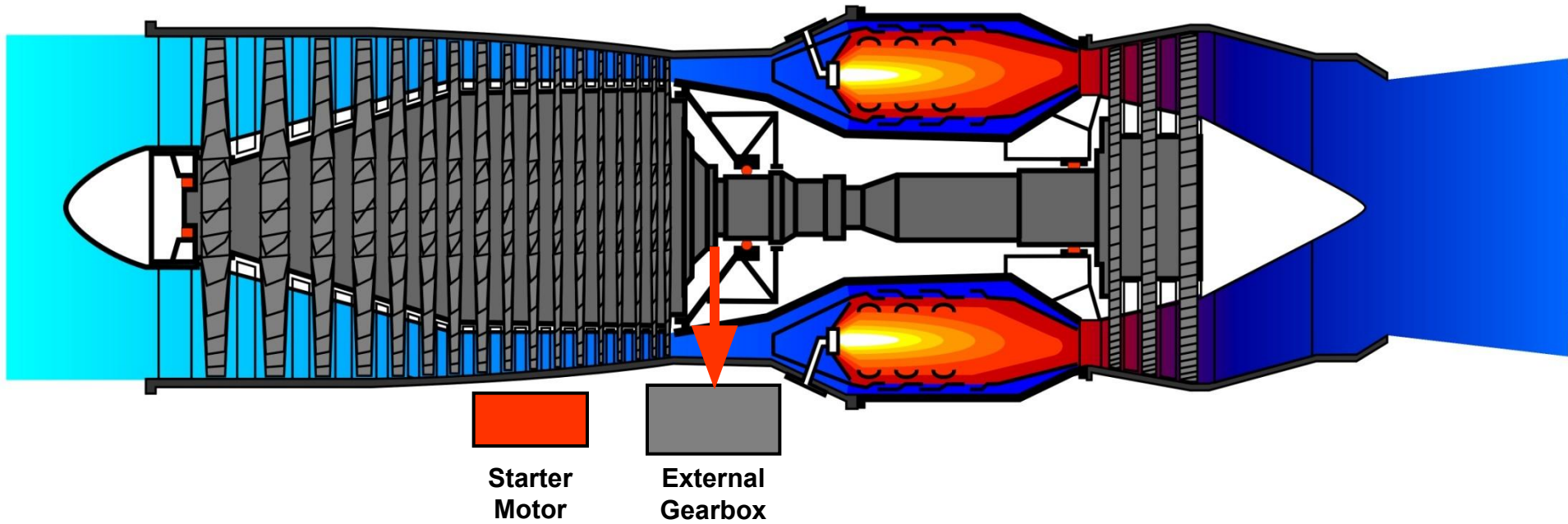
GEAR TYPE PUMP

TYPICAL FUEL SYSTEM - GAS TURBINE ENGINE

JET ENGINE STARTING AND IGNITION SYSTEM

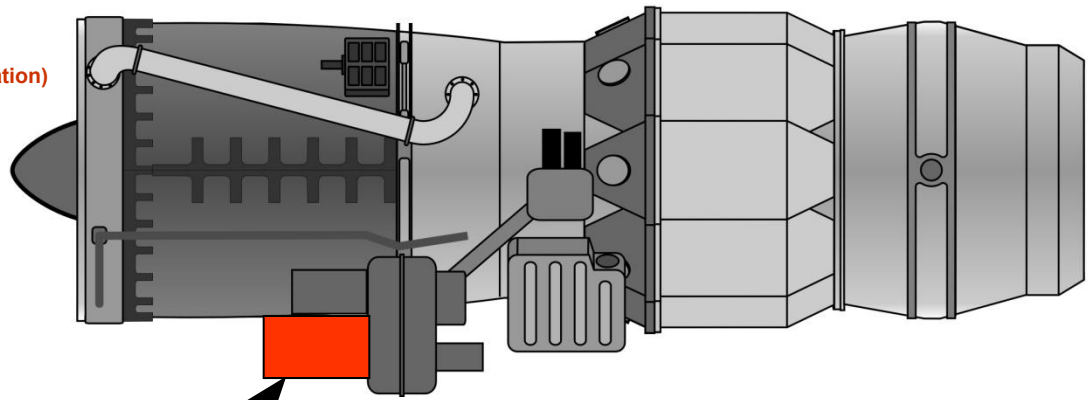
GETTING A JET ENGINE GOING

The basic sequence of the start cycle



JET ENGINE STARTING AND IGNITION

Rolls-Royce AVON - (Graphic Representation)

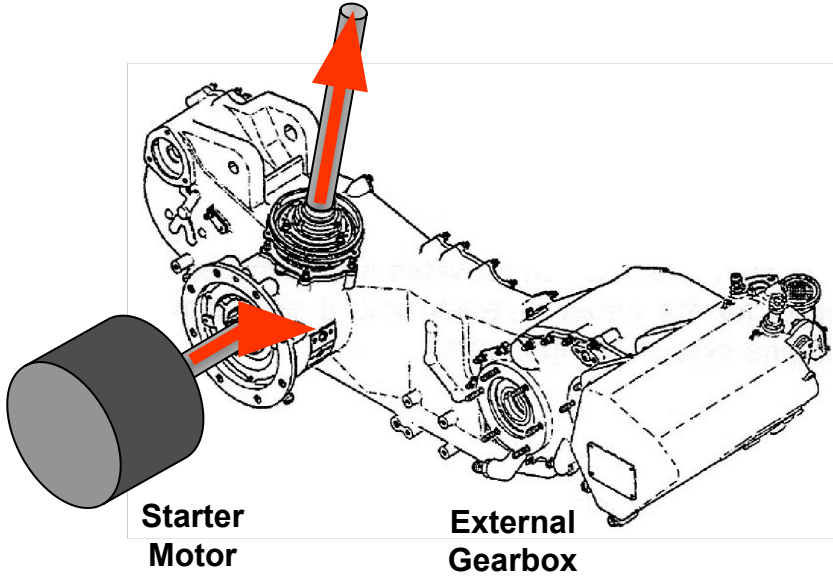


Drive to
Engine Shaft

Starter
Motor

Air
Out

Clutch
Mechanism



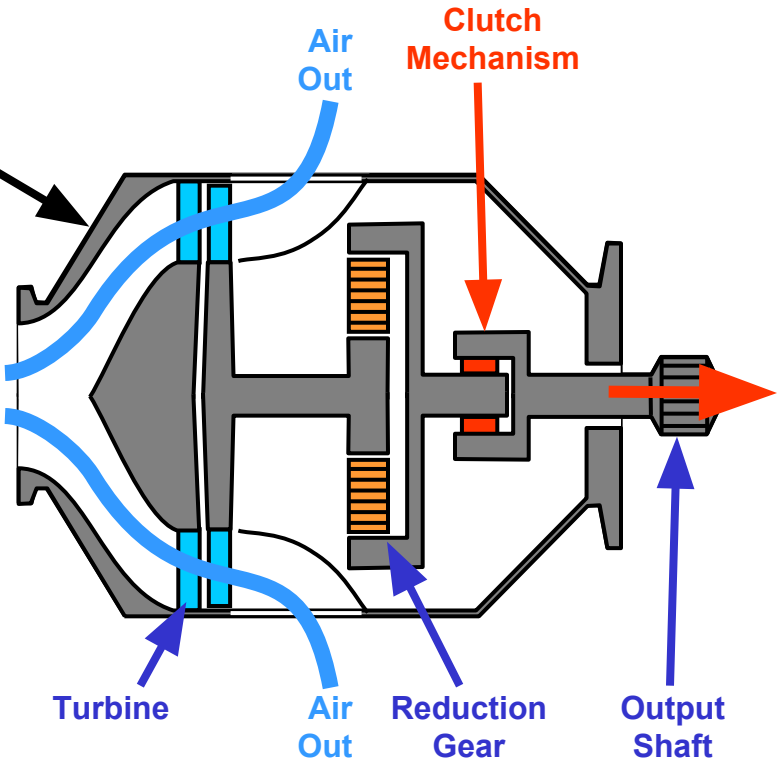
Air
In

Turbine

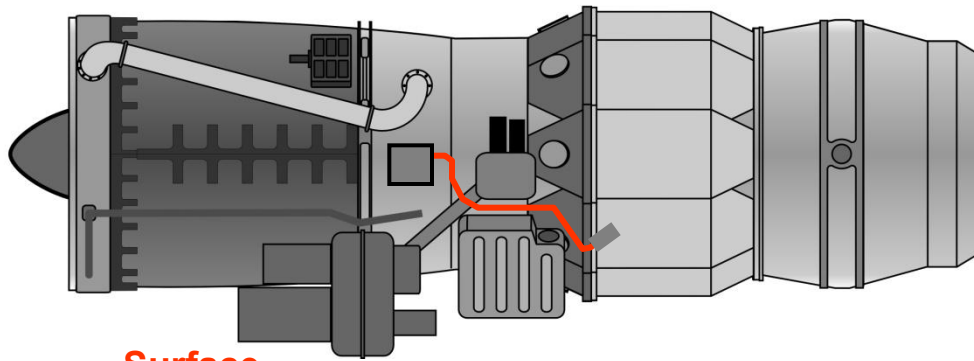
Air
Out

Reduction
Gear

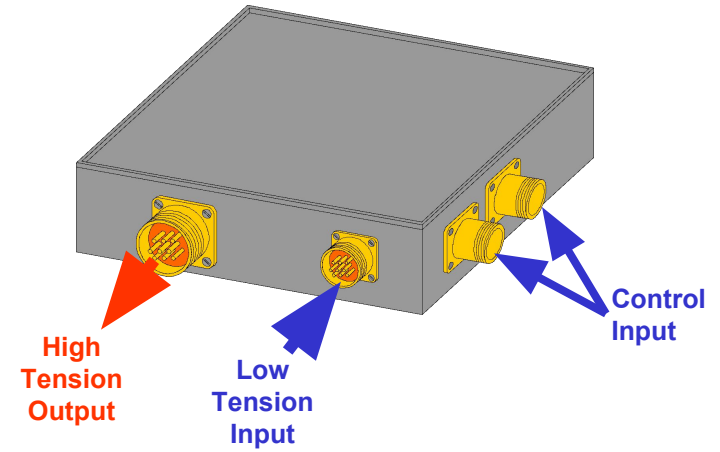
Output
Shaft



JET ENGINE STARTING AND IGNITION

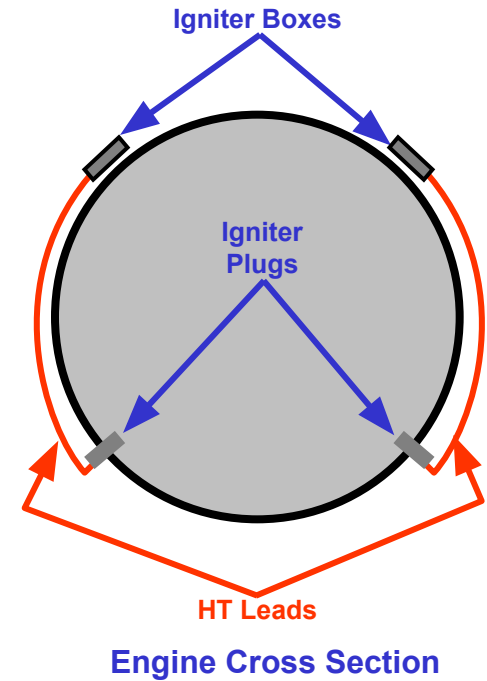
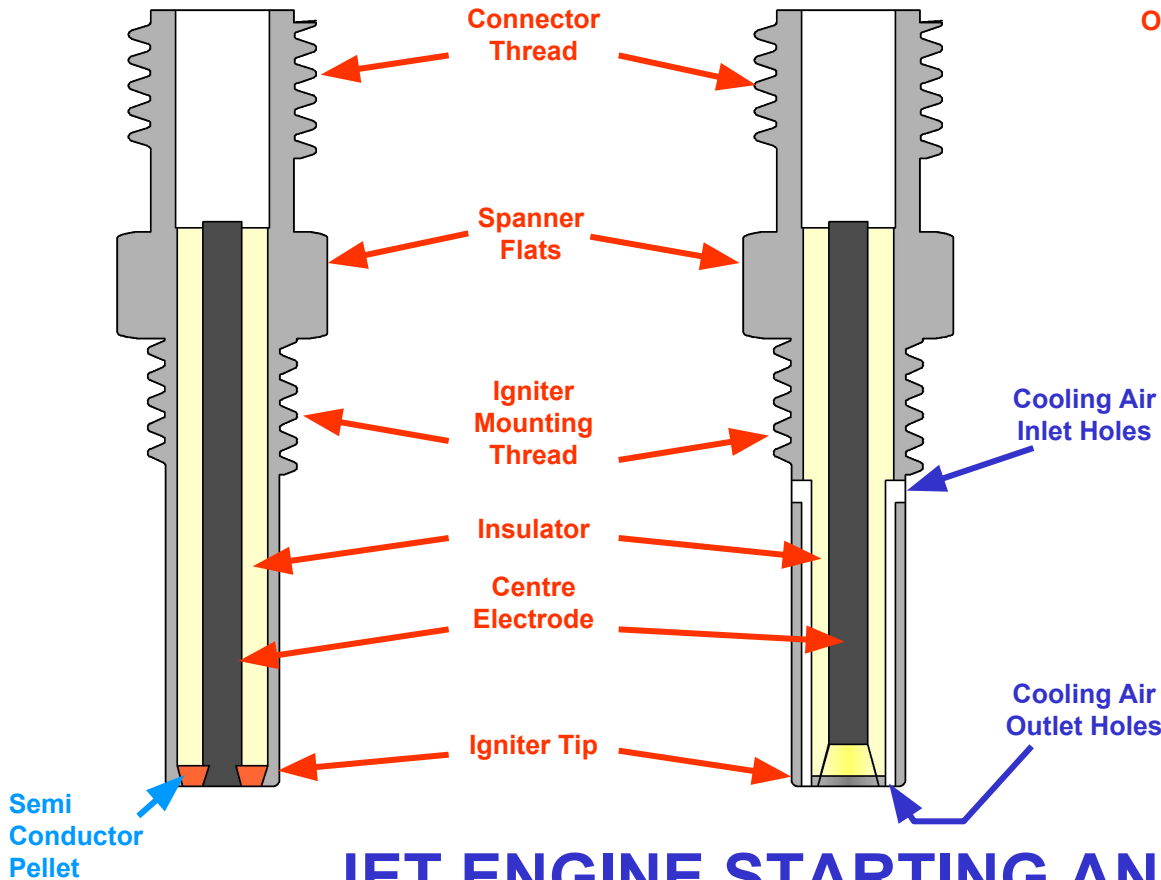


Igniter Box – (Not to Scale)



Surface Discharge

Air Gap



JET ENGINE STARTING AND IGNITION

Thrust Augmentation

What is it?

A method of extracting more power from Internal Combustion Engines

Piston engines:-

Supercharging or turbo charging

Forced air aspiration

Higher Octane fuels

More power per bang

Gas turbines engines:-

Booster engines

HS121 Trident and early V/STOL a/c

Water injection + more fuel

Methanol

Water injection

Afterburning

Mainly military but also Concorde

Other methods

Rocket boosters

RATO – Rocket Assisted Take-Off

Single use only

Thrust Augmentation

Piston engines:-

Supercharging or turbo charging

Forced air aspiration

Higher Octane fuels

More power per bang

Gas turbines engines:-

Booster engines

HS121 Trident and early V/STOL a/c

Water injection + more fuel

Methanol

Water injection

Afterburning

Mainly military but also Concorde

Other methods

Rocket boosters

RATO – Rocket Assisted Take-Off

Single use only

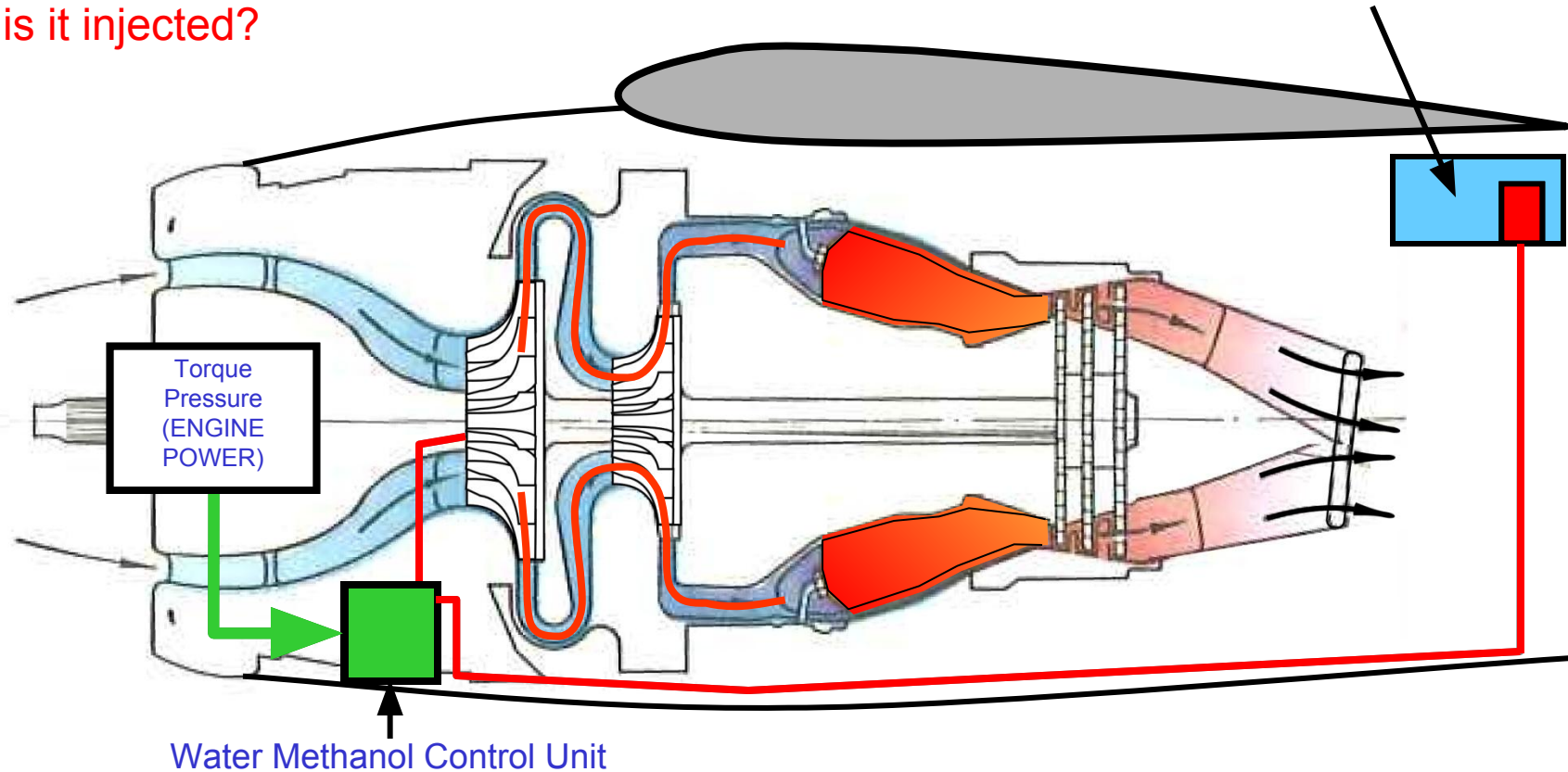
Thrust Augmentation

Water Methanol Injection

Dart Engine Series

Water Methanol
Tank and Pump

How is it injected?



Water cools the turbine and increases pressure and therefore velocity = more power.

Cooler turbine means more fuel (methanol) can be added = more power.

In the Dart engine, more power means coarser propeller pitch = more thrust.

Use limited because of combustion characteristics

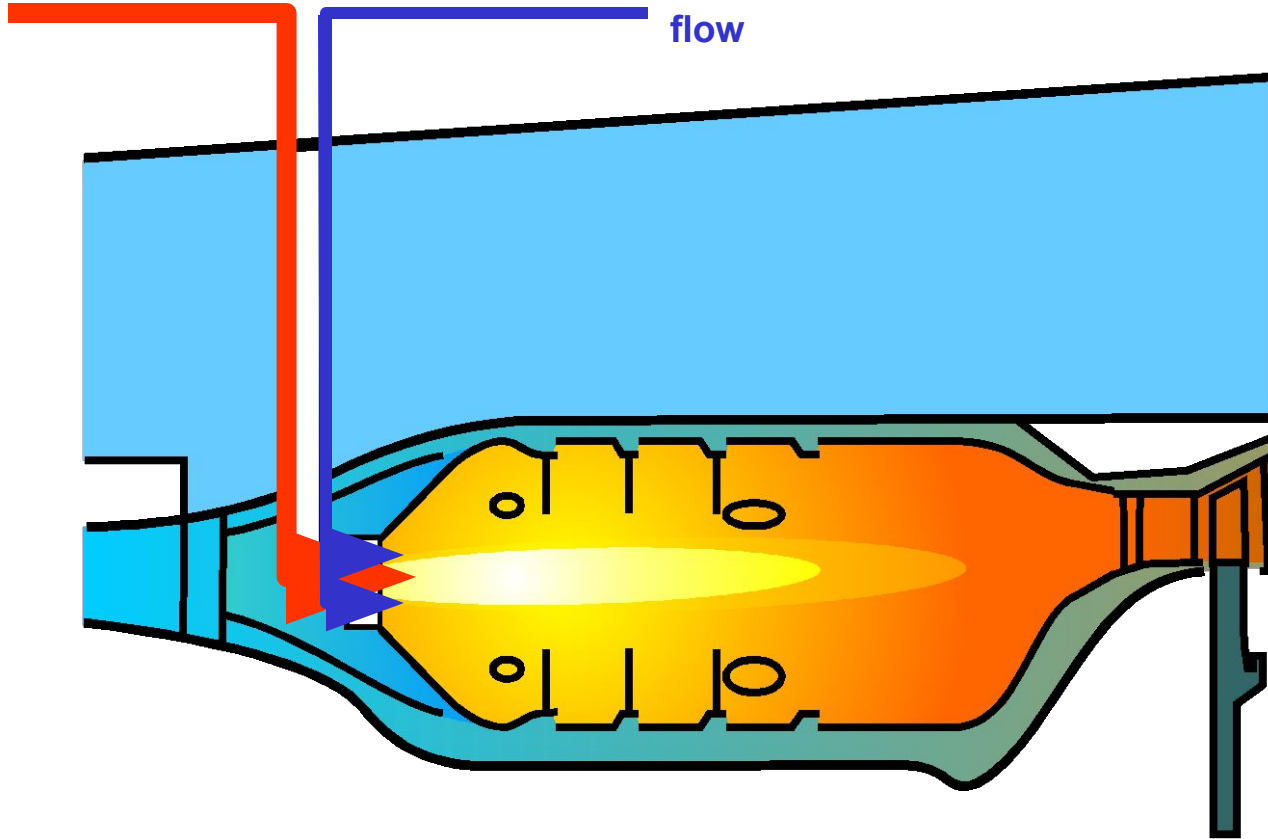
Thrust Augmentation

Water Injection

Rolls-Royce Spey Series

Increased
Primary
Fuel flow

Water
flow



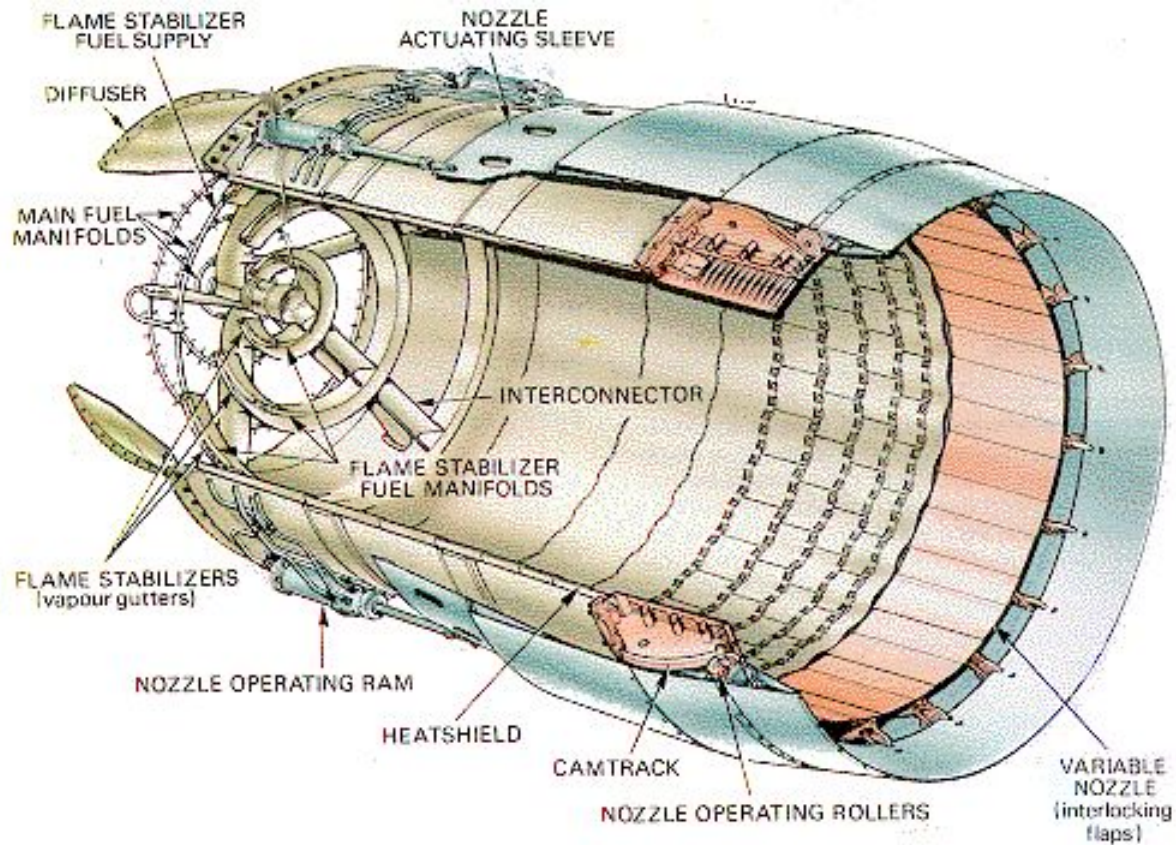
Water cools the turbine and increases pressure and therefore velocity = more power.

Cooler turbine means more fuel can be added = more power.

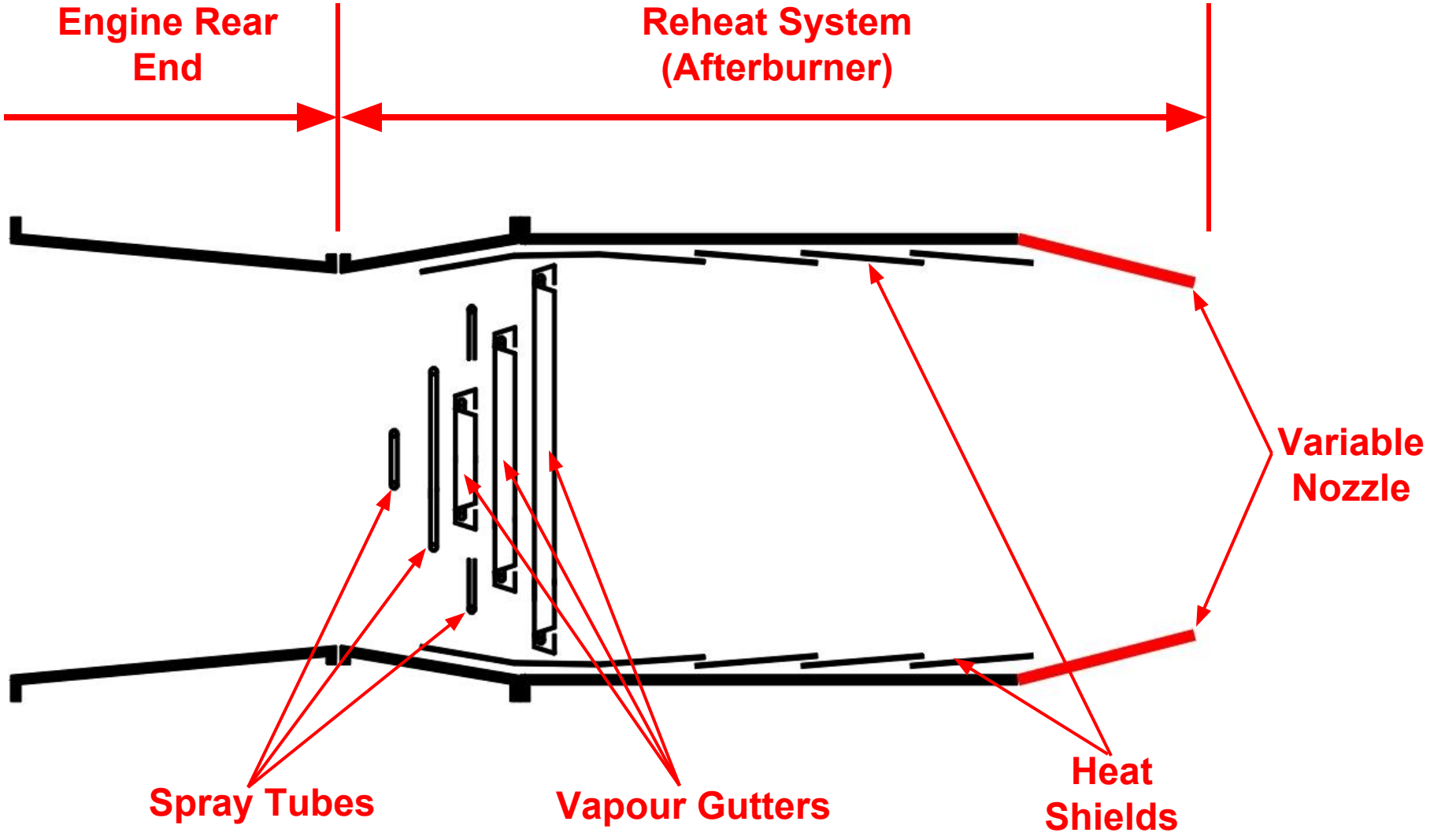
Thrust Augmentation

Typical Reheat System

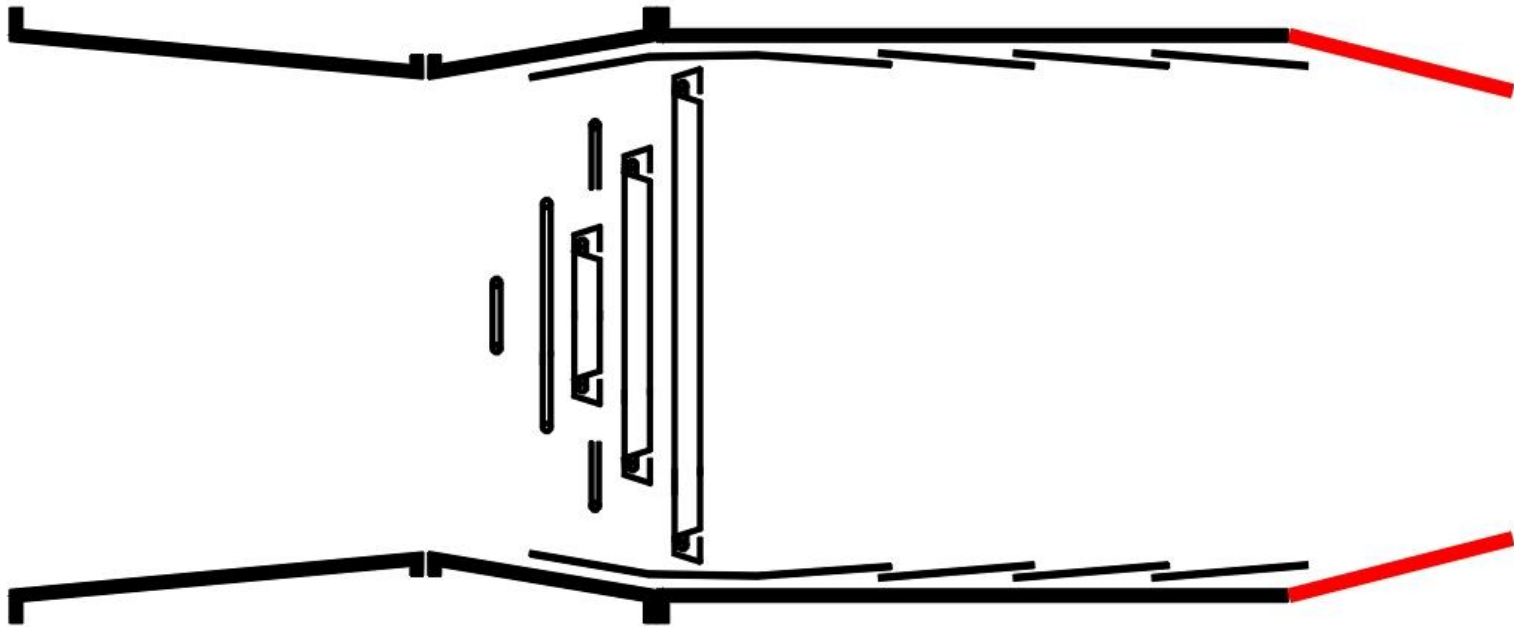
Rolls-Royce Phantom Spey and Jaguar Adour series



Thrust Augmentation

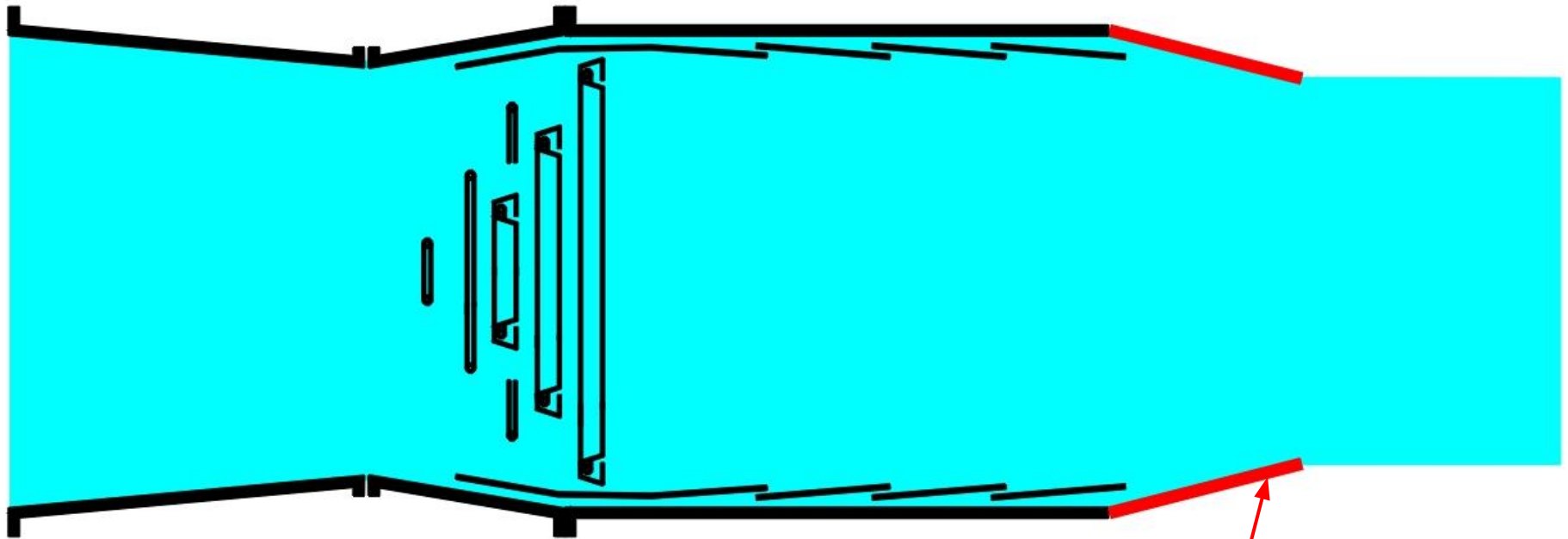


Thrust Augmentation



Thrust Augmentation

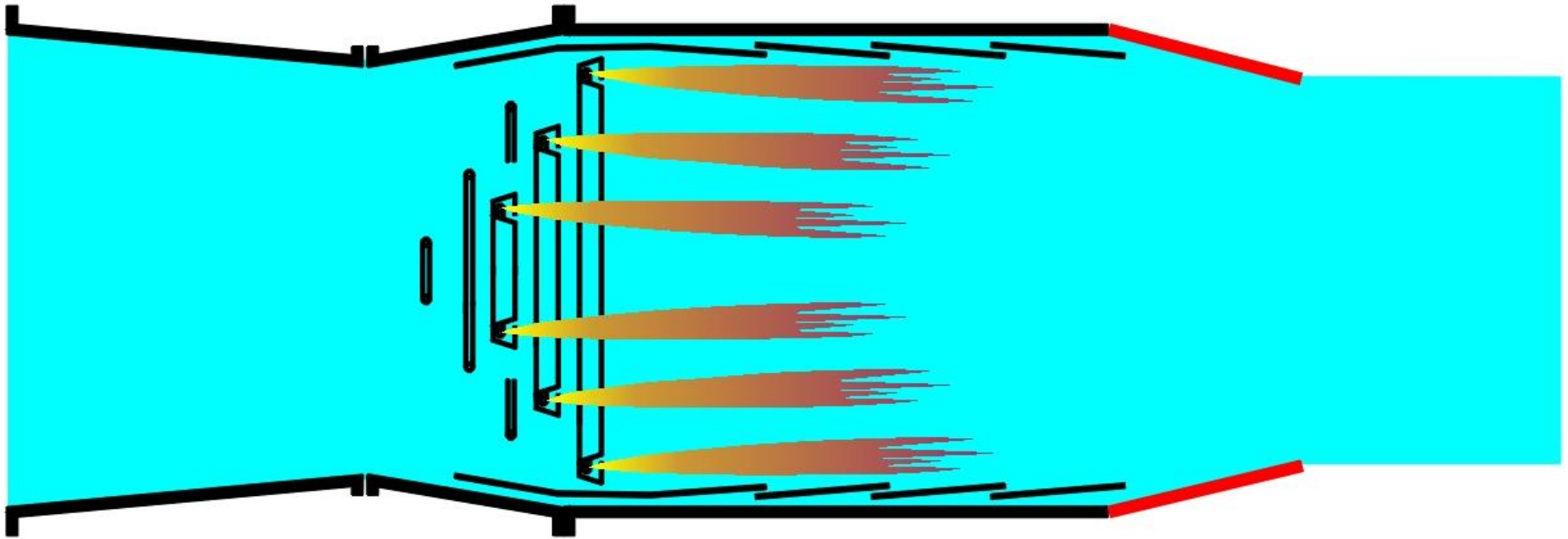
Engine Gas Stream *without* Reheat



Variable Nozzle 'Closed'

Thrust Augmentation

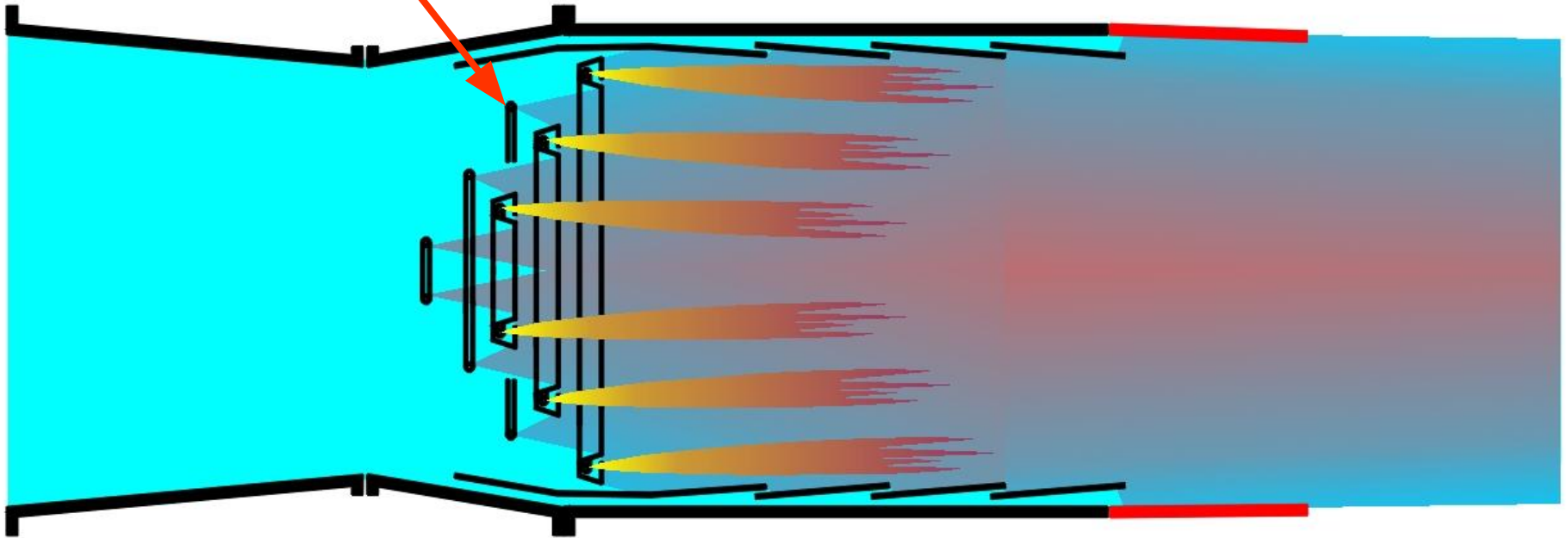
Reheat Starts with Vapour Gutters Spraying Fuel



Thrust Augmentation

Then Spray Tubes add more fuel to combustion process

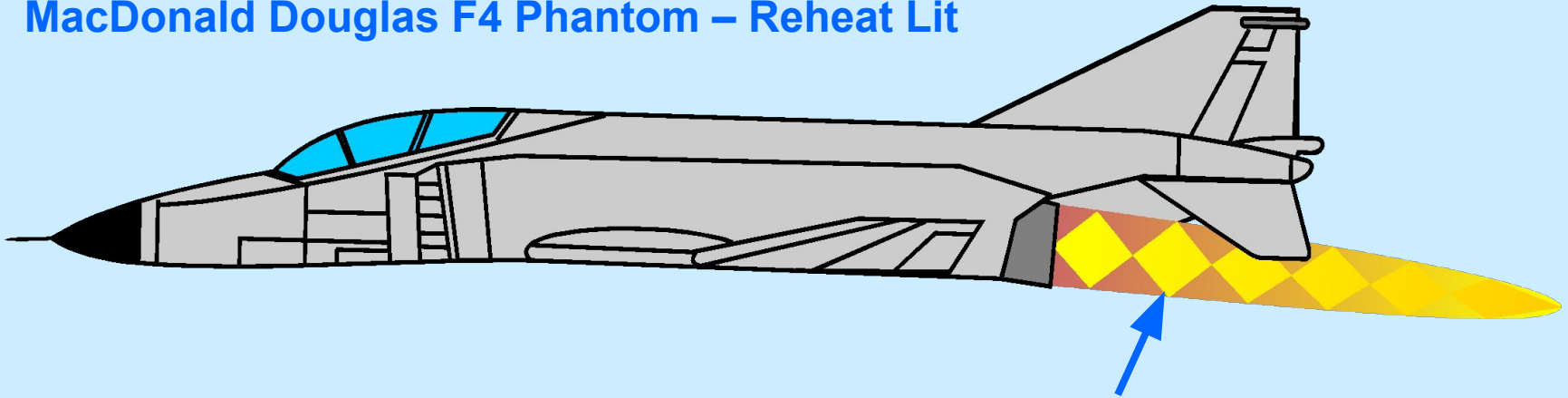
And Variable Nozzle 'Opens'



This is to prevent higher jet pipe pressure
to ensure that reheat has no effect on jet engine

Thrust Augmentation

MacDonald Douglas F4 Phantom – Reheat Lit

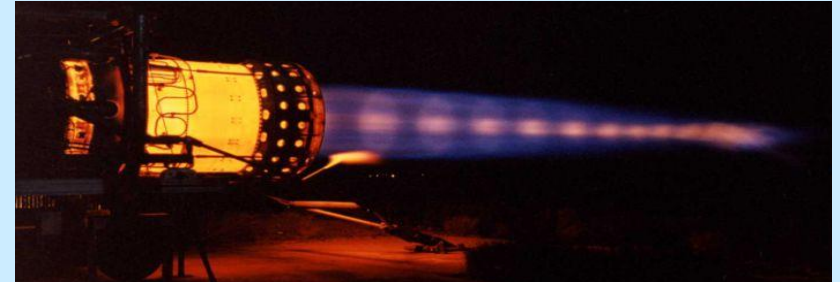


Visible 'Shock' waves appear in exhaust plume!



FA18 Hornet on Carrier
Take-off – Reheat Lit

Does the Afterburner get **hot**?



Just a little bit!!

Use of the afterburner is restricted to take-off and combat only (except Concorde of course) due to the extremely high fuel consumption rate

Thrust Augmentation

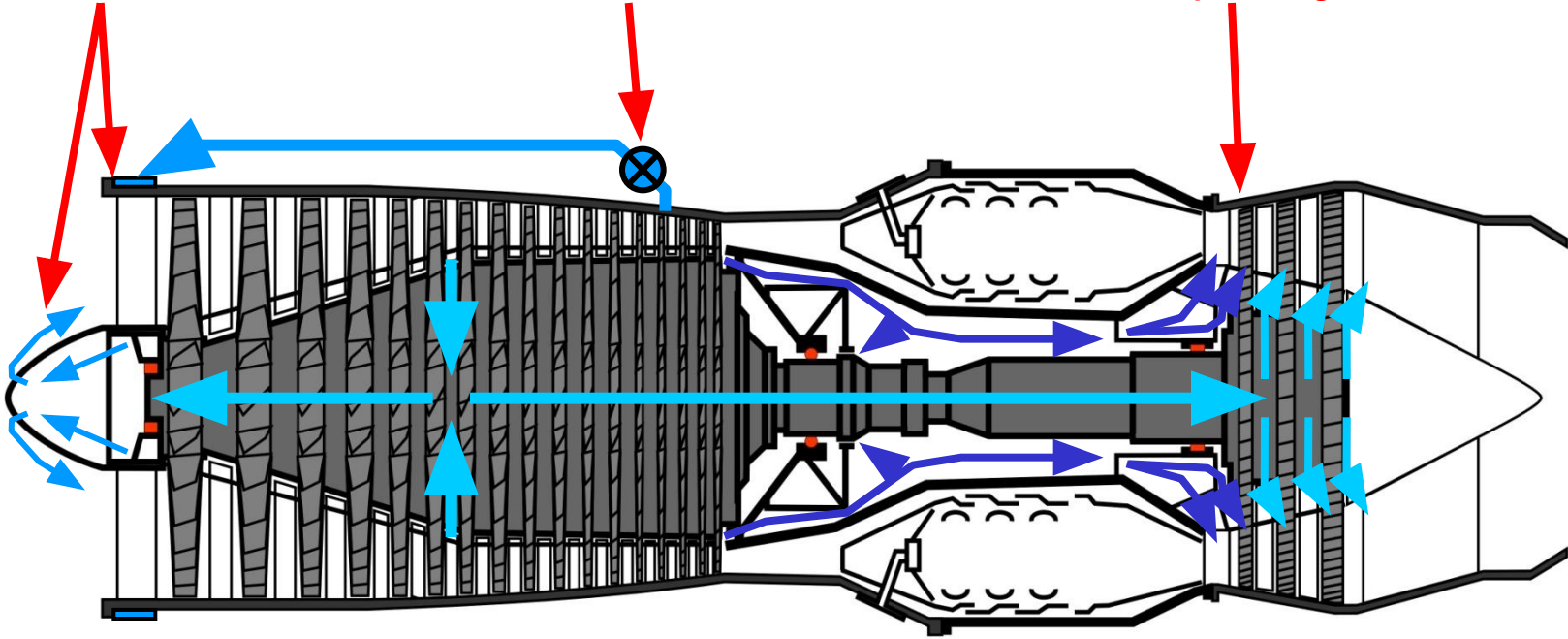
JET ENGINE INTERNAL COOLING AND SEALING

PREVENTING BURN OUT OF THE **HOT** SECTION

**Inlet Vane
and Bullet
Anti Icing Air**

**Anti Icing Air
Control Valve**

**Disc Cooling,
Blade Film Cooling
and Tip Sealing**



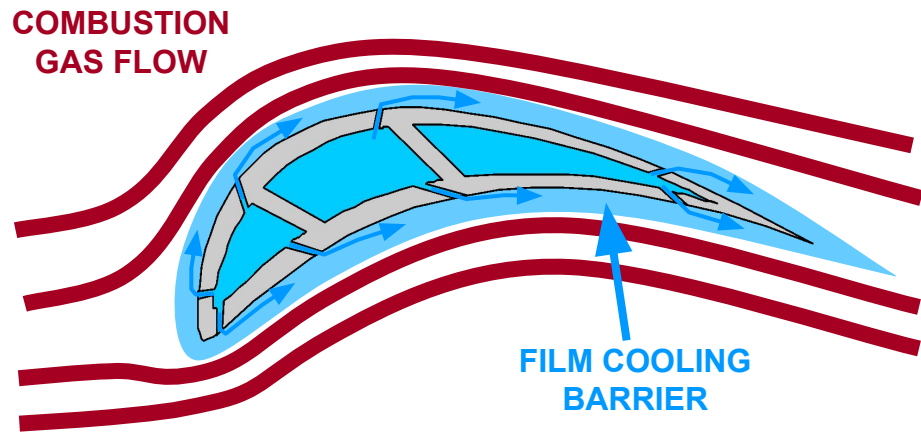
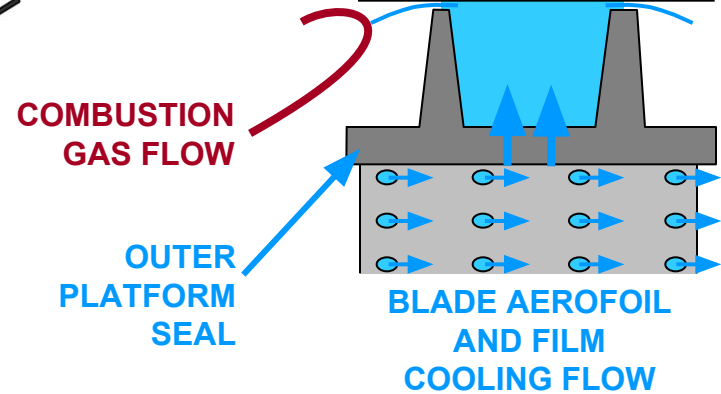
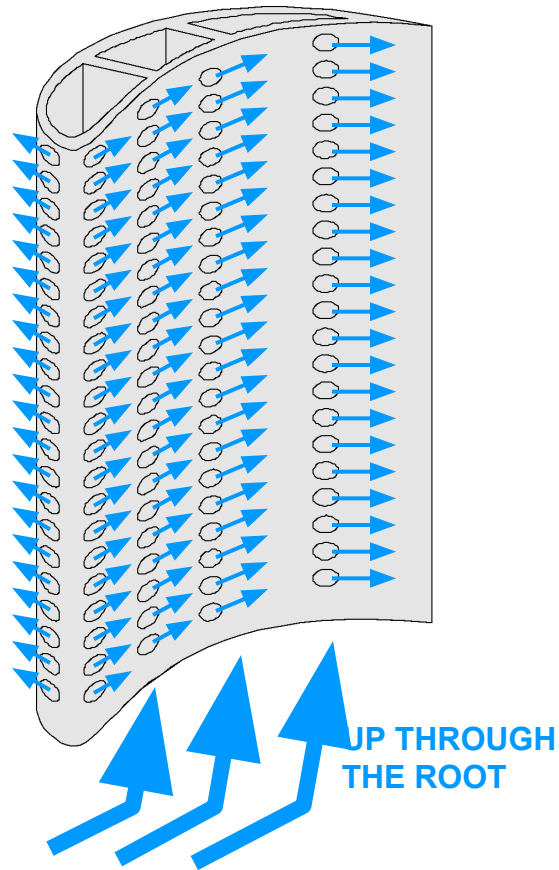
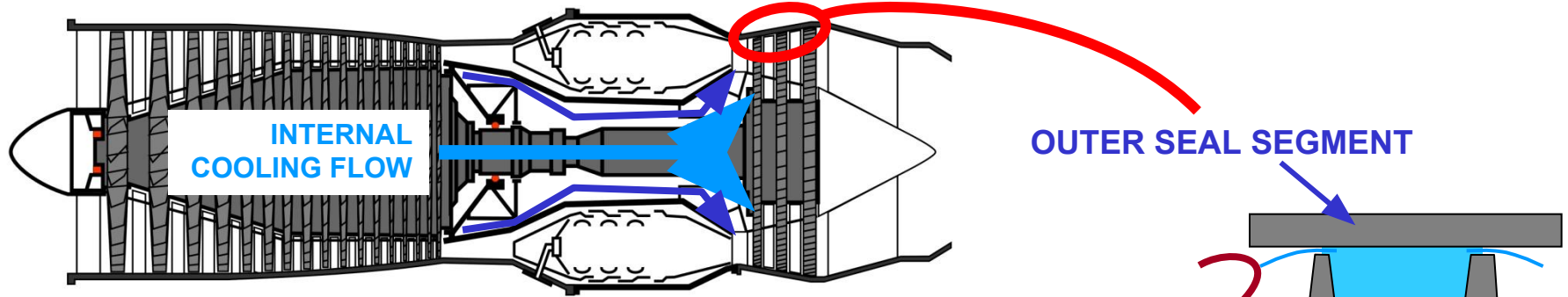
Stage 6 Internal Cooling Air

Stage 14 Anti Icing Air

Stage 17 Internal Cooling and Sealing Air

**Notional Internal Cooling and
Sealing System**

JET ENGINE INTERNAL COOLING AND SEALING

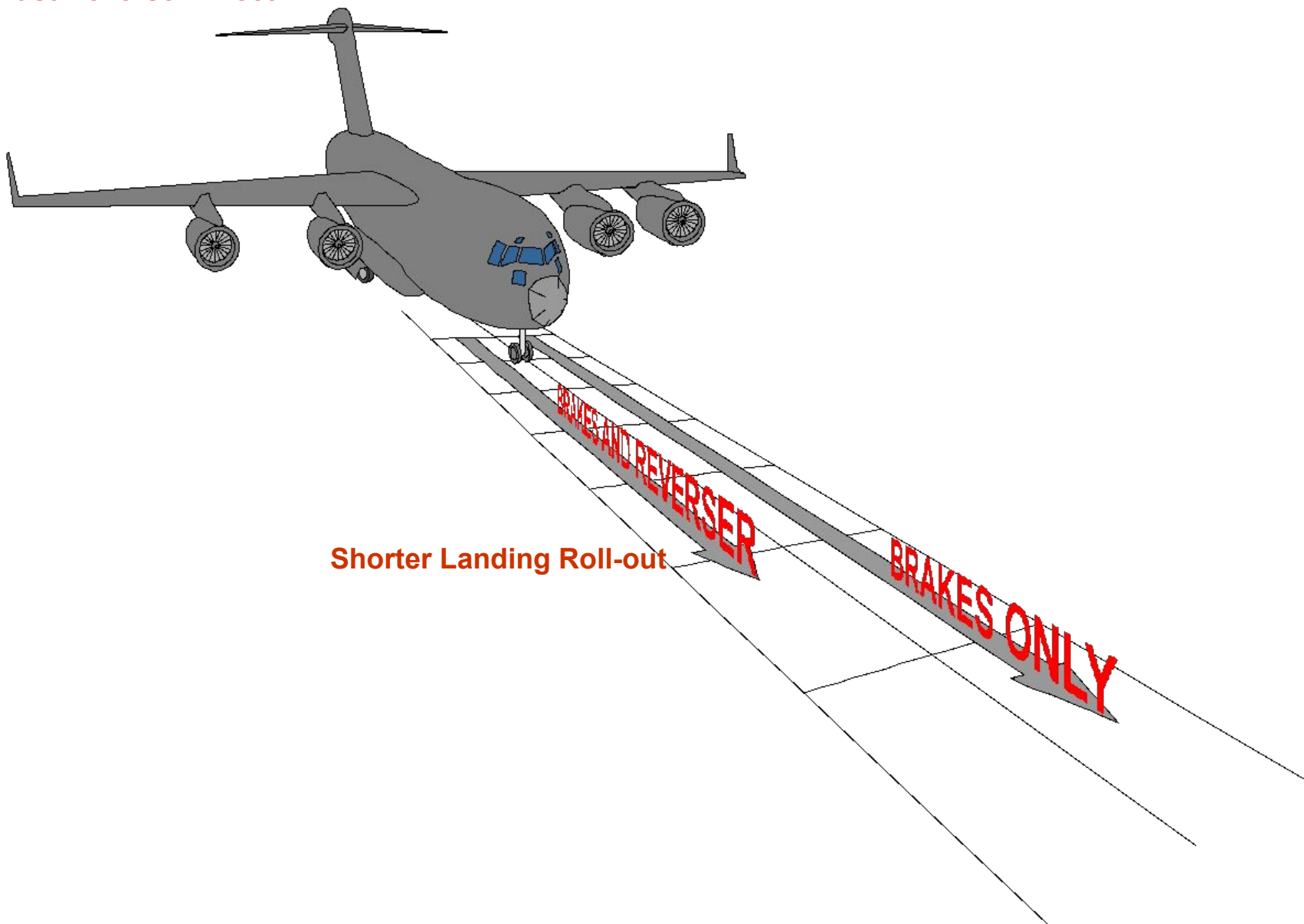


JET ENGINE INTERNAL COOLING AND SEALING

JET ENGINE THRUST REVERSERS

Putting the Brakes On

Thrust Reverser Effect



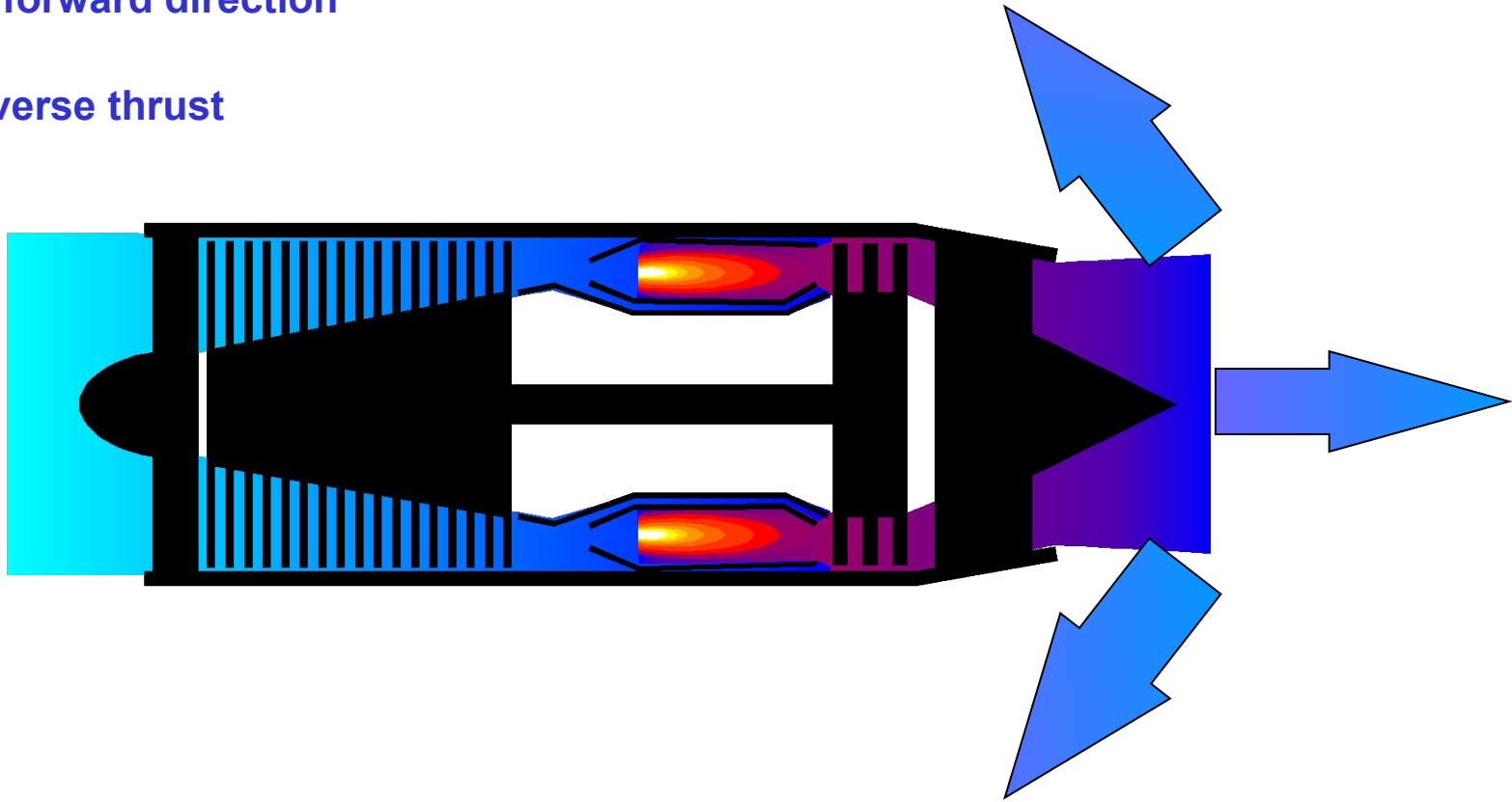
JET ENGINE THRUST REVERSERS

BASIC PRINCIPAL

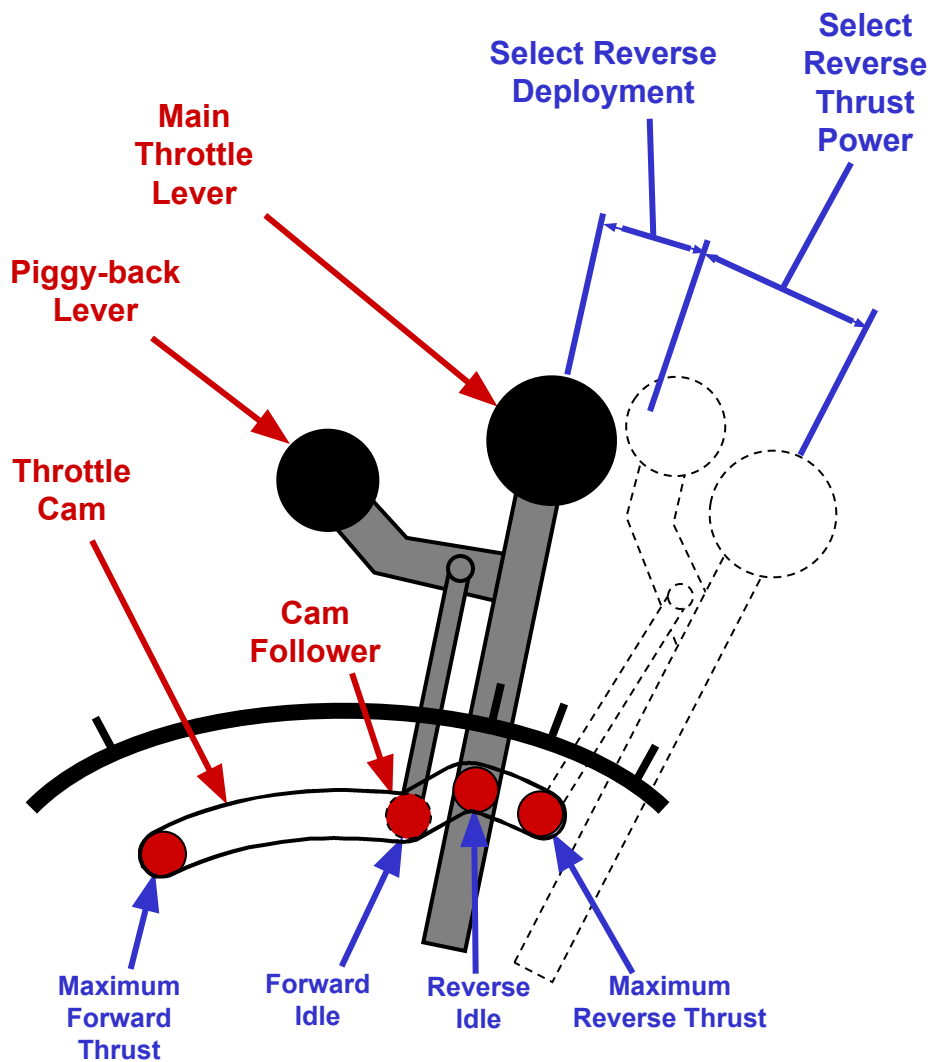
Divert some or all of the rearward directed jet

in an almost forward direction

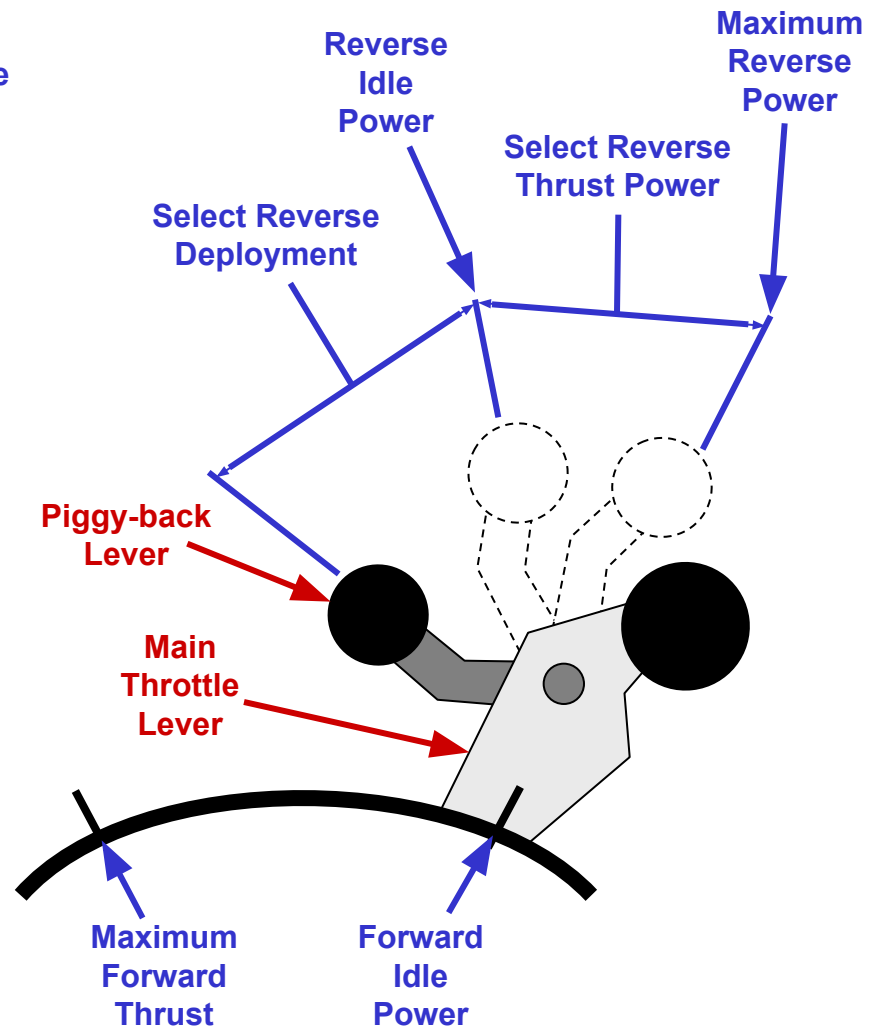
providing reverse thrust



TRANSLATING COWL TYPE
JET ENGINE THRUST REVERSERS

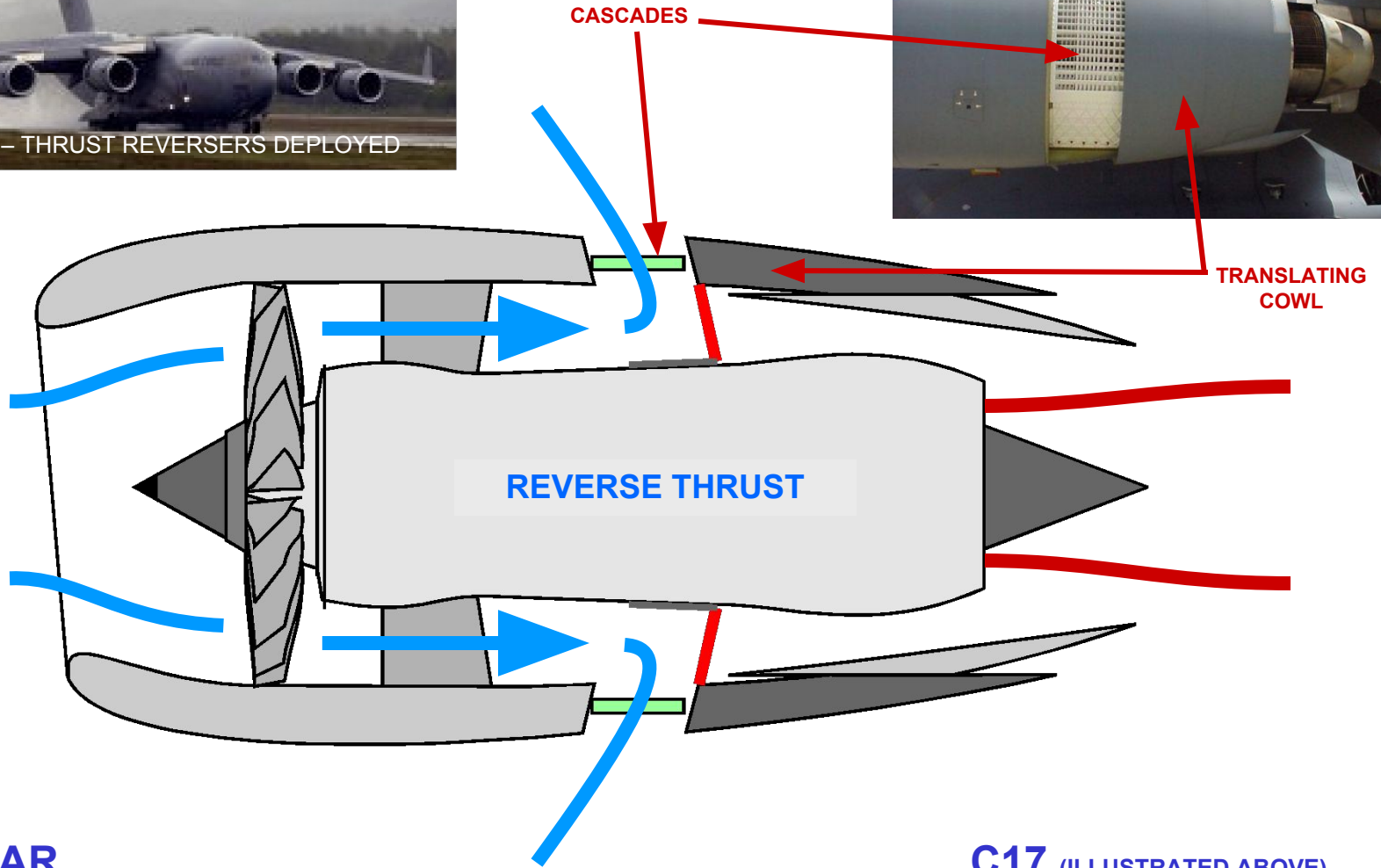
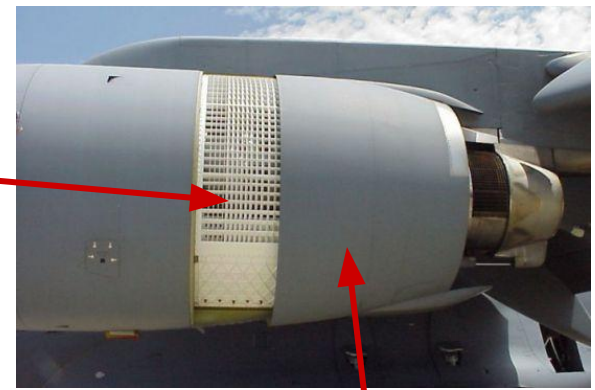


Electro/Hydro Mechanical Control System

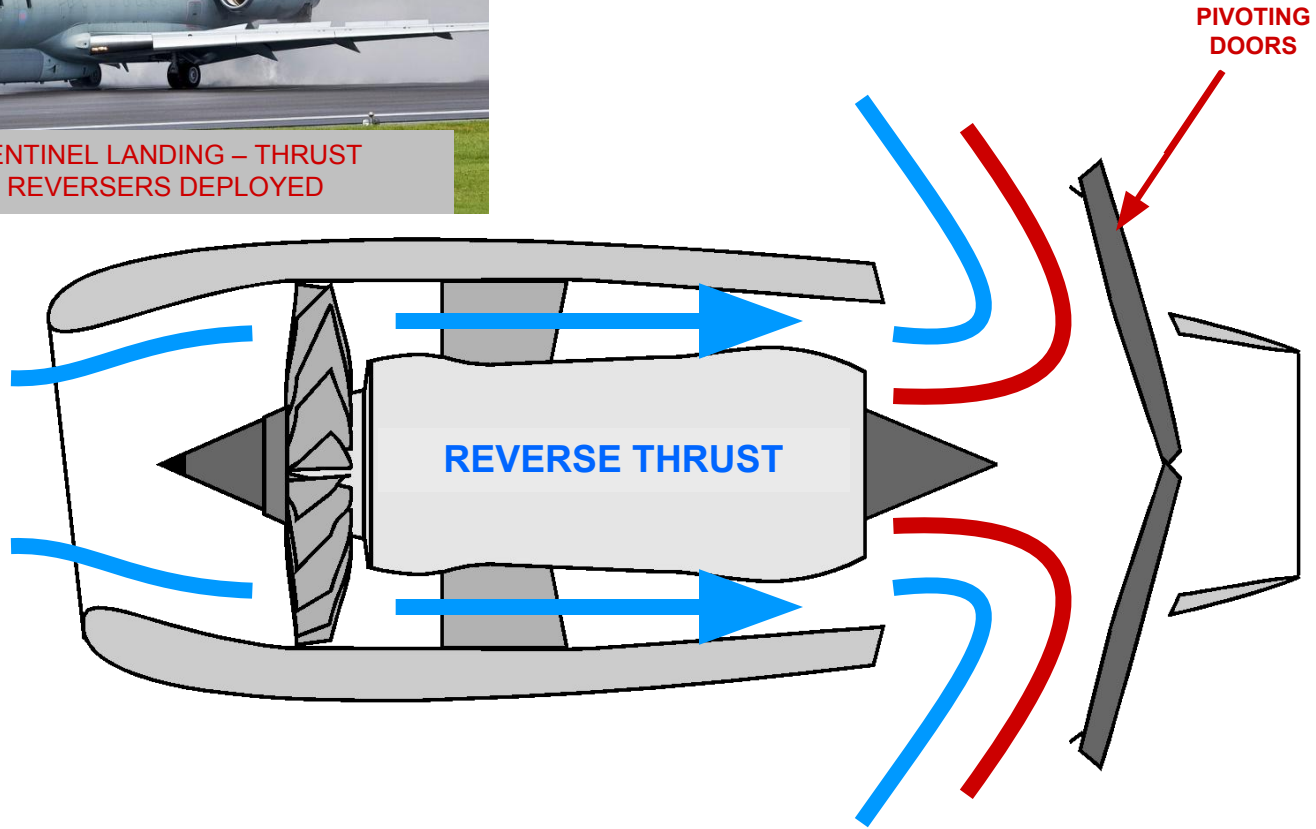


FADEC Control System

Selection, Sequencing and Safety Features JET ENGINE THRUST REVERSERS



TRANSLATING COWL TYPE JET ENGINE THRUST REVERSERS

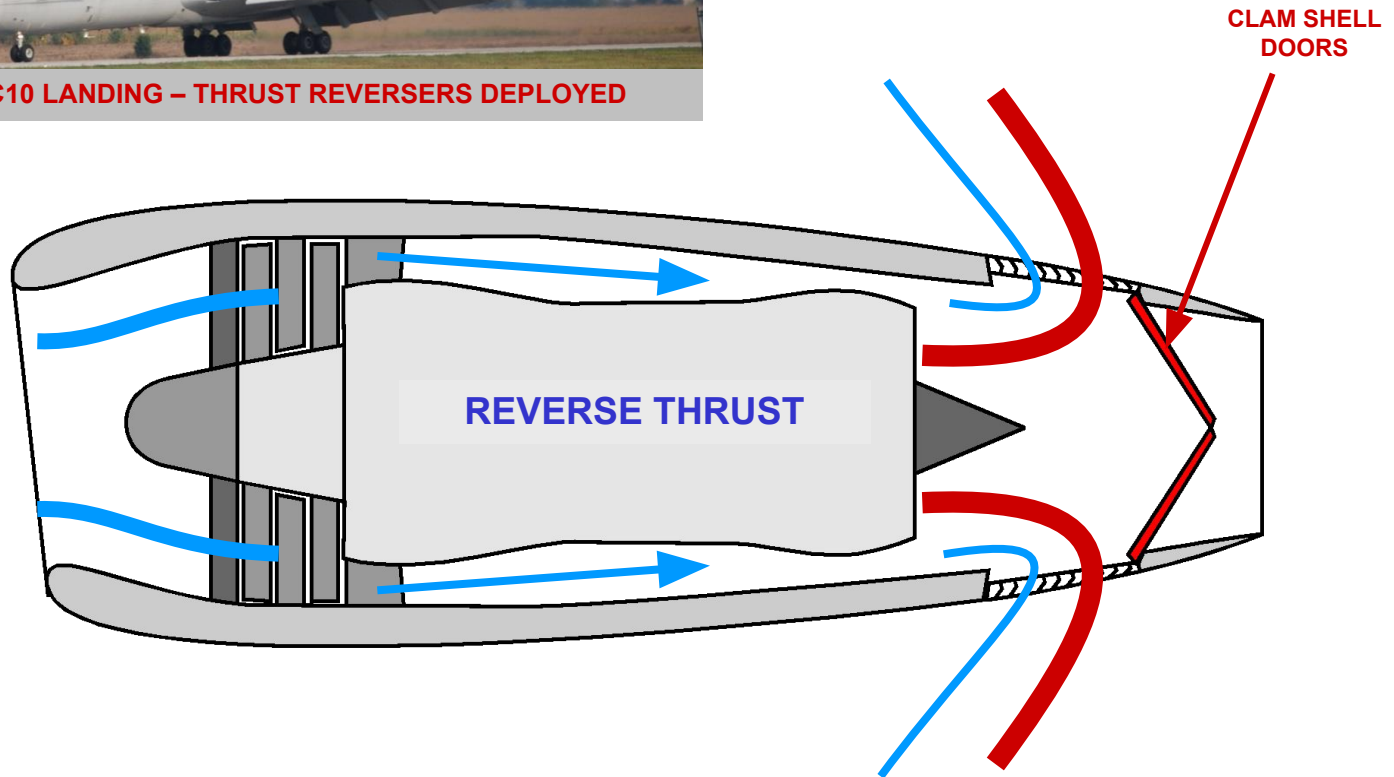


SENTINEL (SHOWN ABOVE)

**PIVOTING DOOR TYPE
JET ENGINE THRUST REVERSERS**



VC10 LANDING – THRUST REVERSERS DEPLOYED

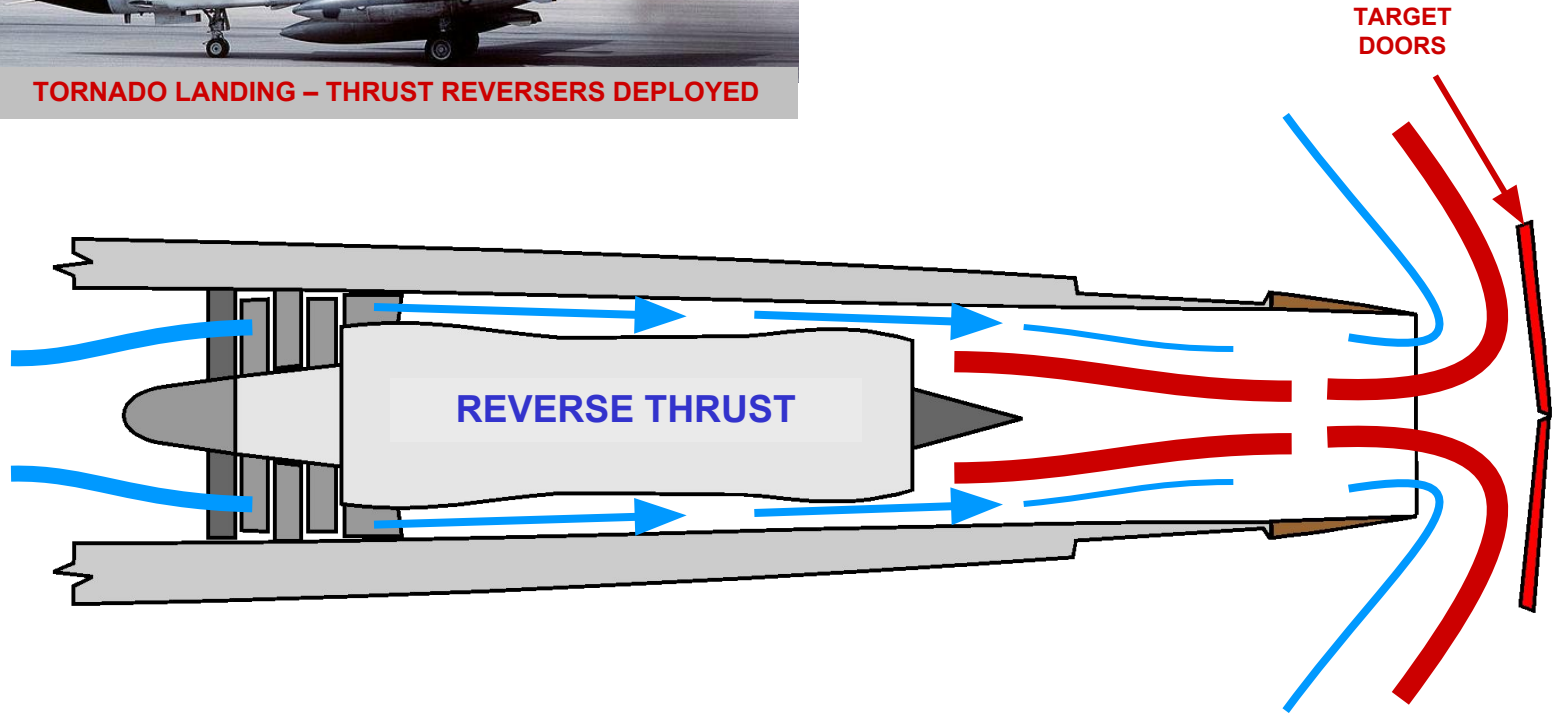


VC10 (SHOWN ABOVE)

**'CLAM SHELL' DOOR TYPE
JET ENGINE THRUST REVERSERS**



TORNADO LANDING – THRUST REVERSERS DEPLOYED



TORNADO (SHOWN ABOVE)

**TARGET DOOR TYPE
JET ENGINE THRUST REVERSERS**



HERCULES

REVERSE PITCH PROPELLER

SEE PROPELLER SECTION FOR DETAILS



VULCAN

DROGUE PARACHUTE

**OTHER METHODS OF BRAKE ASSIST
JET ENGINE THRUST REVERSERS**